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# TECHNICAL EVALUATION OF THE 8-INCH MAJOR CALIBER LIGHTWEIGHT GUN MOUNT, MARK 71 MOD O

D. L. Bowen



U.S. NAVAL WEAPONS LABORATORY DAHLGREN VIRGINIA

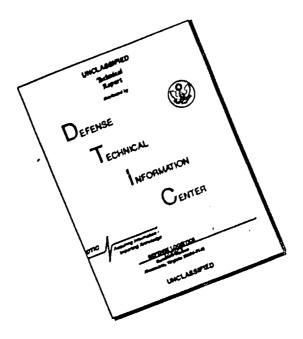






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1. NWL Technical Report TR-2854 of February 1973 "Technical Evaluation of the 8-Inch Major Caliber Lightweight Gun Mount, MARK 71 MOD O" contains two errors. The following pen-and-ink corrections should be made by addressees:

a. Interchange the page number and legends on pages A-19 and A-20.

b. The figure on page A-17 should be labelled "Figure 18", "Pressure Transducer Locations - MARK 71 MOD 0 8"/55 MCLGM".

F. W. Kasdor

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### TECHNICAL EVALUATION OF THE 8-INCH MAJOR CALIBER LIGHTWEIGHT GUN MOUNT, MARK 71 MOD 0

bу

D. L. Bowen
Test and Evaluation Department

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#### **FOREWORD**

This is the final report on the technical evaluation of the prototype 8"/55 Caliber Gun Mount, MARK 71 MOD 0. This work was conducted under NAVORD ORDTASK 553 111 090 1.

This report was given technical review by P. J. Olenick, Jr. and H. P. Caster of the Project Engineering Division of the Test and Evaluation Department.

Released by

L. T. BLADES, CDR, USN

Acting Head, Test and Evaluation

Department

#### **ABSTRACT**

A technical evaluation was conducted on the 8-Inch Major Caliber Lightweight Gun Mount MARK 71 MOD 0 at the Naval Weapons Laboratory between 24 September and 12 November 1971. The evaluation included determining performance parameters of the laying and loading systems, electrical power consumption levels for the mount, checking misfire and regunning provisions, a safety analysis, and firing tests. During the firing tests:

- a. The mount was proofed satisfactorily.
- b. Rate of fire of the mount was determined to be 11.7 to 12.6 rounds per minute.
- c. Accuracy data were gathered indicating that round-to-round accuracy was satisfactory; although, a new 8-inch range table is recommended.
- d. Barrel life and velocity loss data were examined (including that from the operational evaluation) and, although too limited to be conclusive, the data do not indicate barrel wear and velocity loss to be excessive.
- e. A gun blast profile for the gun was established, blast being slightly less severe than that of an 8-inch bag gun.
- f. Smoke and carbon monoxide measurements were taken indicating it would be unsafe to circulate air within the mount to a ship's manned space during firing or for maintenance personnel to enter the mount after firing until the mount is purged.

During the testing, several problems were encountered which required minor design changes by the contractor, and several changes are recommended for incorporation in the production version. After completion of the technical evalutation, performance of the mount was judged acceptable and certification for release to operational evaluation was recommended.

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#### I. INTRODUCTION

The Major Caliber Lightweight Gun Mount (MCLGM) is a result of Navy and Marine Corps efforts beginning in 1960 to provide the fleet with the first major caliber gun mount to be designed since the end of World War II. This mount would provide firepower necessary to support troops ashore at ranges well beyond that of 5.0 guns and would provide greatly increased firepower for the fleet, particularly for smaller ships. A 175mm/60 caliber mount was specified by Reference 1 because of the potential advantages of achieving commonality in ammunition with the Army 175mm gun. Northern Ordnance Division of FMC Corporation at Minneapolis (NOD/fmc) was awarded the contract to design and develop the mount. A prototype was fabricated and subjected to testing at NOD/fmc, including a 50,000 cycle life and reliability test on the loading system. The mount was disassembled and delivered to the Naval Weapons Laboratory, Dahlgren, Virginia in August 1970 where it was reassemble checked, proof-fired and evaluated to verify that contract requirements were met.

The Army, however, developed only one type of projectile for an anticipated family of ammunition; terminated plans for future ammunition development; planned the phaseout of the 175mm gun; and commenced development of an "up-gunned" 8-inch howitzer. Accordingly, the original SOR, Reference 1, was revised and superseded by Reference 2 in October 1969. The MCLGM was directed to be an 8-inch gun in production using the basic design established in the development of the 175mm prototype. To meet these new requirements, a program was authorized in February 1971 to convert the prototype MCLGM from its 175mm/60 configuration to an 8"/\$5. Thus, the Naval Weapons Laboratory, by Reference 3, was assigned the responsibility to effect this conversion and to evaluate the 8"/\$5. MCLGM MARK 71 MOD 0.

The oscillating assembly was returned to NOD/fmc in March 1971 for modification of the recoil and counterrecoil system to handle the increased loads when firing 8"0 MARK 25 projectiles. Since the cartridge case remained the same for both the 175mm and 8"0 applications, only minor modifications were required to the ammunition handling system to accommodate the heavier projectile. These were accomplished on the mount without removing it from its permament emplacement. Two 175mm liners were converted to an 8"0 by Watervliet Arsenal. A four caliber extension was added to one liner so that it might be used in the mount, the second liner was converted to an 8"/51 caliber test gun for interior ballistic development programs. Two new 8"/55 caliber liners were also machined from new forgings. The converted oscillating assembly was returned to NWL in August 1971 and the mount was reassembled. The 8"0 MCLGM prototype mount was proofed with the EX 28 MOD 0 liner (with the extension) on 3 September 1971.

Reference 4 requested a concurrent evaluation of the 8.0 MCLGM MARK 71 MOD 0 at NWL Dahlgren, however, a separation of the Technical and Operational Evaluations were subsequently authorized and they were performed by NWL and OPTEVFOR, respectively. Both evaluations have now been completed. A preliminary report of results of the technical evaluation was submitted by Reference 5, and this document serves as the final report. Results of the Limited Scope Operational Evaluation (Project 0/S 173) were reported by Reference 6.

The general objectives of the Technical Evaluation were to demonstrate mount performance parameters and to assure that the system was in suitable condition to undergo a Limited Scope Operational Evaluation (OPEVAL). The use of the term "OPEVAL" elsewhere in this report refers to this Limited Scope Operational Evaluation which was conducted at NWL without a complete system evaluation of fire control and ship installation interfaces. An at-sea OPEVAL of the MARK 71 MOD 0 mount was subsequently recommended in Reference 6. The technical evaluation was begun on 24 September 1971 and concluded, except for several tests of secondary importance, on 12 November 1971. The OPEVAL was begun on 15 November 1971 and concluded on 7 January 1972. During January and February 1972, the Technical Evaluation was completed. Although the results of the OPEVAL are reported elsewhere. (Reference 6) firing data accumulated are included here since they apply to such items as ballistic performance of the mount which is of interest from the technical viewpoint.

#### II. DESCRIPTION OF THE MOUNT

#### A. General

The 8"/55 Major Caliber Lightweight Gun Mount MARK 71 MOD 0 consists functionally of three systems. A gun laying system positions the gun while the gun loading system automatically loads single rounds of ammunition into the breech, completes the firing circuit, and disposes of empty cases. A control system regulates and monitors mount operations. Characteristics of the mount are listed in Table 1.

#### B. Majo: Units

The major units of the 8''/55 MCLGM are: (See Figure 1 of Appendix A and Table 2.)

Upper Structure. The primary components of the upper structure are the stand, carriage, slide and shield. They are all above-deck components.

Stand. The stand is a circular weldment on the open deck that forms a bearing and rollerpath foundation for the mount's rotating structure. A ring gear on the stand mates with a drive pinion to move the rotating structure in train.

Carriage. The carriage consists of a base ring and two trunnion supports (see Figure 2 of Appendix A). All components of the upper structure are mounted on the carriage. A water shield and a water seal are mounted to the stand and carriage to protect the thrust and radial roller bearings from water in a shipboard installation. The train stowing pin is used during mount transport or major repair to lock the gun mount carriage. The pin is installed through a hole in the base ring into the stand. The hole, which is located near the rear of the mount, is covered unless the pin is in place. The upper accumulator is located on the carriage. It supplies hydraulic fluid to the gun loading system components on the rotating structure.

Slide. The slide (see Figure 3 of Appendix A), which is a rectangular wildment with an integral trunnion on each side, is supported by the carriage trunnion supports. The slide contains components of the gun loading system for firing and for disposing of empty cases. The gun barrel which fits into the gun housing, combines with the slide to make up the oscillating assembly. The elevation system moves the oscillating assembly through a gear segment on the bottom of the slide to elevate or depress the gun.

#### TABLE 1

#### **Mount Characteristics**

Ammunition	8"/55 MARK 25 projectiles with new design cartridge assembly.
Rate of fire	12 rounds per minute
Train limits	300 degrees (150° right or left)
Elevation limits	-5 degrees to +65 degrees
Train velocity	30 degrees per second
Train acceleration	60 degrees per second per second
Elevation velocity	20 degrees per second
Elevation acceleration	40 degrees per second per second
Retardation force	216,000 pounds
Length of recoil	36 inches
Mount height (above mounting surface)	168 inches
Trunnion height (above mounting surface)	105 inches
Base ring diameter	246.375 inches
Working circle (including barrel)	394.5 inches
Clearance circle (center of mount to 18 inches beyond shield)	175.3 inches
Mount weight	170,000 pounds
Loader capacity	75 rounds
Power (peak running load)	679 Kw

TABLE 2
Major Components of the Mount

Component	MARK	MOD
Stand	28	0
Carriage	41	0
Elevation Power Drive	72	0
Elevation Receiver-Regulator	56	0
Train Power Drive	73	0
Train Receiver-Regulator	55	0
Slide	34	0
Gun Housing	21	0
Loader	15	0
Shield	67	0
Gun Mount Control	116	0
EP 1 Panel	294	0
EP 2 Panel	295	0
EP 3 Panel	293	0
Upper Accumulator Power Drive	74	0
Lower Accumulator Power Drive	75	0
Gun Barrel	EX 28 EX 28 EX 30	0 1 0

Shield. The shield is a streamlined aluminum cover that mounts on the carriage base ring to protect the above-deck components. The shield has a gun port on the front, left and right access doors, and a circular access cover on the rear. A gun port shield, through which the gun barrel protrudes, moves with the oscillating assembly, keeping the gun port covered.

Gun Laying System. The gun laying system consists of the train and elevation system. The train system moves the rotating structure in the deck plane; the elevation system moves the oscillating assembly perpendicular to the deck plane. Both systems mount on the carriage.

Train and Elevation Systems. The train and elevation systems are similar. They are electrically powered hydraulic drives, each consisting of an electric drive motor, CAB (combination A and B end) units, and a gear-reduction and drive pinion. The train drive pinion revolves around the train circle gear in the stand; the elevation drive pinion moves the gear segment on the slide. The train and elevation systems are controlled by an electronic servo control unit in the EP-2 panel and a receiver-regulator mounted on each CAB unit. Air motors form an auxiliary and emergency power source to drive the train and elevation system.

Automatic Two-Speed Control. A two-speed synchro system controls the gun laying system. The one-speed synchro positions the mount within two degrees of the gun order. When the error signal is reduced to 2 degrees of arc, the 36-speed (fine control) synchro assumes position control of the mount.

Remote and Local Control. A control switch on the EP-2 Control Panel is positioned to select the mode of control for the gun laying systems. Weapons Control (Dummy Directors at NWL) control the train and elevation systems in remote control. The train and elevation systems are controlled locally from the EP-3 Test Panel for test or maintenance.

Automatic Limit-Stops. Both train and elevation systems have a pointing cutout system and a limit-stop system to prevent the gun from moving into specific zones where damage would result. In addition, if a failure of the pointing cutout and limit-stop systems allows the gun to travel beyond its maximum safe limits, emergency limit-stop valves are actuated which set the power-off brake and open safety relief valves to halt the gun mount.

Power-Off Brakes. The power-off brakes hold the train and elevation drives whenever the systems are shut down, or stop, and hold them in the event of a power failure. When the train and elevation systems are activated, the hydraulic system releases the power-off brakes.

Gun Loading System. The gun loading system loads ammunition into the breech, fires the round, extracts the empty case, from the breech, and ejects the case from the mount. Components of the gun loading system are: loader drum, hoist, cradle, rammer, case extractors, breechblock, empty case tray and empty case ejector, which are all powered by the lower and upper accumulator systems.

Lower and Upper Accumulator Systems. The accumulator systems are sources of hydraulic power for gun loading system components. The lower accumulator supplies power to the lower structure, and the upper accumulator supplies power to upper structure components. The upper and lower systems each employ both main and auxiliary motor/pump assemblies to produce hydraulic power. The main pumps provide full power for the automatic loading system, while the auxiliary pumps provide only limited power for emergency or check-out operations in step mode.

Loader Drum. The loader drum, located below the stand and on the mount's vertical centerline, supports 25 ammunition clips. Each clip stores three rounds of ammunition and is positioned in a clip track on the drum. These clip tracks mate with transfer drive tracks and strikedown station tracks to guide the clips from the transfer station to the upper hoist or to the strikedown hoist. The upper hoist is on the vertical centerline of the mount. The strikedown hoist is outboard of the drum.

Hoist. The hoist receives individual rounds of ammunition from the transfer station and raises them into the cradle on the rotating structure.

Cradle. A cradle moves single rounds of ammunition from the hoist to the slide in alignment with the barrel. When aligned with the hoist, the cradle, being on the centerline of mount rotation, can receive rounds while it rotates with the upper structure. After the cradle raises and is latched to the slide, the rammer, which is mounted on the slide, rams the round into the breech.

Oscillating Assembly. The oscillating assembly is supported on, and pivots about, the trunnion supports and is made up of:

Breechblock. The breechblock is a vertical slide type containing the firing mechanism. The breechblock operating mechanism is mounted on top of the gun barrel housing. When the breechblock is closed, the firing circuit may be closed. The counter recoil action of the gun after firing activates the breechblock operating mechanism to open the breech.

Extractor. The empty case extractor mechanism, mounted on top of the gun barrel housing, has two extractor arms extending into extractor pockets in the breech face. When the breechblock rises (opening the breech), it actuates the extractor mechanism which kicks the empty case out of the breech and into the empty case tray.

Gas Ejector. After each firing, compressed air is discharged into the gun barrel to expel residual propellant gas remaining in the bore. The gas ejection system is activated by the breechblock at the end of the opening cycle.

Empty Case Tray. The empty case tray is a cylindrical tube pivoted at one side by an attachment to the slide. In the down position, it locks to the gun housing and recoils on a splined shaft driven by the empty case tray drive. The tray receives the empty case extracted from the breech and raises to align the empty case tray with the empty case ejector.

Empty Case Ejector. The empty case ejector is a tube mounted on top of the slide, along with an associated chain track assembly, which expels empty cases from the mount. The chain and attached pawl operate when an expended case in the empty case in the empty case tray is in alignment with the tube.

Recoil. Firing the gun causes recoil action in which the gun barrel housing slides backward in the slide on the keys supporting the housing. Recoil cylinders, machined in the gun barrel housing, control the deceleration of the recoiling components.

Counterrecoil. Two counterrecoil cylinders are mounted on the inboard sides of the slide walls in back of the gun barrel housing. They are nitrogen charged cylinders that help decelerate recoil. They also return and hold the gun in battery until the next firing.

Gun Barrel. The gun barrel fits into the gun barrel housing, and is secured by four segments of interrupted threads and a barrel locking key. The barrel is a two piece unit -a tube and a liner.

Gun Mount Control. The appropriate control mode of the gun mount is selected through switching at the control panel. Gun mount control is the combined control of the gun loading system and the gun laying (train and elevation) system to position the gun, and to load the breech with ammunition. The gun loading system may be controlled in automatic or step; the train and elevation systems in remote or local.

Automatic Control. Automatic control is the normal mode for the gun loading system. In this mode, the components of the gun loading system operate automatically in a synchronized sequence, and load orders initiate the operation of the gun's loading system.

Step Control. Step control is used during maintenance or testing, or if automatic control is inoperative. In step control, individual components of the gun loading system are operated by switches on the EP-2 control panel.

Remote Control. The train and elevation systems can be operated in remote control. In this mode, the mount is automatically positioned by gun orders generated by the fire control system or at NWL by dummy directors.

Local Control. In this mode the operation of the train and elevation systems are controlled from the EP-3 test panel for testing, checking and emergency purposes.

#### Control Panels. The control panels in the system are:

The Power Panel (EP-1), Figure 4 of Appendix A, contains circuit breakers and rotary switches used by the gun captain to distribute power to the gun mount motors, control circuits, and indicator light circuits.

The Control Panel (EP-2), Figure 5 of Appendix A, is the control panel for the gun mount. This panel has the electrical equipment for selecting, controlling and monitoring all phases of gun mount activity. Firing can be initiated at EP-2 by closing a firing key on the panel.

The Test Panel (EP-3), Figure 6 of Appendix A, is used to test and check the train and elevation systems. Test equipment can be plugged into the panel for checking the gun laying system and an operator can train and elevate the gun using the local control station at EP-3 panel.

Electrical Power. 440 volt, 60 Hertz, three-phase power is the main supply powering the train and elevation systems, the main and auxiliary loading system motors, the anti-icing and the ventilation systems. Also, the 440 volt power is transformed in the EP-1 panel to 115 volt, 60 Hertz, single phase power for the lighting, alarms, direct current conversion, power contactors, firing circuit relays, and the train and elevation test circuits synchros. A 115 volt, 400 Hertz, single-phase power line supplies the train and elevation synchros and the electronic servo control units. A separate 115 volt, 60 Hertz, single phase source is used for the firing circuit transformer, but a DC source is used instead of the transformer at NWL.

Auxiliary Equipment. The auxiliary equipment of the mount includes:

Ventilation System. The ventilation system expels hot air from the gun house through the base ring into the loader room. A heat exchanger is built into the ventilation system to cool hydraulic fluid of the upper system.

Cable Reel. The cable reel assembly is composed of two grooved reels—one at the stationary center section of the loader drum and one suspended from the carriage—that feed power supply lines from the ship into the rotating structure. The power supply lines (electric and pneumatic) are anchored at the carriage to provide sufficient slack and to prevent tangling.

Anti-Icing System. An anti-icing system to ensure normal operating of the mount under freezing weather conditions is provided. The base ring water shield, the gun port shield and the empty case ejector door are anti-iced. The system consists of a fluid heating and supply tank, fluid circulating motor pump, and a piping system. The system is charged with an ethyleneglycol-water mixture.

Ammunition Strikedown System. A strikedown hoist is installed in the NWL mount outboard of the loader drum. Carts for placing cartridges and projectiles on the hoist pawl were provided. (See Figures 7 and 8 of Appendix A.) This system is not representative of planned shipbcard installations.

#### C. Installation

The general arrangement of the mount as installed at NWL can be seen in Figure 9 of Appendix A. The mount is operated from the control shelter located directly aft of the mount. A 5"/54 Caliber MARK 42 MOD 4 Mount is installed at the left of the 8"/55 mount, and a 5"/54 Caliber MARK 45 MOD 0 Mount is at the extreme left. The van trailers in the lower left house radar instrumentation and blast measuring equipment. Figure 10 of Appendix A shows the mount during installation of the oscillating assembly after conversion to 8-inch. The view is from the port side and the elevation power drive can be seen in the lower part of the picture. Figure 11 of Appendix A shows the mount installation from the front after decking and a water tank used to collect cases during firing have been installed. An X-band doppler radar can be seen at the left of the mount. An overall view showing the loading of charges and projectiles is given by Figure 12 of Appendix A, and Figure 13 of Appendix A is a view of the mount firing.

#### III. DESCRIPTION OF TEST EQUIPMENT

#### A. Train and Elevation Power Drive Tests

Power drive tests were conducted using the following equipment supplied with the mount:

Limiter and demodulator model G 2120897 Order signal generator 2626237 Recorder MARK 220 2860047

#### B. Loading System Tests

Event cycle times were obtained using the following equipment supplied with the mount:

Limiter and demodulator model G 2120897
Recorder MARK 220 2860047
Tachometer and synchro assembly LM 2626222
Connecting test cable T14159
Oscillograph hookup cable 2626238
Timer M6X3 on the inside of the back door of the EP-2 panel

Times were recorded on the recorder by positioning the timer to the function being performed by the mount (such as cradle raise) and measuring time between pips (indications of switch actuations) on the record. A back-up time was obtained from the timer reading. Position vs time, velocity vs time and acceleration vs time data were obtained from the recorder traces, with the tachometer and synchro assembly providing the input. Brackets and gearing are included on the mount at several locations for mounting the assembly. Procedures are given in the Maintenance Requirement Cards (MR cards) and calibrations were provided by NOD/fmc.

#### C. Electrical Power Consumption Tests

Power consumption was measured using the following equipment:

Halltron power computer model PC-2K4-3 including two model CT 2000L current transformers for measuring the 440-volt, 3 phase power.

Halltron power computer model PC-51 (400 Hz, 115 VAC) for measuring the 110-volt, 400 Hz, single phase power.

Newport model 60 differential amplifier with low pass filter

Sangamo Electric Co. Model 3600 14 channel tape recorder with wide band group II electronics.

Incor galvonometer driving amplifier.

Honeywell model 1508 24 channel oscillograph with model N-3300 galvonometer.

Portable cassette audio recorder.

The power computers were supplied by Ohio Semitronics. Inc. of Columbus, Ohio. They are precalibrated devices utilizing the "Hall Effect" in a multiplier circuit to obtain the true product of voltage and current in an AC circuit. The output is VDC proportional to the true power consummed by the load (El  $\cos \theta$ ). The power computer transformers were placed on the power lines supplying the EP-1 panel. The tape recorder provided a record of real power consummed, and an oscillograph was used for reproduction.

#### D. Firing Tests

All chamber pressures taken during firing were obtained using 1/30 in<sup>2</sup> area copper crusher disks in NWL Type D gauges. The disk lots were calibrated against measurements taken with piezoelectric gauges in 5"/54 gun firings.

Velocities were obtained using a chronograph with coils suspended in front of the gun with the projectiles magnetized, an X-band doppler radar system and/or a C-band doppler radar system. X-band doppler data were considered the primary system.

Equipment used with the coil velocity system was:

Coils (fabricated at NWL)
AUL Instrument Co. Model 1000 shaping adapter
Naniloa Model 282 time interval counter
Sangamo Electric Co. Model 4700 14 channel tape recorder

The counter was used for immediate read-out while the tape recorder gave a permanent record of flight time between coils.

Equipment used with doppler systems was:

X-Band Radar of special manufacture C-Band Radar of special manufacture Wave Tech model 112 frequency oscillator Sangamo Electric Co. Model 3500 tape recorder General Radio variable filter Electrac model 215T filter Hewlett Packard model 5248 frequency meter Shaping amplifier (NWL fabricated) 3 Darcey model 361 A-R doppler counters 50 ms shaper (NWL fabricated) Tektronics model 184 timer mark generator Storage register (NWL fabricated) Franklin model 3200 digital printer Honeywell model 1508 duect print oscillograph Newport model 60 data amplifier Endevco model 4401 conditioning unit

Figure 14 of Appendix A is a schematic giving the arrangement of this equipment. All of this doppler equipment was located in a van in the foreground of Figure 9 of Appendix A except for the radars which were positioned to the right of the mount (X-band) and to the left-rear of the mount (C-band). Figure 15 of Appendix A shows the C-band radar in position.

Intensity of muzzle flash was measured using:

EG&G photodiode model SGD-444 Wide band-pass optical filter 1L/WB-510 Newport model 60 data amplifier CEC oscillograph model 5-133 with a 7-362 galvanometer

The output of the detector is amplified and matched to the galvanometer in the oscillograph recorder through the amplifier.

Ranges of falls of shot were obtained by observers manning four range stations along the Virginia river shore, each provided with a theodolite to obtain the sight angle of the splash. The gun and the station are at known locations and the theodolites were referenced on known landmarks. A computer program was designed to select the median of the intersections of lines of sight from each station as the fall of shot and, given the gun train, to compute the range and drift from the line of fire. Meteorological data was obtained on the surface using local equipment and

at altitude from the Naval Air Station, Patuxent River. A Beukers system at NWL provided back-up data. Knowing gun trunnion height, tide level, and meteorological conditions, range table ranges could be computed using NAVORD OP 1041 and OP 1333. These ranging methods and computations were in accordance with current general NWL procedures.

Mount data was obtained using the following transducers:

Recoil pressure - Dynisco PT-81A, 5000 psi strain gauge - bridge type pressure transducer

Counterrecoil pressure - Dynisco PT-81A, 3000 psi

Strain measurements – Miro-Measurements type EA-06-250BF-350W single arm strain gauges.

Recoil displacements – NWL fabricated rack and pinion assembly driving a 10-turn potentiometer (Spectral model 510).

Small displacements of liner and breechblock with respect to the housing – Bournes Linipot model 2001416014-.44 (2000 ohms at .44 inches full displacement)

Timing - Tektronix time mark generator Model 184.

Transducer signals were applied to Endevco model 4470 signal conditioning units with appropriate plug-in mode cards and Newport Model 60 data amplifiers. All data were recorded on magnetic tape (Sangamo model 4700 or 3600 recorder) and reproduced on a Honeywell model 1508 oscillogram using M-3300 galvonometers.

#### E. Gun Jump Test

An attempt was made to study gun jump using three Mitchell 35mm cameras and an horizontal and vertical scale attached to a velocity coil.

#### F. Cook-Off Test

An empty projectile was instrumented with iron constantan thermocouples with the leads fed through a pipe fitted to the base plug. An empty case with a brass plug with a hole for the pipe was used for loading and ramming the projectile. Thermocouple outputs were recorded on a Honeywell model 153X62P16-X-SONG recorder.

#### G. Gun Blast Test

Gun blast pressure measurements were made using the following equipment:

Crystal Research Inc. piezoelectric pressure transducers Endevco charge amplifiers Honeywell model 1612 direct print oscillograph Sangamo model 3500 tape recorder Tektronix model 184 time mark generator

The transducer output was amplified and recorded with the timing pulser on both the oscillograph, for direct readout, and the tape recorder for the final record.

#### H. Smoke and Fumes Test

Equipment used in detecting toxic gases in the mount was:

American Instrument Co. 100cc/min peristaltic pumps
Mine Safety Appliances Co. (MSA) hand operated grab samplers
MSA carbon monoxide colorimeter indicator tubes (0-1000 ppm range)
Kitagawa Co. carbon monoxide length of stain indicator tubes
(0-3000 ppm range)
MSA Monitaire portable sampler (for lead detection)
MSA Lira model 300 dual beam infrared carbon monoxide analyzer
Newport model 60 data amplifiers
Sangamo Electric Co. Model 4700 or 3600 tape recorder
Honeywell model 1508 oscillograph using galvonometer M-3300

Sampling via the indicator tubes was governed by a sequence timer and solenoid valve arrangement. The continuous infrared analyzer output was amplified and recorded on the tape recorder. Recorded data was then reproduced on an oscillograph.

#### IV. TEST PROCEDURE

#### A. Train and Elevation Power Drive Tests

Train and elevation power drive tests were conducted using the order signal generator as the remote order source and the test panel (EP-3) as the local order source (shown in Figure 16 of Appendix A). Most testing was done in remote control. The limiter-demodulator and the double channel recorder were used to record error (difference between the B-end drive position and the ordered position) vs time, velocity vs time, or position vs time. The order signal generator provided the order signals to the stators of the 1X and 36X control transformer synchros in the (train or elevation) receiver-regulator.

Error Trace. Beend response error trace voltage was obtained from the 36X test control transformer synchro in the receiver-regulator. The rotor of this synchro is geared to the Beend response. The stators were electrically connected to the 36X control transmitter synchro stator in the EP-3 panel for local control, or to the order signal generator which electronically simulates the control transmitter synchro stator output in remote control. Rotor output voltage of this test synchro was applied through the test circuitry to the limiter-demodulator unit and then to the recorder.

Position Trace. Bend position trace voltage was obtained like the error trace voltage except that the test synchro stator was connected to a reference synchro which was kept in a fixed position. The order signal generator provided the order signal to the control transformer synchro stators, the 36X control transmitter synchro stator, and the test synchro rotor, mechanically linked to the B-end, provided the B-end position voltage to the limiter-demodulator and recorder.

Velocity Trace. B-end velocity trace voltage was obtained from the DC techometer generator located in each receiver regulator. The tachometer generator is directly geared to the regulator B-end response gearing and furnishes a voltage proportional to B-end velocity. This voltage was applied through test circuitry to the limiter-demodulator and recorder.

Input Signals. The Order Signal Generator provided the several types of orders for the various power drive tests: static orders, constant velocity orders or SHM orders. Instantaneous switching from a static order to another static order was accomplished to obtain maximum accelerations, velocities and synchronizing times. Automatic switching where the operator switched from a static order to a constant velocity order at certain points enabled the mount to synchronize to a variable order with a known initial error.

#### B. Loading System Tests

Loading system cycle times were taken following detailed procedures given in MR card C-1. The tachometer-synchro assembly was coupled to the component cycled by attaching it to permanent brackets and gearing on the mount. Rate and position vs time records were taken, with appropriate ratios relating component movement to synchro rpm supplied by NOD/fmc. Switch action which initiated and terminated each cycle was superimposed as pips along the time axis. Traces with an empty mount were taken on 24 and 25 September 1971, and traces of cycles with a loaded drum, clips. cradle, and rammer were taken later at convenient times during the evaluation. Empty case ejector cycle times could not be obtained because of a bent mounting bracket for the tachometer-synchro assembly.

#### C. Ammunition Selectability

The ammunition selectability test was conducted by loading only one cell per clip (25 full charge rounds) since each clip can contain only one type of round. All round assignment switches were positioned at PD fuze and standard charge. Clip assignment switches were positioned as follows:

Clip	Cell Loaded	Type Assigned
1	1	1
2	2	1
1 2 3 4 5 6 7 8	2 3 2 2 3	į
4	2	2 5 2 6
5	2	5
6	3	2
7	1	6
8	1	2
	2 3	2
10	3	2
11	1	3
12	1	2 2 2 3 3 3 3
13	3	3
14	1	3
15	3	4
16	1	4
17	2	4
18	2 3 2	4
19	2	3
20	1	
21	1	! 5
22	3	6
23	1	6
24	2	6
25	2 3	6

The gun was to be fired in continuous load with the operator selecting the following in sequence:

Type Selected	Number of Rounds Fired
2	5
5	2
1	4
3	5
6	5
4	_4
	25 Rounds Total

The actual firing sequence was modified somewhat as explained in the Results and Discussion. (MARK 18 MOD 9 projectiles (obsolete) were used for this test to save on test costs.) The windshield of round ten broke, damaging the velocity coils and velocities were taken only for the first nine rounds. Other instrumentation for this test included recoil and counterrecoil pressures and durations, recoil displacement, ejection time and interval between rounds.

It should be noted that in order to accomplish this test, a minor wiring change had to be made in the EP-2 panel to eliminate the long round capability (since no long rounds were available during the evaluation). Normally types 1 and 3 are long rounds and the hoist would have to shift to the long flight to load.

#### D. Electrica! Power Consumption

The 440 VAC-60Hz power and the 110 VAC-400Hz synchro power supplied to the EP-1 panel were monitored while various components of the mount were operated. Full mount operation with the anti-icing system heater and pump running was attempted but could not be accomplished because cold weather conditions were not severe enough.

#### E. Misfire Procedures

A check of misfire procedures, except for aspects covered by the safety analysis, consisted merely of checking the ability of the mount to unload a charge and to load a type 6 charge without a projectile. No attempt was made to use the misfire tray because of known deficiencies in the tray.

#### F. Regunning Procedures

Regunning was performed by removing the entire EX 28 MOD 0 liner and tube assembly, removing the liner from the tube on stationary supports on the ground, installing the EX 28 MOD 1 liner in the tube, and installing the assembly in the mount. This procedure was used in lieu of the standard procedure of replacing the liner with the tube still in the mount because of expected difficulty and potential damage in removing the liner, such as had occurred with 175mm assemblies of this design. On 20 April, after the Technical Evaluation, the EX 28 MOD 1 liner was removed from the tube in the mount, and then reinstalled. No serious problems or damage occurred.

#### G. Firing Tests

For most firing tests, the mount was instrumented for obtaining recoil pressure, counterrecoil pressure, and recoil displacement vs time data. Data were recorded on a common time base and referenced to the application of firing current for each round. Indications of projectile exit from the muzzle were provided by using strain gauges on the muzzle. For rapid fire programs, the recorders were run continuously so the time between applications of firing current could be obtained to get rate of fire.

All rounds fired with pressure gauges were fired in step load removing the empty powder cases by hand to prevent losing the gauges in the mount.

Powder case separation necessitated heat treatment of cases. All cases fired on and after 12 November 1971 were dipped in a sulphuric acid gun barrel decoppering solution for cleaning, and remaining dirt and corrosion were removed by hand prior to loading. All MARK 25 projectiles were loaded with cement to 260 lbs. Cartridges and projectiles fired on and after 12 November were wiped with a degreaser prior to loading in the mount.

#### H. Projectile Cook-Of?

Attempts were made to check the possibility of a projectile cook-off in a hot gun by ramming an instrumented projectile as soon as possible after the last round was fired interfering as little as possible with other test requirements. After the last round of a test was fired the instrumented projectile and its cartridge case were loaded at the strikedown hoist into the drum, run through the loading system and rammed. The case was removed by hand and the thermocouple leads connected. The location of the thermocouples in the projectile were as follows:

1/2-inch aft of the nose plug on the projectile inner wall.

14 inches forward of the projectile base on the projectile inner wall.

1/2-inch forward of the front of the base plug on the projectile inner wall.

Imbedded in the rotating band at the maximum diameter.

Imbedded in the rotating band at the maximum diameter diametrically opposite the above thermocouple.

On the front of the base plug.

#### I. Reduced Charge

A reduced charge round of 67.00 lbs. of NACO propellant, SPCF 11073 was used early in the testing of the 8"/55 MCLGM. This charge was estimated based on testing conducted in an 8"/51 barrel in a test girder to give the desired reduced charge velocity of 2220 ft/sec. This charge was revised to 66.07 lbs. based on results of probing rounds fired from the EX 28 MOD 1 liner on 4 November and was used during the OPEVAL.

#### J. Gun Blast

Initial blast pressure measurements were made during three series of five single full charge rounds fired on 4, 6 and 7 October 1971. Twelve blast pressure transducers were arranged at various positions in the horizontal plane of the gun muzzle as shown in Figure 17 of Appendix A. The transducers were placed edge-on to the shock front in a horizontal attitude. Figures 18 and 19 of Appendix A show the gauges arranged in front of the mount.

Additional testing was done during the rapid fire test of 12 November using one piezoelectric transducer positioned in the horizontal plane, 17 feet from the gun muzzle along a radial line 60° from the bore axis. The transducer was in a plane with the gun if it would have been at 0°30′ elevation. The test was fired at 10° elevation. During the firing of 13 December 1971 (actually during the OPEVAL test), two piezoelectric transducers were used. One was located 30 feet from the muzzle along a radial line 150° from the bore axis and the other was at 20 feet on a 90° radial. Both again were in a horizontal, gun bore-axis plane with the gun at 0°30′ elevation. The firings were conducted at 20° elevation. The first transducer was about 11 feet below the horizontal, bore-axis plane and the second was about 8 feet below.

#### K. Smoke and Fumes

Smoke and fumes were monitored during testing on 12, 22, 23 and 30 November and 3 and 13 December 1971. Visual observations were used as a basis for qualitative observations on smoke buildup, and the infrared analyzer allowed continuous monitoring for carbon monoxide. The colorimetric indicator tubes were also used in conjunction with pumps and solenoid valves operated periodically by timers during firings to measure carbon monoxide concentrations at locations other than those monitored by the infrared analyzer. For the tests of 12, 22 and 23 November, the analyzer sampling tube was located above the deck on top of the loader drum at the rear of the mount under the left trunnion support. For the tests on and after 30 November, the tube was located at the top of the shield over the breech. Extensive amounts of air from above the loader drum were also drawn through two MSA "Monitaire" units to detect the presence of lead. Additionally, NOS, Indian Head personnel circulated large quantities of gas from the breech area through a Dewar apparatus cooled by liquid nitrogen to observe directly for the presence of any oxides of nitrogen. NOS, Indian Head personnel also obtained gas samples from the same area for subsequent detailed laboratory analysis of all possible toxic constituents.

During all firings, the ventilation system was operating and all doors were closed. Doors were opened when stoppages occurred due to mount casualties and after completion of the firings.

#### L. Anti-Icing System

The anti-icing system test was conducted on 8 February 1972 between 0500 and 0900 hours EST when the ambient temperature ranged between 16° and 18°F. The mount was sprayed periodically with water around the gun port shield and water seal. The mount was checked for operability in train and elevation and anti-icing fluid temperature was recorded every half-hour.

#### V. RESULTS AND DISCUSSION

#### A. Train and Elevation Power Drive Tests

Train and elevation drive tests were accomplished 20-22 October 1971. No problems were encountered except that with the dither pot, described in Appendix B, which was temporarily repaired on 21 October and permenantly repaired on 23 October. Results of these tests are presented in Appendices D and E. All data in Tables D-1 of Appendix D and E-1 of Appendix E indicate performance meets or exceeds the requirements of Reference 7 and is satisfactory. The recorder trace of each test is included in Appendices D and E for reference and record purposes.

#### B. Ammunition Loading System

Loading system tests were conducted on 24-25 September 1971, except that empty case ejector traces could not be taken because a bracket for mounting the tachometer-synchro assembly was bent in shipment or during installation of the slide after conversion. Results are presented in Table F-1 of Appendix F. The recorder traces are also included in Figures 1 through 24.

#### C. Ammunition Selectability

Rounds were loaded for the selectability test in the morning of 27 September 1971, and fired that afternoon. Test results are tabulated in Appendix G (ballistic results) and Table 3 (actual firing order used). A problem with the cradle loaded indications caused by photocell sensitivity caused a stoppage after round 11 (a problem listed in Appendix B). Bulbs were replaced and voltages were adjusted, rendering the mount operable again. The test was resumed, skipping the first Type 3 round which was in clip 11 (round 12 of the firing order specified in the procedure). This round was picked up, then, as the last Type 3 round (round 16).

It can be determined from Table 3 that for each clip the drum indexes, approximately 2.3 seconds are added to the interval between rounds. The longest interval between rounds was 43.0 seconds between rounds 15 and 16 when the drum had to index from clip 19 to clip 11.

Because the mount fired only those rounds selected by the operator and functioned as designed and described in OP 4116, results of this test are deemed to be satisfactory.

TABLE 3 FIRING ORDER FOR SELECTABILITY TEST OF 9-27-71

Round No.	Round Type	Clip No.	Cell No.	Required Indexing (No. of Clips)	Required Cell Transfers	Time Between Rounds (sec)
1	2	4	2	0	Rd in Breech at 7-0	T-0
2	2	6	3	0	Rd in hoist at 7-0	4.8
3	2	8	1	2	нт, тн	6.1
4	2	9	2	I	2H, 3H, HT, TH, 2H	7.4
5	2	10	3	I	3H, HT, TH, 2H, 3H	7.4
6	5	21	1	11	НТ, ТН	28.2
7	5	5	2	9	2H, 3H, HT, TII, 2H	26.1
8	I	20	I	15	3H, HT, TH	38.1
9	1	1	1	6	2H, 3H, HT, TH	18.3
10	1	2	1	1	2H, 3H, HT, TH	7.5
11	1	3	3	1	2H, 3H, HT, TH, 2H, 3H	7.3
			Stoppage d	ue to casualty -		
12	3	12	1	0	Rd in breech at T-0	T-0
13	3	13	3	0	Rd in hoist at T-0	4.8
14	3	14	1	1	нт, ти	5.2
15	3	19	2	5	2H, 3H, IIT, TII, 2H	16.6
16	3	11	1	17	3H, HT, TH	43.0
17	6	22	3	<b>I</b> 1	2H, 3H, HT, TH, 2H, 3H	31.4
18	6	23	1	1	нт, тн	5.1
19	6	24	2	1	2H, 3H, TH, HT, 2H	7.3
20	6	25	3	1	3H, TH, HT, 2H, 3H	7.2
21	6	7	1	7	нт, тн	18.8
22	4	15	3	8	2H, 3H, HT, TH, 2H, 3H	24.5
23	4	16	1	1	нт, т <i>н</i>	5.1
24	4	17	2	1	2H, 3H, HT, TH, 2H	7.4
25	4	18	3	i	3H, HT, TH, 2H, 3H	7.4

Hoist position.
Transfer position
Clip transferred from Cell 3 at Hoist position to Transfer position.
Clip transferred from the Transfer position to the Cell 1 at the Hoist position.
Clip transferred from the Cell 1 at Hoist to Cell 2 at Hoist.
Clip transferred from Cell 2 at Hoist to Cell 3 at Hoist. T HT TH 2H 3H

#### D. Electrical Power Consumption

Electrical power tests were conducted on 14 and 17 January 1972 after completion of the OPEVAL. The test could not be conducted prior to the OPEVAL because of transducer procurement lead time. Results of the tests are in Appendix H, with measurements listed in Table H-1. Reproductions of the power vs time analog data for each test are included as Figures 1 through 12 of Appendix H. Probably the most significant result is the peak true power demand of the mount of 637 kw, 599 kw peak without the anti-icing system and 38 kw for the anti-icing system. Theoretical peak running load power was computed to be 679 kw.

#### E. Misfire Procedures

Misfires occurred during the TECHEVAL due to primer blowback and were purposely caused during the OPEVAL by loading empty charges. Procedure 1 of OP 4116 specifies remote unloading and ejecting the charge and loading of a Type 6 clearing charge. Procedure 2 specifies unloading the charge out the rear of the mount using a misfire tray. Procedure 1 was generally adequate. Procedure 2 could not be checked because of the non-availability of an adequate misfire tray. Reference 8, the safety analysis report and Reference 6, the OPEVAL report treat this subject in more detail.

#### F. Regunning Procedures

Regunning was performed once during the technical evaluation when the liner for the EX 28 MOD 0 gun barrel assembly was replaced with that of the EX 28 MOD 1 assembly on 9 October 1971. After the operational evaluation, during the weeks of 17 and 24 April 1972, the liner for the EX 28 MOD 1 was removed and reinstalled, the EX 28 MOD 1 barrel assembly was removed, the EX 30 MOD 0 assembly was installed, and the liner for this barrel assembly was removed and reinstalled. Also, during the week of 8 May 1972, the EX 30 MOD 0 barrel assembly was removed and the EX 28 MOD 1 was installed. On 31 May, the liner for the EX 28 MOD 1 was removed and the liner for the EX 28 MOD 0 (after modification for a larger chamber) was installed.

In all cases, the liner was removed from the tube without excessive force. Dow Corning Molycote Type G lubricant was used, and seizing and scoring such as occurred with the 175mm liner did not occur. A wooden platform was constructed to give working space for men working at the barrel, icebreaker and bushing.

Raising the breechblock for access to the upper key for liner removal involves using a chain hoist without much overhead clearance, and may involve some trial and error until personnel become experienced.

Upon the last regunning (liner replacement), a five-man weight handling crew and a two-man gun crew performed the entire operation (excluding gathering of tools, etc.) in two hours and thirty-seven minutes. Several errors were made in the operation, and, especially if another man was added to the gun crew, this time could be reduced to less than two hours. Regunning by replacement of the liner-tube assembly was accomplished by the same number of people in about six hours. Differences in the two regunning procedures require that the slide be backed out of battery and the bushing and icebreaker removed to replace the liner-tube assembly. These operations are not required to replace the liner only, but the breechblock must be hoisted up to gain access to the upper liner locking key. Actual crane time to remove the liner or liner-tube assembly is the same.

#### G. Proof Tests

Proof testing was conducted with the EX 28 MOD 0 liner on 3 September 1971 and with the EX 28 MOD 1 liner on 13 October 1971. Because of limitations on the recoil system of this prototype mount, proof rounds were not fired at elevations higher than 10°. No damage was incurred by the mount, and instrumentation measurements are included in the tables of Appendix G.

#### H. Cartridge Assembly

It was decided to test the mount using MARK 25 260 lb projectiles and full charges which would produce a nominal muzzle velocity of about 2700 f/s with an upper service pressure limit of 56 ksi (or 20.8 tsi(Cu)) using flashless NACO propellant. A reduced charge to give 2220 f/s was also decided upon. The charge determination work is reported in References 9 and 10. Cases used were the EX 1 MOD 1 cases for the 175mm. All charge assemblies were not optimized for this evaluation but used off the shelf components to save time and money.

Partial case separation occurred in early 8"0 firings with service and proof charges. Failures of a similar nature had also occurred in 175mm firings. It was found that after heat treating, the cases did not separate at service pressure, hence, only heat-treated cases were used for full charge firings in rapid fire Since cases were softer after heat treating, they had a tendency to deform. In some instances at proof pressure, cases ruptured at the extractor pockets and expanded into the head space jamming the breech closed.

A problem was encountered during rapid fire in that an occasional round would fail to ram fully. In some cases, this was due to pieces of plug jamming between the case and the chamber. At first, plugs with a very small radius at the leading edge were used and these would clip when sliding over the liner breech face

upon ramming. Plugs of a later design increased this radius and another spring was added under the cradle restrainer to solve this problem. All plugs were gauged as well. On 9 December 1971 a fire-out was attempted but a gun stoppage occurred after round 19 because the case failed to extract. An accumulation of fine soot was noticed in the chamber which had built up from firing heat-treated cases which had a dirty scale on the exterior surface. All cases were then cleaned by dipping in acid, rubbing dirty spots with emory cloth, and degreasing. All projectiles were wiped off with rags dampened with degreaser. All ammunition thus treated extracted satisfactorily.

Several stoppages were caused by primer blowback around the ignition element with both the MARK 37 MOD 2 and the MARK 38 MOD 2 primer used for full and reduced charges, respectively. Primer blowback caused residue around the firing pin, keeping the pin from protruding from the breech block and contacting the primer of the next round.

The propellant used during the TECHEVAL and OPEVAL was an 8"0 bag gun granulation, NACO formulation and was not especially designed for this gun. It should be noticed on Table G-1 of Appendix G that trunnion velocity uniformity was frequently undesireably high, and muzzle flash was frequent.

#### I. Rate of Fire

The rate of fire of the mount is highest, of course, when the same type of ammunition is fired from successive clips in continuous load. The data on Table G-1 of Appendix G indicate that the rate of fire is 11.7 rounds per minute at 0° elevation, increasing to 12.6 rounds per minute at 35° elevation, then decreasing to 12.0 rounds per minute at 60°. Rate of fire for firing in step control was about 3.5 rounds per minute, and the rates of fire for the selectability tests conducted during the TECHEVAL and OPEVAL were between 4 and 5 rounds per minute.

## J. Gun Jump

It was decided not to conduct a test series designed explicitly to determine gun jump since gun jump is related to the ammunition characteristics and range tables as well as the mount. A small experiment was conducted, however, on 2 November 1971, the details and results of which are included in Appendix 1.

## K. Accuracy and Range Tables

Table 4 is an array of data taken from Table G-1 of Appendix G which shows accuracy of each shoot arranged by gun elevation. Acceptable accuracy is a corrected D/R of 0.70%, and from Table 4 it can be seen that on only one occasion out of fifteen for full charges was this exceeded. It was exceeded on two out of five for reduced charges, but one of these was at 60° elevation (the corrected D/R criteria of 0.70% can not be expected to apply at elevations above 45°). In four out of the twenty occasions (including the three where the corrected D/R was greater that 0.70%), the uncorrected D/R was less than the corrected D/R.

I greement with the range table predictions was not good. Mean residuals, differences between corrected range and range table range, varied from -11 yds to 713 yds, showing considerable bias toward overshoot. Only one negative value out of 20 resulted.

#### L. Barrel Life

During the initial checkout, the Technical Evaluation, and the Operational Evaluation of the 8.0 MCLGM, 120 rounds were fired in the EX 28 MOD 0 liner and 473 rounds were fired in the EX 28 MOD 1 liner. Bore enlargement at the origin of bore was only 0.024 inch on the EX 28 MOD 0 barrel and 0.084 inch on the EX 28 MOD 1 barrel. The erosion gauge reading on the EX 28 MOD 1 was 2.75 inches

## M. Velocity Loss

Mean velocities for the service charge with the EX 28 MOD 0 barrel ranged from 2784 f/s to 2708 f/s. Data from the six occasions for which mean velocities are available does not indicate a significant velocity loss slope.

Mean velocities for the service charge with the EX 28 MOD I liner are available from fifteen occasions when four or more rounds were fired, and range from 2711 f/s to 2674 f/s. Linear regression equations were fitted to the mean velocity vs bore enlargement at the origin data and to the mean velocity vs erosion gauge reading data. The equations are:

V = 2707 - 410.4 ( $\Delta$  Do) with a standard deviation of  $\pm 7.5$  f/s

V = 2673 - 5.489 (BEGR) with a standard deviation of  $\pm 9.2$  f/s

TABLE 4 8"/55 MAJOR CALIBER LIGHTWEIGHT GUN MOUNT MARK 71 MOD 0 RANGE ACCURACY DATA

Gun Elevation (degrees)	Range <sup>1</sup> Residual (yards)	Corrected Range Standard Deviation (yards)	Corrected D/R (%)	Date Fired (1971)
		FULL CHARGE		
8	153	59	.32	22 Nov
	92	107	.55	22 Nov
	-11	79	.48	30 Nov
10	79	166	.83	2 Nov
	97	83	.41	9 Nov
	8	83	.44	12 Nov
20	614	143	.51	4 Nov
	392	101	.36	13 Nov
	332	170	.59	13 Nov <sup>2</sup>
	312	148	.53	13 Nov <sup>2</sup>
27	543	116	.35	16 Nov
	580	106	.30	23 Nov
	563	135	.34	1 Dec
	713	131	.43	1 Dec
	541	169	.49	3 Dec
		REDUCED CHARGE		
35	100	105	.40	16 Dec
	114	71	.25	16 Dec
45	503	185	.56	3 Dec
	533	281	.93	3 Dec
60	610	224	.74	22 Dec

<sup>1</sup> Range Residual - The difference between the mean range corrected to range table conditions and the range table range. A negative value indicates an undershoot.

These two groups were fired at 19° 30' elevation, but are included under 20° for purposes of this table.

where

V = velocity in f/s

 $\Delta$  Do = bore enlargement at the origin in inches

BEGR = bore erosion gauge reading in inches

Mean velocities for the reduced charge with the EX 28 MOD 1 liner are available from five occasions and range from 2168 f/s to 2140 f/s. These data are insufficient to establish an approximate velocity loss slope.

# N. Projectile Cook-Off Tests

A projectile instrumented with thermocouples was rammed and seated on three occasions:

- 1. After a 15-round rapid fire burst on 2 November. The projectile was ramined and the thermocouples were connected to the recorder seven minutes after firing the last round.
- 2. After 75 rounds rapid fire (4 bursts) on 12 November. The projectile was rammed and the thermocouples were connected nine minutes after firing the last round.
- 3. After 75 rounds on 13 December (during the OPEVAL) when the projectile was rammed and connected 20 minutes after firing.

Temperatures on the rotating band and on the inside of the projectile were highest on 12 November 1971. Temperatures at the base plug were higher than those at other locations in the projectile, reaching 178°F at 70 minutes after the last round was fired. Temperatures on the rotating band, at the base plug and near the nose plug vs time are plotted in Figure 20 of Appendix A for the round rammed on 12 November 1971.

It was noted that after firing the 75 rounds on 12 November, the barrel chamber was warm, but the hand could be placed against it indefinitely without real discomfort. The temperature on the outside of the barrel about three feet from the muzzle reached 504°F 30 seconds after the last round was fired, however.

## O. Reduced Charge

The reduced charge was defined as a charge which would give a velocity of 2220 f/s using the MARK 25 projectile. Reduced charges were fired in rapid fire

on 13 and 14 September during the Technical Evaluation and firing data are tabulated in Appendix G. The mount operated satisfactorily on both occasions.

# P. Safety Analysis

A safety analysis was conducted on the 8.0 MCLGM and is reported by Reference 8. No disqualifying safety problems were uncovered, although a number of improvements from a safety viewpoint are recommended.

#### O. Blast Tests

Blast test data from the firings of 4, 6 and 7 October 1971 are tabulated in Table J-1 of Appendix J.

It was noticed by operating personnel and observers during various firings, however, that certain rounds had a seemingly greater shock and louder report than others; hence, the measurements of 12 November and 13 December. Since muzzle flash often accentuates blast effects, the purposes of these extra experiments were to examine differences between blast characteristics of flashing rounds and non-flashing rounds (none of the slow fire rounds fired on 4, 6 and 7 October flashed). Examination of the data taken on 12 November (only the data from the first 26 rounds were examined) indicated:

- 1. The average free-air peak pressure of the primary blast was 9.3 psi for the 8 rounds which flashed and 10.3 psi for the 18 non-flashing rounds.
- 2. All but one of the flashing rounds had a double peak, the time between peaks being about 0.4 milliseconds, and the average value of the second peak being 9.0 psi.

For the test of 13 December, the data from a sample of flashing and one non-flashing rounds from the 150°-30 feet transducer was played back over a longer time interval and examined:

- 1. There was no discernible difference in the free-air blast wave between flashing and non-flashing rounds (the average for the flashing rounds was 0.68 psi and that for the non-flashing was 0.65 psi).
- 2. A secondary shock was detected at between 65 and 85 milliseconds after the initial shock for the flashing rounds only, the secondary peak pressure being less than the primary peak and varying between 0.46 and 0.59 psi in value.

The double peak pressures occurring in flashing rounds is not significant because the peaks are too close together in time and there is no effective increase in peak pressure. The accentuated blast effects of flashing rounds could be explained by the delayed secondary blast noted in the test of 13 December.

#### R. Smoke and Fumes

The most significant carbon monoxide measurements were taken on 12 November and 13 December 1971 during 75-round firings. Carbon monoxide concentration versus time for these tests are shown in Figure 21 of Appendix A. Significant amounts of lead or oxides of nitrogen were not detected during any of the firings, but smoke would build up in the mount until the shield doors were opened A waiting period of about 15 minutes, depending on ambient wind conditions, was required to clear the mount.

In summary, the only toxic hazard detected in the mount is from carbon monoxide. Peak transient concentrations of 3000 parts per million (PPM) were detected in the gun house, cradle and loader drum areas. After firing, residual concentrations of above 1000 PPM were observed.

# S. Anti-Icing System Test

Water was sprayed on the mount at intervals during 0500 to 0900 hours on & February 1972 when ambient temperatures ranged from 15° to 18°F. The mount operated satisfactorily in train and elevation when checked at half hour intervals. No ice formed on the gun port shield, the empty case ejector door or the base ring, whereas it did form on other parts of the shield as can be seen in Figures 22 and 23 of Appendix A. Anti-icing fluid temperature started at 42°F and rose to 62°F at 0530, 72°F at 0600, and stayed between 60°F and 76°F until 0900.

#### VI. CONCLUSIONS AND RECOMMENDATIONS

#### A. Conclusions

It can be concluded from the test results that:

- 1. the train and elevation power drives respond satisfactorily to local or remote orders from dummy directors.
- the loading system can sustain rates of fire between 11.7 and 12.6 rounds per minute in auto-load, can be cycled in step-load, and can select and load any of six types of ammunition which have been predesignated.
- 3. misfires can be extracted and ejected through the empty case door, and a clearing charge can be loaded and fired without a man entering the mount.
- 4. the mount can be regunned in less than two hours provided the crew is experienced and the facilities and effort are well coordinated.
- 5. mechanical performance of the cartridge case and the primer, and the ballistics and flash of the propellant are not generally satisfactory. Excessive headspace (static clearance between the base of the case and the breechblock), and linear motion and axial expansion between the barrel and housing under dynamic loads are contributors to the marginal performance of the cartridge case.
- 6. ballistic accuracy of the MARK 25 projectile fired from the mount is satisfactory, although the range tables for both the full and reduced charge are deficient and generally biased so that the projectile will fall longer than predicted (up to 3% longer). This bias is not unique with the 8".0 MCLGM, being related to the projectile and therefore noted in all 8".0 mounts.
- 7. barrel life and velocity loss results are not conclusive due to the limited barrel erosion, however, barrel life to about twenty percent expended is roughly comparable to that of the 8.0 barrel MAR. 16 and acceptable when using NACO propellant.
- 8. some minor deficiencies from a safety viewpoint were uncovered, however none are inherent or disqualifying.

- 9. the gun blast profile for this gun is slightly less in magnitude than that for an 8"/55 caliber bag gun under the conditions tested.
- 10. smoke and carbon monoxide concentrations during firing make it unsafe to circulate the air in the mount to other spaces in a ship or to enter the mount for repairs until after the mount has been properly ventilated.

#### B. Problem Areas

There are several circumstances avoidable only at high cost or by schedule slippage which detracted from both the Technical and Operational Evaluations:

- 1. the prototype MCLGM was originally designed, built and evaluated as a 175mm mount and later converted to 8".0. Certain structural members in the loader drum area will require strengthening in production mounts and the oscillating assembly and carriage modified to accommodate increased projectile weights and firing loads.
- 2. the mount had been subjected to considerable testing as a 175mm prior to conversion to 8".0 (e.g., the loading system was subjected to a 50,000 cycle life test at NOD/fmc) and hence some parts were worn and fatigue failures were more likely to occur than in a new mount.
- 3. an extremely severe and arbitrary schedule prevented the contractor from completing all necessary work prior to commencement of the evaluation. No correspondence from the contractor was received indicating that the installation, conversion and his checkout were completed prior to the Technical Evaluation.
- 4. contractor personnel were required to operate the mount during some of the Technical Evaluation because local personnel had still not completed schooling on the mount.
- 5. problems were encountered throughout the Technical Evaluation with the cartridge assembly.

#### C. Modifications to Mount

It was concluded that the design of certain components of the mount which affected performance adversely during testing should be modified during the evaluation. Design modifications included were:

- 1. the photoelectric detectors were changed by installing light sources opposite the photocells and removing the reflectors.
- 2. the counterrecoil cylinder seals were redesigned.
- 3. a new, stronger cradle restrainer spring arrangement was installed.

#### D. Release to OPEVAL

Based on results of the Technical Evaluation, the performance of the 8"/55 Major Caliber Lightweight Gun Mount MARK 71 MOD 0 was judged to be acceptable and certification for release to OPEVAL was recommended on 12 November 1971 by Reference 11. The Operational Evaluation was begun under the auspices of COMOPTEVFOR on 15 November 1971.

#### E. Recommendations

The following recommendations are offered for the production mount design:

- 1. a fold-away platform is needed on the carriage for maintenance operations on the chamber, breechblock, and firing pin assembly and for access to barrel keys during regunning.
- 2. ladder rungs on the shield are needed and provision should be made for attachment of scaffolding for access to the bushing and icebreaker during regunning.
- 3. breechblock headspace and any clearances allowing aft movement of the breechblock with respect to the housing should be minimized.
- 4. a cartridge development program is needed to eliminate primer blow back, and muzzle flash; increase round-to-round velocity uniformity; and, to develop a reliable case, plug and crimp.
- 5. a time between rounds fired function should be included on the timer in the EP-2 panel, or provided in addition to it, to provide an overall check on the "health" of the loading system.
- 6. a new 8.0 range table is needed for both full and reduced charges for the MARK 25 projectile.

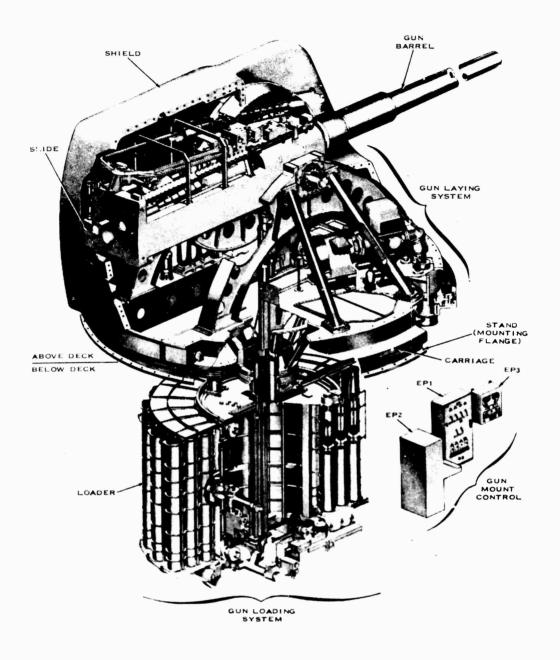
Additional recommendations from a safety viewpoint are provided in Reference 8.

## **REFERENCES**

- 1. CNO Conf ltr OP344G/701E/pnl Ser 0255P70 to CNM of 26 Feb 1965, Subj: SOR No. 12-10
- CNO Conf ltr OP723C/701E/ec Ser 0601P70 to CNM of 1 Oct 1969, Subj: SOR No. 12-10R1
- 3. NAVORD ORDTASK 553-111-090-1-U-1210330 of 22 Mar 1971, Subj: MCLGM 8" Conversion
- 4. NAVORD Conf ltr ORD-5531/456:EJK Ser 03819 to CNO of 12 Aug 1971 Subj: Project Assignment Request for 8"/55 Caliber EX 71 MOD 0 Gun Mount
- 5. NWL Conf ltr TP:DLB/PJO:jca 8300 Ser 0720067 to NAVORD of 19 Jan 1972, Subj: 8"/55 MCLGM EX 71 MOD 0 Technical Evaluation; preliminary report of
- 6. COMOPTEVFOR Conf ltr 722:rg 3930 (0/S173) Ser 0122 to CNO and NAVORD of 20 Mar 1972, Subj: Final Report on Project 0/S 173, "Conduct a Limited Scope Operational Evaluation of the 8"/55 MCLGM EX 71 MOD 0"
- 7. NAVORD Specification XWS-4952
- 8. NWL ltr ESE:JSN:vdc 8000 to NAVORD (ORD-5531) of 5 May 1972, Subj: MARK 71 MOD 0 Eight-Inch Lightweight Gun System; safety evaluation of
- 9. NWL ltr TPB:JLM:pjc 8010/1-21 to NOSIH of 13 Jun 1972, Subj: 8-Inch Major Caliber Lightweight Gun (MCLG) Reduced Charge; establishment of
- 10. NWL ltr TPB:BZJ:scc 8010/1-21 to NOSIH of 22 Jun 1972, Subj: 8-Inch Major Caliber Lightweight Gun Propellant Evaluation; results of
- 11. NWL Conf msg 122230Z Nov 1971
- 12. NWL ltr FVE:MHP:ljb 10552 to NAVORD of 6 Jan 1972, Subj: Electromagnetic Susceptibility Test Plan for 8-Inch 55 Caliber Gun Mount MARK 71 MOD 0; report of

APPENDIX A

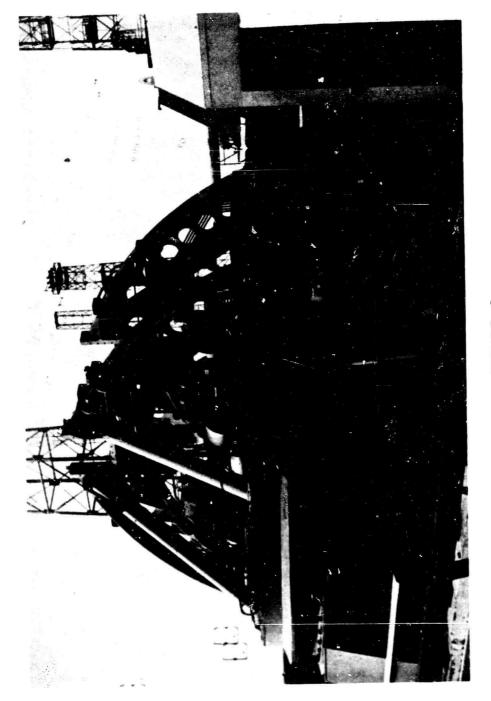
Figures 1 Through 23



PHD-2497-11-72

FIGURE 1

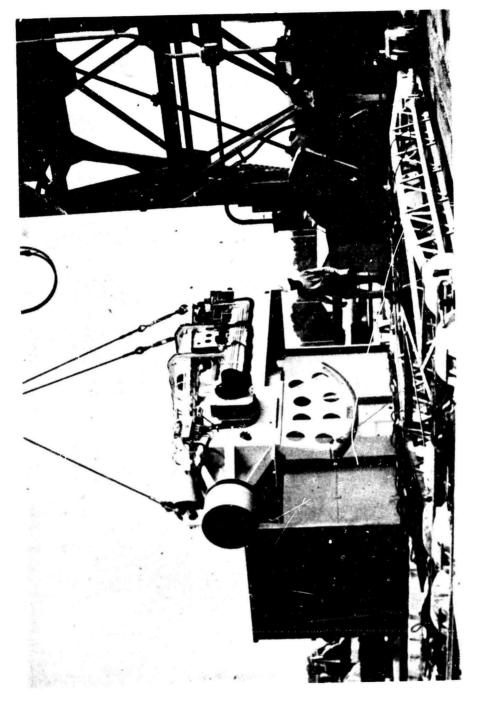
Cut-Away View of the 8"/55 Major Caliber Lightweight Gun Mount MARK 71 MOD 0



PHD-2498-11-72

FIGURE 2

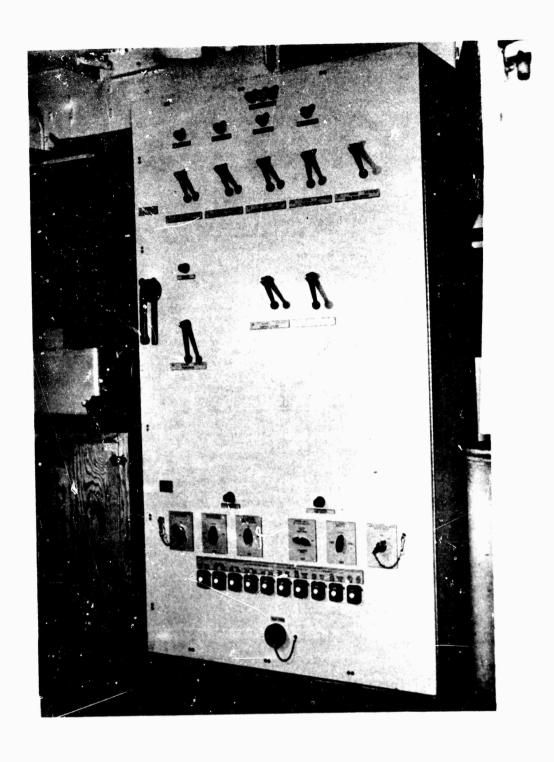
View of the Base Ring and Trunnion Supports of the 8"/55 MCLGM Installed at NWL



PHD:2499-11-72

FIGURE 3

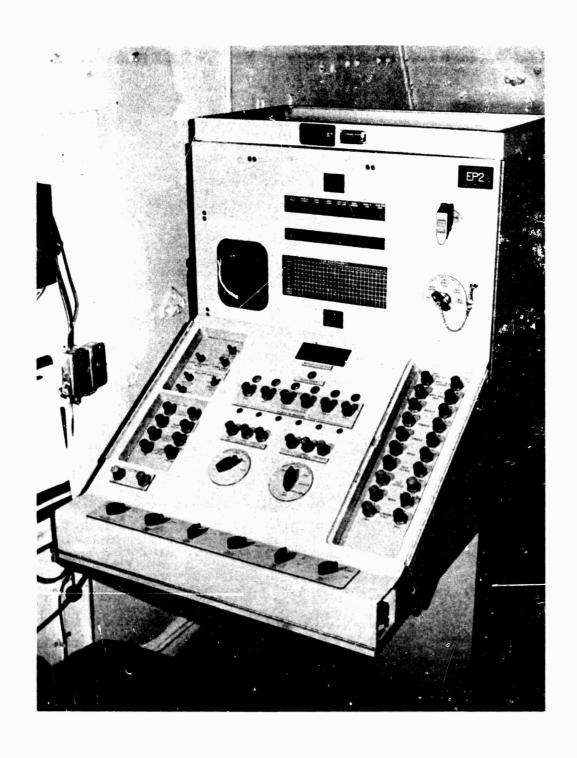
View of the Slide During Assembly of the MCLGM After Conversion to the 8"/55 Caliber



PHD-2500-11-72

FIGURE 4

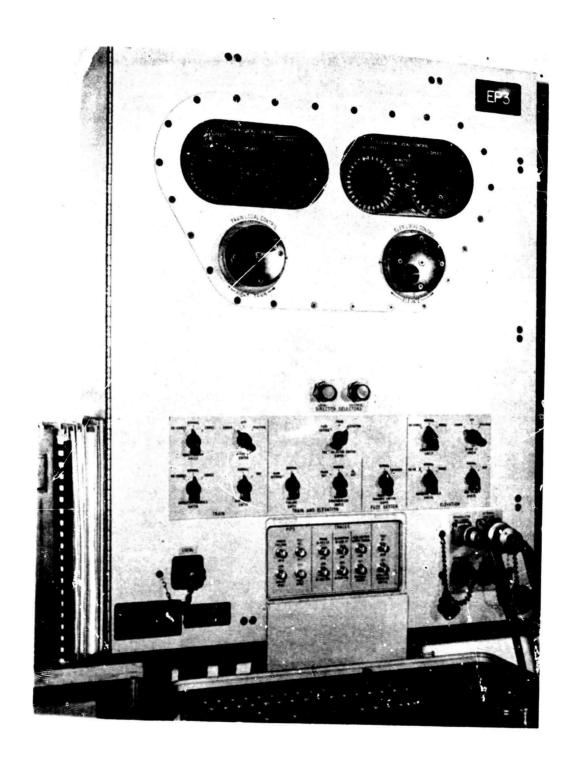
View of the 8"/55 MCLGM EP-1 Panel



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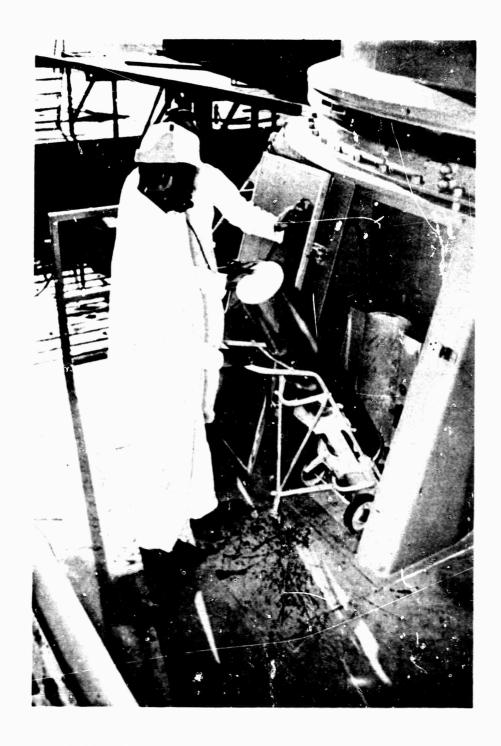
FIGURE 5

View of the 8"/55 MCLGM EP-2 Panel



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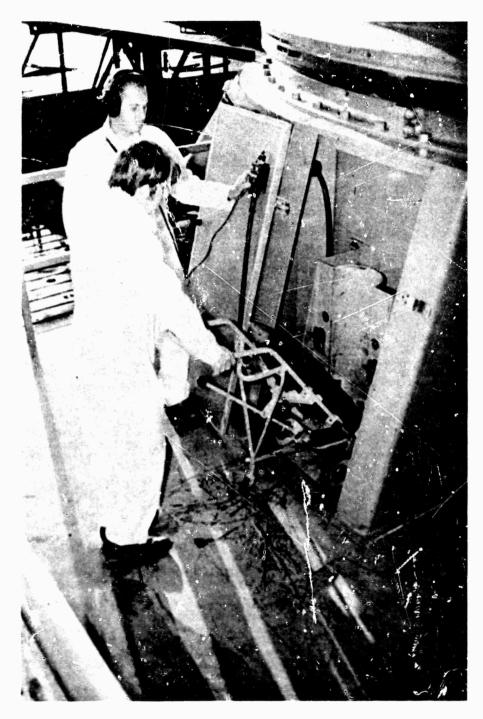
FIGURE 6
View of the 8"/55 MCLGM EP-3 Panel



PHD-2503-11-72

FIGURE 7

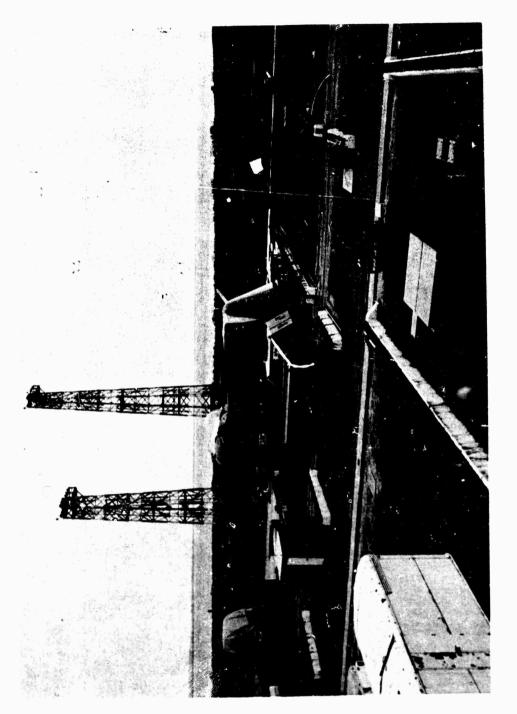
Photograph of a Cartridge Being Placed in the Strikedown Hoist of the 8"/55 MCLGM at NWL



PHD-2504-11-72

FIGURE 8

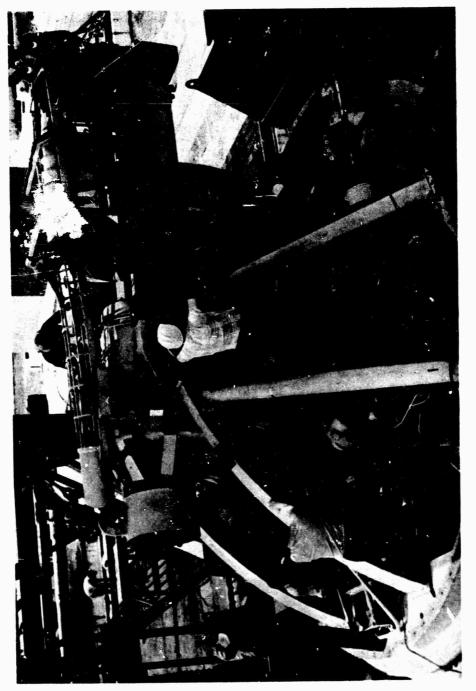
Photograph of a MARK 25 Projectile Being Placed in the Strikedown Hoist of the 8"/55 MCLGM at NWL



PHD-0180-2-72

FIGURE 9

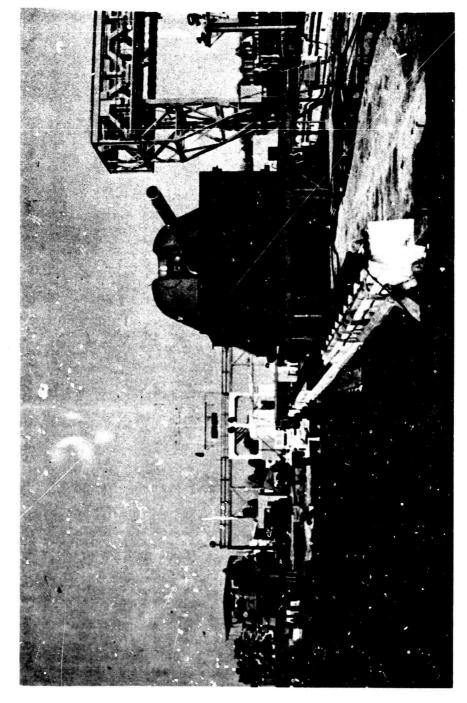
8"/55 MCLGM Installed at NWL as Viewed From Behind



PHD-2505-11-72

FIGURE 10

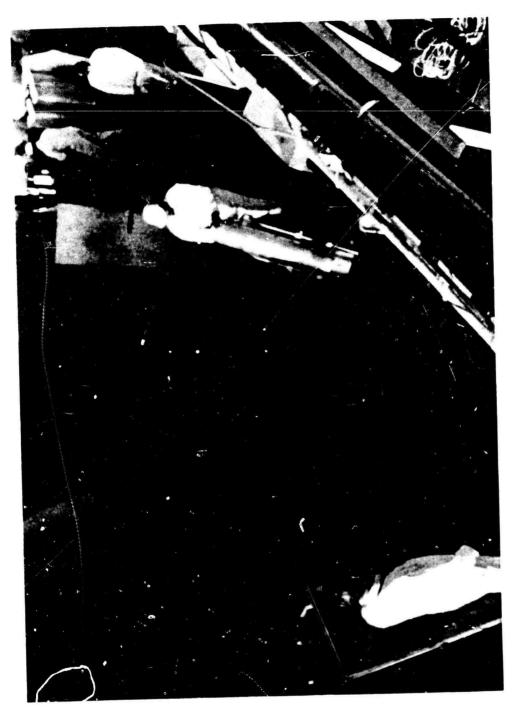
View of the MCLGM During Installation of the Slide After Conversion to 8"/55 Caliber



PHD-2506-11-72

FIGURE 11

8"/55 MCLGM Installation at NWL as Viewed From in Front of the Mount



PHD 2507-11-72

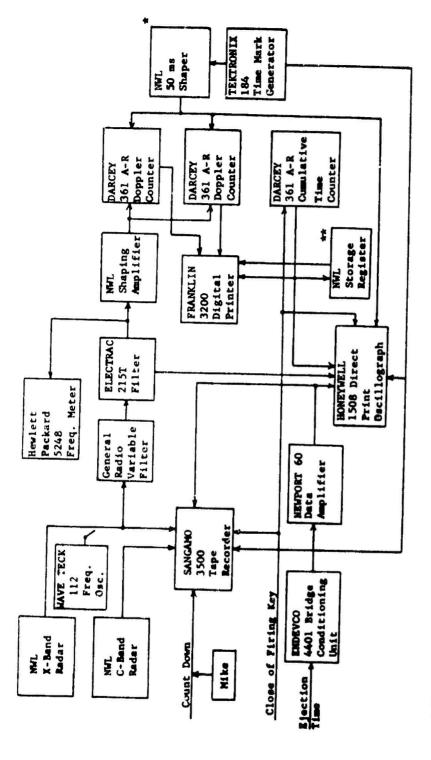
FIGURE 12

View of the Cartridges and Projectiles Being Loaded at the Strikedown Hoist of the 8"/55 MCLGM at NWL

PHD-0218-2-72

FIGURE 13

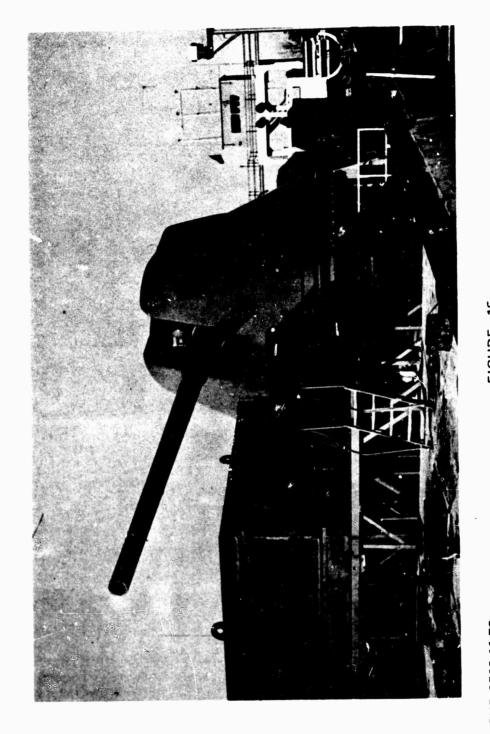
View of the 8".0 MCLGM Firing at NWL



\* Gates counters (on-off alternately) to permit counting of doppler cycles.

FIGURE 14

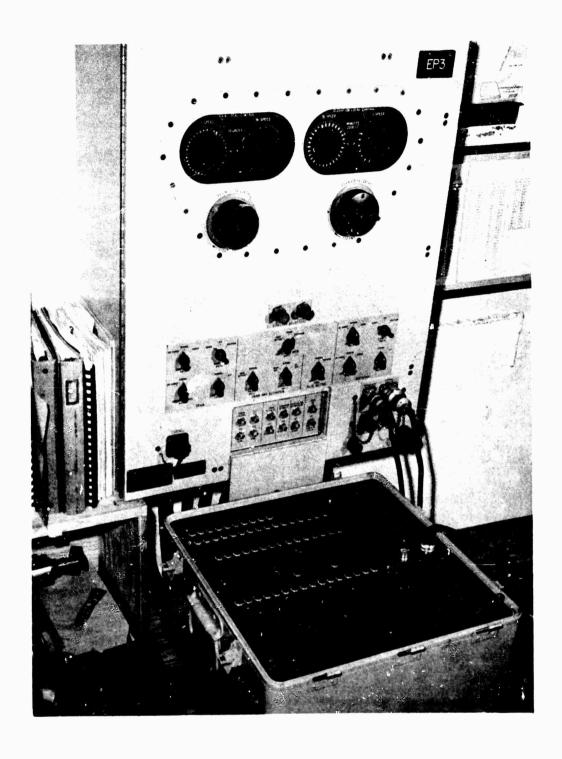
Initial Velocity Data Acquisition System for Doppler Radar



PHJ-2508-11-72 FIGURE 15

View of the 8".0 MCLGM Showing the C-Band Doppler Radar (on Wheels at the Right)

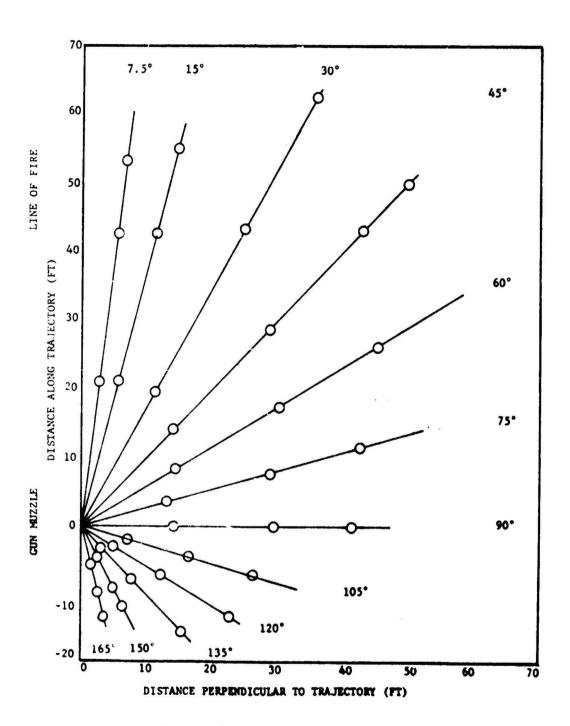
A-15



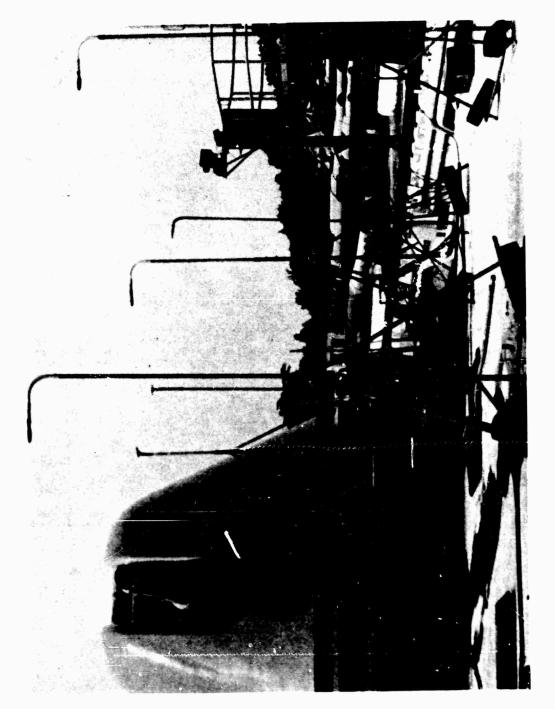
PHD-0172-2-72

FIGURE 16

View of the 8" MCLGM Test Panel (EP-3) and the Order Signal Generator



Distance Perpendicular to Trajectory (ft)



PHD-2509-11-72

FIGURE 18

View of Blast Pressure Transducer Arrangement in Front of the 8"/55 MCLGM

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PHD-2510-11-72 FIGURE 19

View of Blast Pressure Transducer Arrangement in Front of the 8"/55 MCLGM

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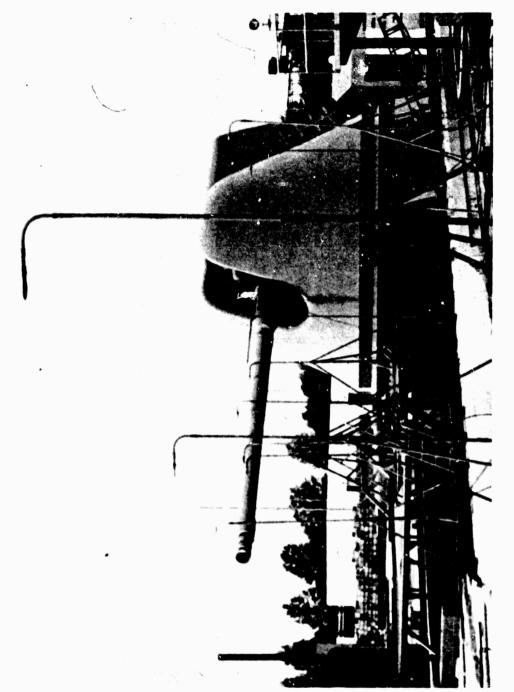
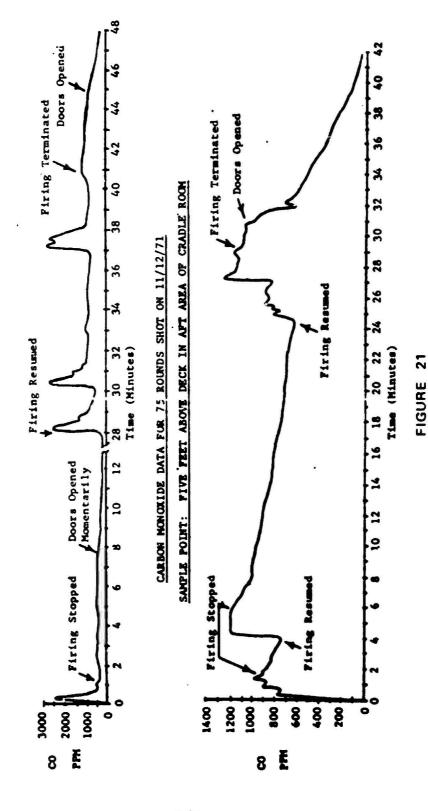


FIGURE 20

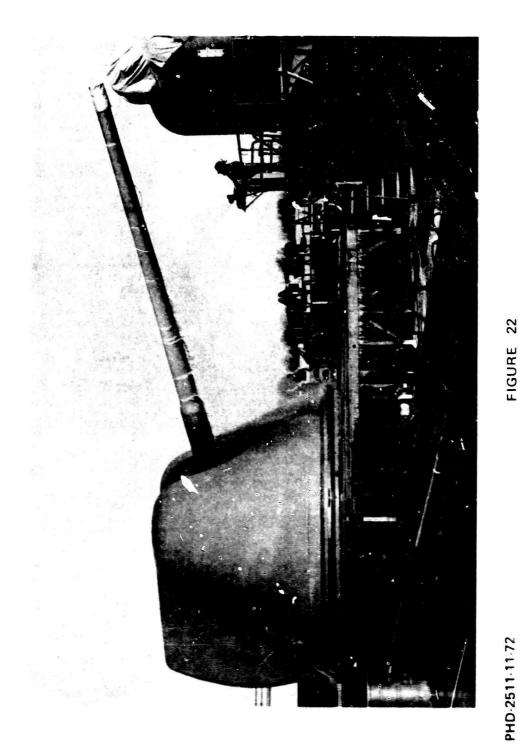
Temperature vs Time at Several Locations on an 8" Projectile

CARBON MONOXIDE DATA FOR LAST 37 ROUNDS SHOT ON 12/13/71
SAMPLE POINT: INSIDE CROWN OF GUN HOUSE, ABOVE BREECH



Carbon Monoxide Concentration vs Time for 8".0 MCLGM

- 4-5



View of the 8" MCLGM During Anti-Icing Test

A-22



View of the Gun Port Shield of the 8".0 MCLGM During the Anti-Icing Test FIGURE 23 PHD-2512-11-72

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# APPENDIX B

Problems Encountered During the TECHEVAL
Table B-1

TABLE B-1

# LIST OF PROBLEMS ENCOUNTERED DURING THE 8" MCLGM TECH EVAL

Date (1971)	Description of Problem	Effect of Problem	Corrective Action
9 September	Extractor cylinder operating piston cap was bowed.	Hydraulic leakage	A redesigned cap was installed.
25 September	Bracket for whiting the tachometer a semily to get em, by case ejector traces was bent in shiraent of the slide to NWL or in installation.	Cannot obtain velocity and position vs. time data for the empty case ejector (Part of MR card C-1 and Tech Eval. Test T-3).	An attempt was made to straighten the bracket, but could not be satisfactorily repaired and this test was omitted.
27-28 Sept	Photocell indicators overly sensitive giving wrong (loaded or unloaded) indications.	Gun stoppage	New type photocells (with the light source on the opposite side of the ammunition train) were installed. This was completed on 30 October.
28 Supt	for the round latch to extend.	Gun stoppage	Small pieces of plug were believed jamming between the case and the chamber. Use of the plug with a radius at the circumference and adding another spring on the cramping the plugs carefully and removing chips around the crimp also reduced the probability of a chip jamming a case and causing a stoppage. The ramer extend indicating switch was adjusted since this was a possible cause.
Throughout the evaluation	Cartridge cases not extracting freely or partially separating.	Gun stoppage	Heat treating cases and dipping in a sulfuric acid solution, degreasing the amountion and chamber, and thoroughly clearing the amountion corrected this problem at service charge pressures. For proof pressure charges, the heat treated cases were not satisfactory and no corrective action proved successful.
20 Oct	One pin of the Dither Pot in the train receiver regulator was in contact with a pin of a transformer.	Train acceleration too slow.	The Dither Pot was rotated $180^\circ$ on the assembly and then the entire Dither assembly was rotated $180^\circ$ .

# TABLE B-1 (Continued)

Date (1971) 22 Oct Prior to 29 Oct I Nov 3 Nov	Primer blowback on reduced charge.  Counter recoil piston picked up material from their bushings.  Cable not locked to empty case tray solemoid housing  Fure for anti-icing pump motor was too small.	Effect of Problem Misfire on next round because of fouled firing pin. Problem was corrected before effects were severe. Intermittent stoppage during simulate cycling. Anti-icing pump wouldn't start.	Corrective Action  Problem is the design/quality control of the MARK 38 MOD 2 primer and not associated with the gum mount. No corrective action was taken which would allow the firing pin assembly to continue functioning after a blowback.  New type bushings were installed with teflon seals. This work was completed on 29 Oct and no problems were encountered after that date.  Locked cable.  Installed a 15-amp fuze.
11 Nov	Screw on switch actuator on the lower accumulator pump sheared.	Lower accumulator failed to start.	Replaced the screw.
12 Hov	Loose bolt at motor contactor in EP-1 panel.	Motors shut down during simulate cycling.	Tightened bolt,
12 Nov	Loose lead in EP-3 panel	Gun stoppage	Lead was pushed back in jack.

Note: Several instances of finding loose or shorted wires during checkout and maintenance are not listed because of their minor consequence.

### APPENDIX C

List of Technical Evalution Tests
Table C-1

### TABLE C-1

## 8"/55 MCLGM TECHNICAL EVALUATION LIST OF TESTS CONDUCTED

### TRAIN POWER DRIVE TESTS

Accuracy Tests
Maximum Velocity and Acceleration Tests
Synchronizing Time Tests
Synchro Power Failure Tests
Limit Stop Tests
Emergency Limit Stop Tests
Frequency Response Tests

### **ELEVATION POWER DRIVE TESTS**

Accuracy Tests
Maximum Velocity and Acceleration Tests
Synchronizing Time Tests
Synchro Power Failure Tests
Limit Stop Tests
Emergency Limit Stop Tests
Frequency Response Tests

### AMMUNITION LOADING SYSTEM TESTS

Drum Indexing
Clip Transferring
Hoist Operation
Cradle Operation
Breechblock Operation
Empty Case Tray Operation
Rammer Operation
Hoist Pawl Positioner Operation
Round Centering Snubber and Latch Operation

### AMMUNITION SELECTABILITY TEST

**ELECTRICAL POWER CONSUMPTION TEST** 

**TEST OF MISFIRE PROCEDURES** 

TEST OF REGUNNING PROCEDURES AND PROVISIONS

**PROOF TESTS** 

**CARTRIDGE ASSEMBLY TESTS** 

**RATE OF FIRE TESTS** 

**GUN JUMP TESTS** 

**ACCURACY AND RANGE TABLE TESTS** 

BARREL LIFE AND VELOCITY LOSS STUDIES

PROJECTILE COOK-OFF TESTS

**REDUCED CHARGE TESTS** 

SAFETY ANALYSIS

**GUN BLAST TESTS** 

**SMOKE AND FUMES TESTS** 

**ANTI-ICING SYSTEM TEST** 

### TABLE C-1 (Continuer

To differs from the list of tests in Robertone 5 in the following ways:

AMMUNITION SELECTABILITY TEST. This test is listed separately rather than as a part of the loading system test.

HEAT RISE. This test was omitted because the schedule precluded doing it after installation of the air conditioning and before the OPEVAL was begun (the est was not regarded to be of primary importance). The heat exchanger was found to be leaking after the OPEVAL, and a new one is being procured. Plans are currently to perform this test after installation of a new heat exchanger and report results by separate correspondence.

RECOIL AND SHIPS LOADING. It is not planned to perform this test since there is doubt as to the importance of the results.

PROJECTILE/CARTRIDGE COOK-OFF. Projectile cook-off is listed but cartridge cook-off is not because it would not present the same degree of hazard if precautions are observed.

EMR INTERFERENCE. This test was conducted during the OPEVAL and is reported in Reference 5.

FIRE OUT. This test is combined with the other tests and . .ot listed separately here.

SMOKE AND FUMES. Smoke and fumes data were accrued during both the TECHEVAL and OPEVAL, and are covered in this report as well as in Reference 6.

ANTI-ICING SYSTEM TEST. Weather conditions did permit some testing of the anti-icing system, hence it is included here.

### APPENDIX D

Train Power Drive Test Data
Table D-1
Figures 1 Through 47

TABLE D-1

# 8"/55 MCLGM TECHNICAL EVALUATION TRAIN POWER DRIVE TEST DATA

### **ACCURACY TESTS**

Figure Number	Input Signal	Error (min)
	Stationary	
1	60°	0.1
1	90°	0.1
1	120°	0.1
	Constant Velocity	
2	5°/sec	0.75
3	-5°/sec	0.5
4	15°/sec	1.3
4	-15°/sec	1.1
5	25°/sec	1.5
6	-25°/sec	1.75
	Simple Harmonic Motion (Amplitude-Period)	
7	$30^{\circ}$ - 9 sec	1.5
8	15° - 4.5 sec	10.0

### **MAXIMUM VELOCITY AND ACCELERATION TESTS**

Figure Number	Maximum Velocity (deg/sec)	Max. Acceleration (deg/sec <sup>2</sup> )
9	31	-62.2
10	-32	61.3

### SYNCHRONIZING TIME TESTS

		Input Signal			
x Figure	,	,	Initial Error	Mount Motion	Synchronizing Time
Number	Order <sup>1</sup>	Order <sup>2</sup>	(deg)	Induced	(sec)
11	Stationary	Stationary	- 5	Right Train	0.63
12	Stationary	Stationary	+5	Left Train	0.68
13	Stationary	Stationary	-20	Right Train	1.31
14	Stationary	Stationary	+20	Left Train	1.32
i.5	Stationary	Stationary	-45	Right Train	2.10
16	Stationary	Stationary	+45	Left Train	2.09
17	Stationary	Stationary	-90	Right Train	3.50
18	Stationary	Stationary	+90	Left Train	3.48
19	Stationary	C.V. 5°/sec	0	Right Train	0.44
20	Stationary	C.V. 5°/sec	-60	Right Train	2.89
21	Stationary	C.V. 10°/sec	0	Right Train	0.37
22	Stationary	C.V. 5°/sec	+60	Left, Then Right	2.31
23	Stationary	C.V. 10°/sec	+60	Left, Then Right	2.12

### TABLE D-1 (Continued)

### SYNCHRO POWER FAILURE TESTS

Figure Number	Mount Motion	Maximum Deceleration (deg/sec <sup>2</sup> )	Distance Traveled After Failure (From Dials)
24	Right Train 31.9 deg/sec	105.5	9° 9′
25	Left Train 32 deg/sec	104.2	10° 23′

### **LIMIT STOP TESTS**

Figure	Mount Motion Into Stop		Maximum Deceleration
Number	Direction	Velocity	(deg/sec <sup>2</sup> )
26	Right Train	15 deg/sec	51.3
27	Left Train	-15 deg/sec	42.7
28	Right Train	Maximum	<b>45</b> .5
29	Left Train	Maximum	48.0
Figure		Mount Motion	Synchronizing
Number	Mount Order	Out of Stop	Time (sec)
30	Constant Veloxity of 15°/sec	Right Train	0.24
31	Constant Veloxity of 15°/sec Constant Velocity of -15°/sec	Left Train	0.24

### **EMERGENCY LIMIT STOP TESTS**

Figure	Mount Motion Into Stop		Maximum Deceleration	Distance Travelo Past Stop Settir
Number	Direction	Velocity	(deg/sec <sup>2</sup> )	(From Dials)
32	Right Train	15 deg/sec	88	1°46′
33	Left Train	-15 deg/sec	67	1°49′
34	Right Train	Maximum	88	6° 59′
35	Left Train	Maximum	77	7°10′

### FREQUENCY RESPONSE TESTS

SHM Input Signal Frequency (Hz)	Amplitude Ratio Output/Input	Approximate Phase Angle (deg)
0.23	1.05	0
0.48	1.12	7.0
0.68	1.15	4.9
0.89	1.30	9.6
1.4	1.45	20.0
2.0	1.55	45.4
2.8	1.45	68.0
3.6	1.28	146
3.9	0.90	166
5.3	0.20	157
7.4	0.33	154
7.9	0.33	163
	0.23 0.48 0.68 0.89 1.4 2.0 2.8 3.6 3.9 5.3 7.4	Frequency (Hz)         Output/Input           0.23         1.05           0.48         1.12           0.68         1.15           0.89         1.30           1.4         1.45           2.0         1.55           2.8         1.45           3.6         1.28           3.9         0.90           5.3         0.20           7.4         0.33

igure ! Accuracy Test, stationary input signal, mount trained at 60°, 90° and 120°

3

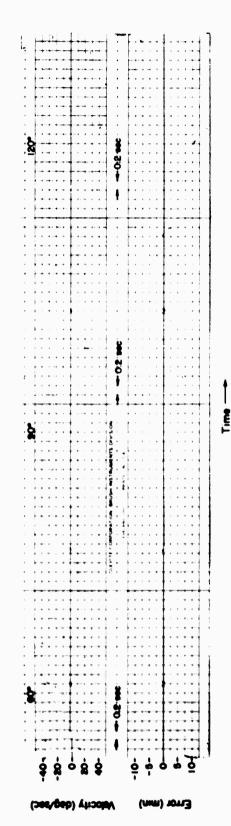


Figure 2. Accuracy Test, contant velocity input signal of 5% sec (right train).

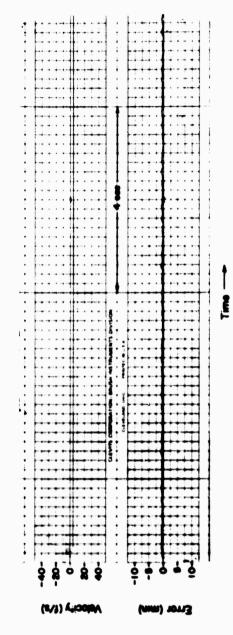


Figure 5. Accuracy Test, content velocity input signal of 25% sec.

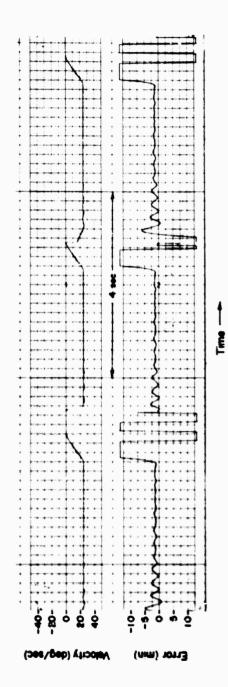


Figure 6. Accuracy Test, constant velocity input signal of -25/sec.

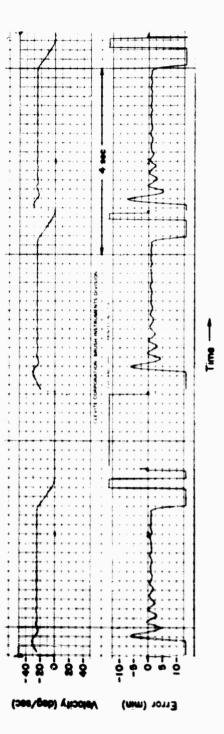


Figure 7. Accuracy Test, SHM input signal of 30° amplitude and 9 sec period.

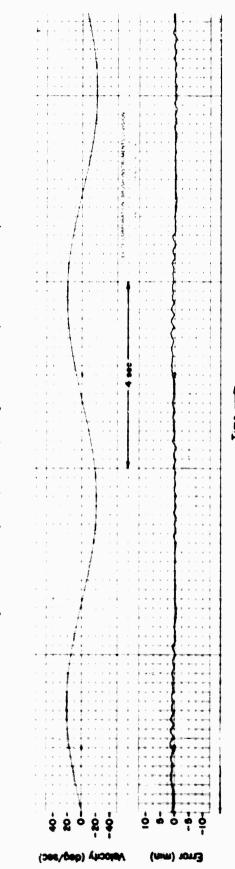
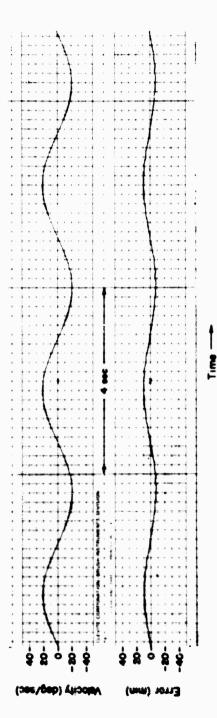
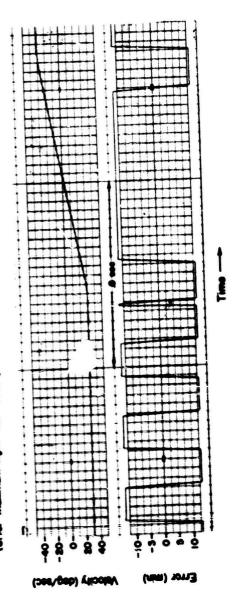


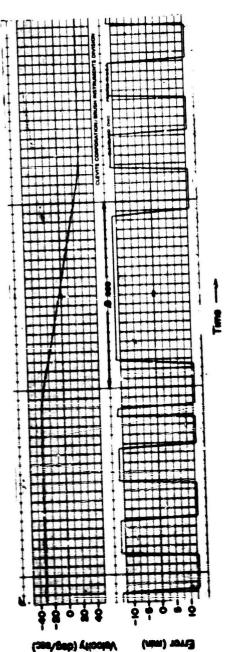
Figure 8. Accuracy Test, SHM input signal of 15° amplitude and 4.5 sec period.



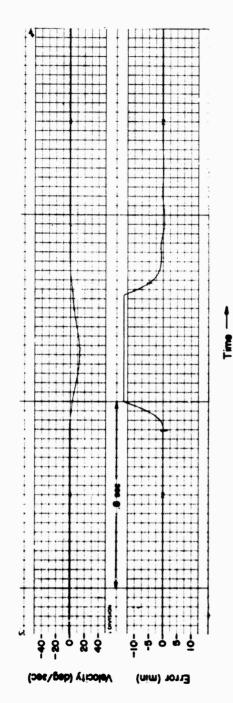
the right and back 2 point Š d Acceleration Test, stationary input signals: Fixed point train velocity is reached) to the uniqued point. Moximum Velocity and (after maximum right h Figure 9.



ğ g o fe st Fixed point to a point input signate: Meximum Vetocity and Acceleration Test, stationary (after maximum left train vetocity is reached) to the Figure 10.



igure II. Synchronizing Time Test, stationary input signal 5° to the right.



re 12. Synchronizing Time Test, stationary input signal 5° to the left.

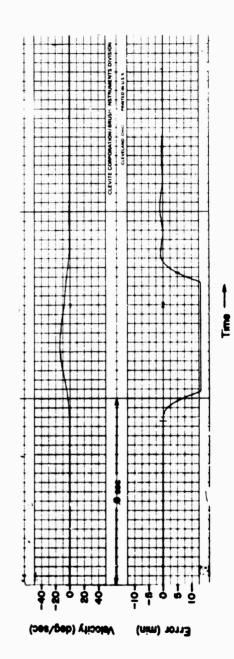


Figure 13. Synchronizing Time Test, stationary input signal 20° to the right

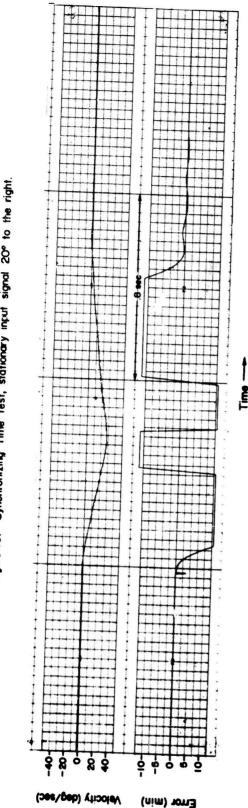


Figure 14. Synchranizing Time Test, stationary input signal 20° to the left.

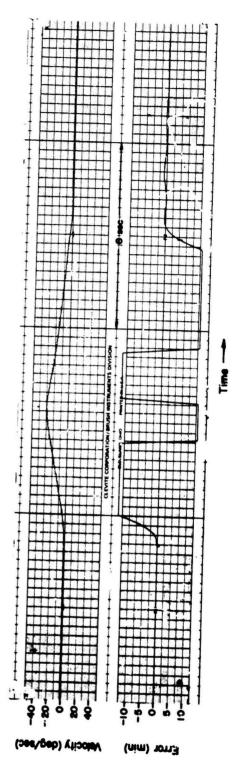


Figure 15. Synchronizing Time Test, stationary input signal 45° to the right.

(

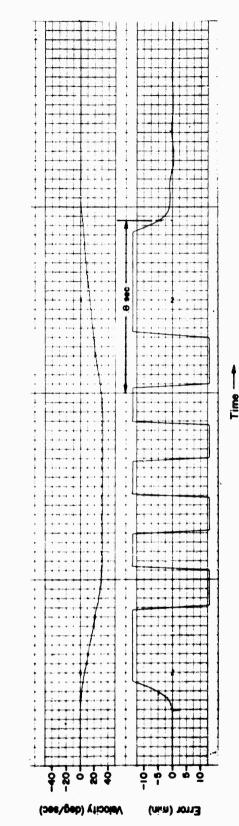


Figure 16. Synchronizing Time Test, stationary input signal 45° to the left.

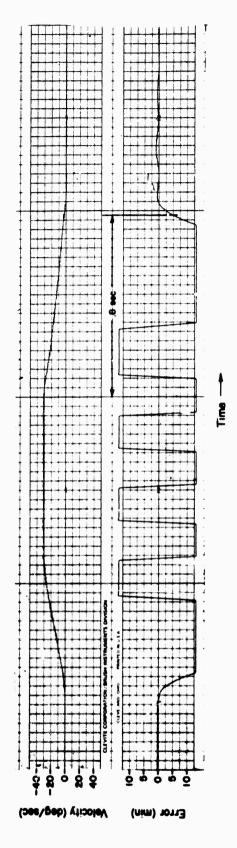


Figure 17 Synchronizing Time Test, stationary input signal 90° to the right.

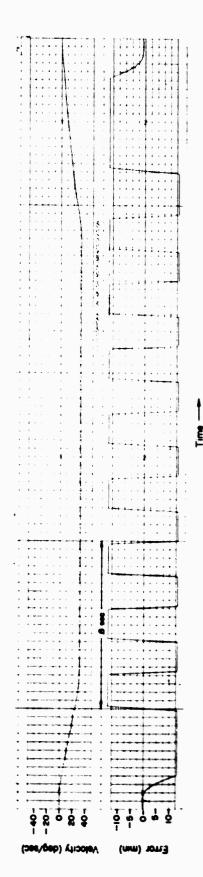
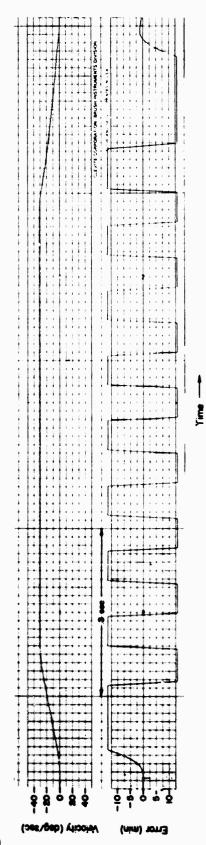
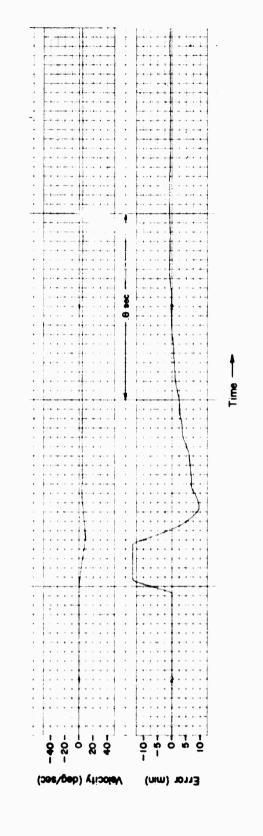


Figure 18. Synchronizing Time Test, stationary input signal 90° to the left.



ဝိ ð Figure 19. Synchronizing Time Test, constant velocity input signal of 5% sec, initial error



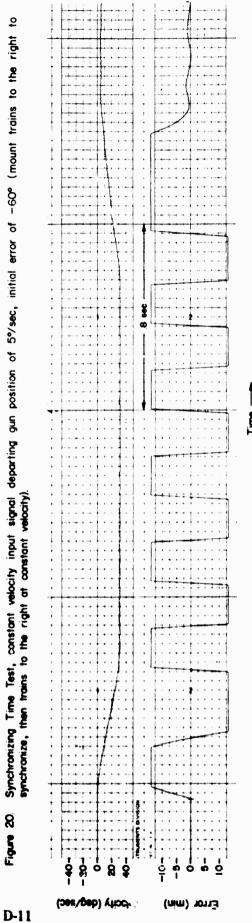
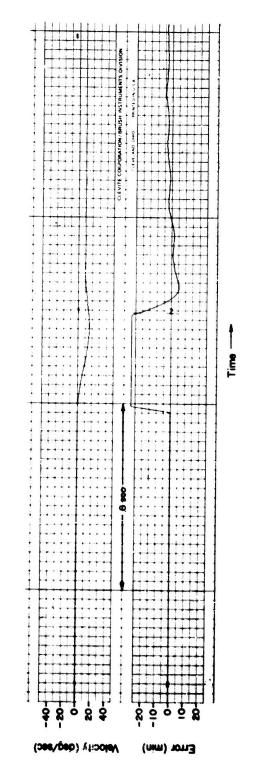
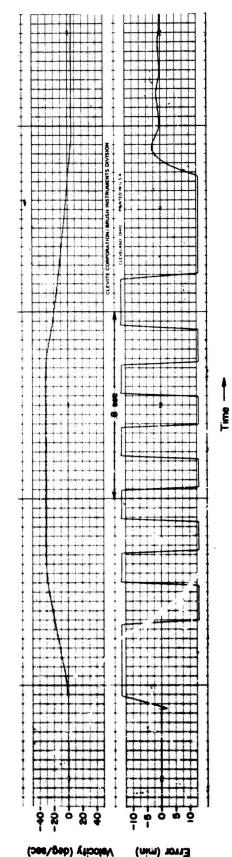
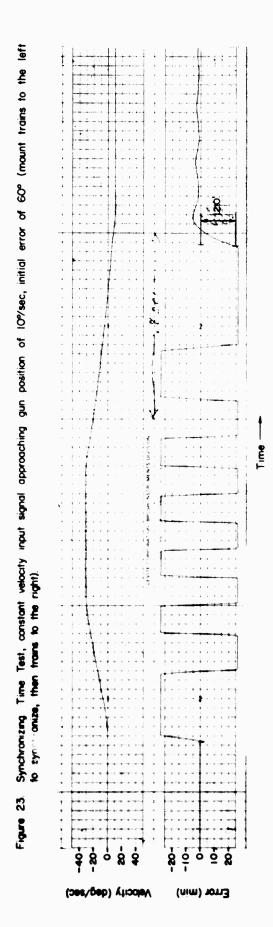


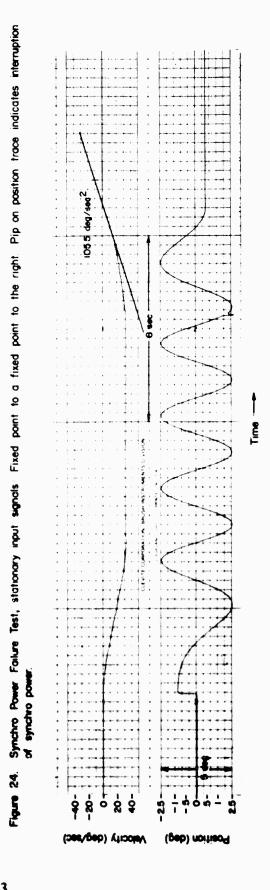
Figure 21. Synchronizing Time Test, constant velocity input signal of 10%sec, initial error of 09



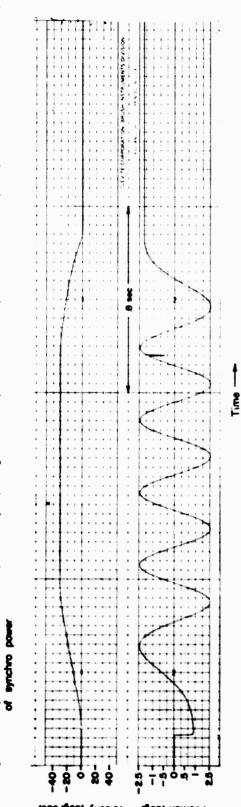
Synchronizing Time Test, constant velocity input signal approaching gun position of 5% sec, initial error of 60° (mount trains to the left to synchronize, then trains to the right).



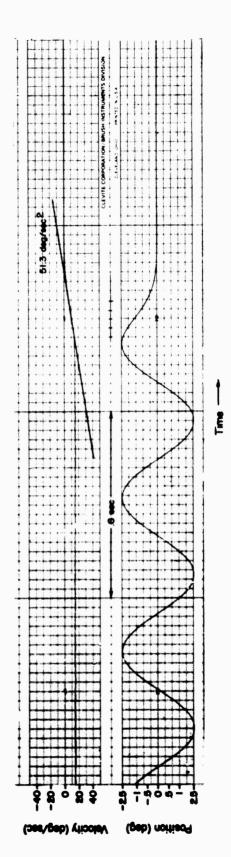




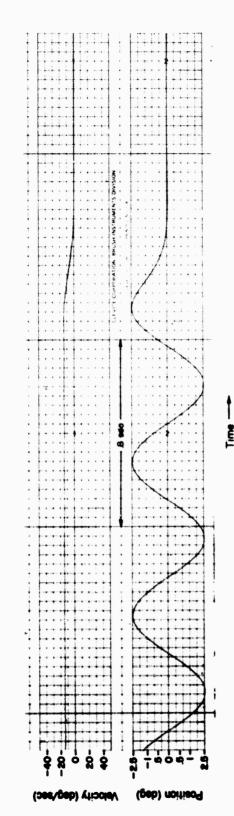
Synchro Power Failure Test, stationary input signals. Fixed point to a fixed point to the left. Pip on position trace indicates interruption



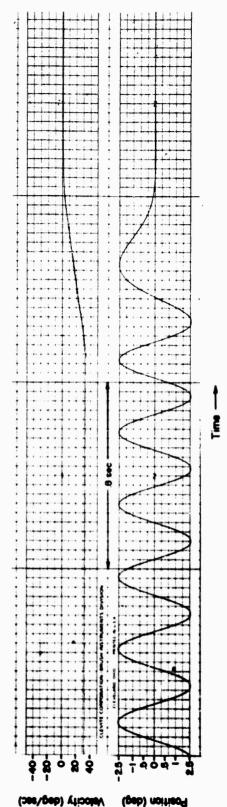
Limit Stop Test, constant velocity input signal of 15%sec (trained into right limit stop). Figure 26.



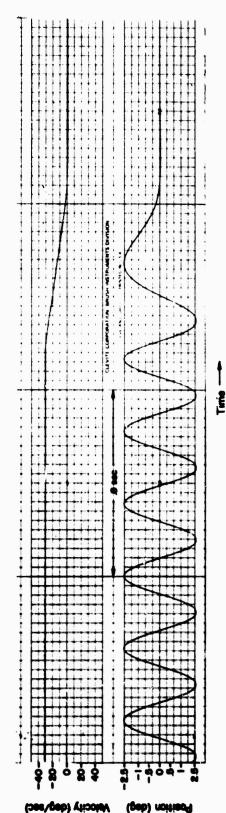
-15%sec (trained into left limit stop). ŏ Limit Stop Test, constant velocity input signal Figure 27.



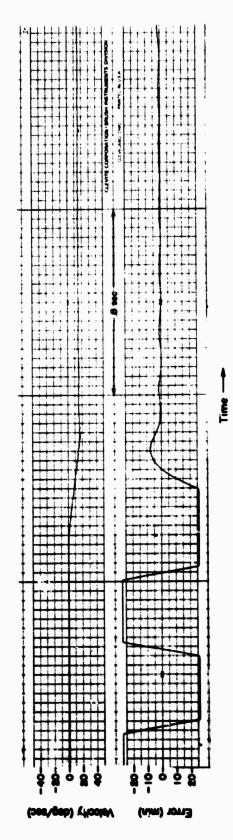
point to the right beyond the right limit stop (trained into right limit 0 2 poert Limit Stop Test, stationary input signus: Fixed stop at maximum velocity). Figure 28



the left limit stop (trained into left to the left beyond point 0 2 input signals: Fixed point Stop Test, stationary of maximum velocity). t Figure 23



Ī 8 Ē (Trained 15/sec 7 input signal Test, constant velocity Sop 3 Figure 30.



Limit Stop Test, constant velocity input signal of -15%sec (trained out of right limit stop). Figure 31.

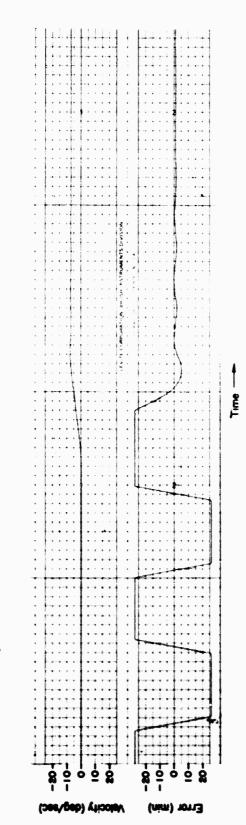
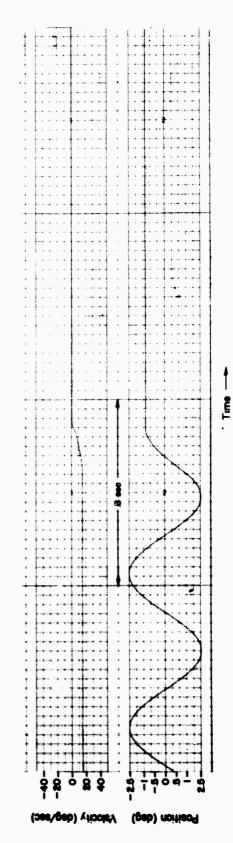
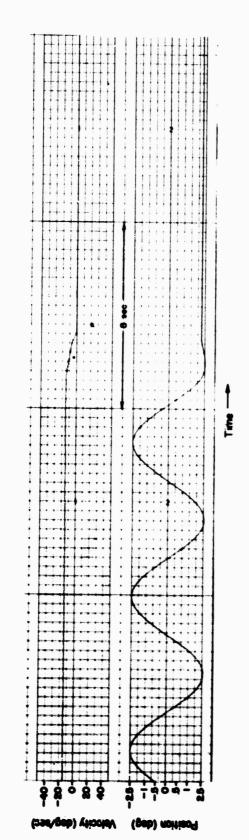


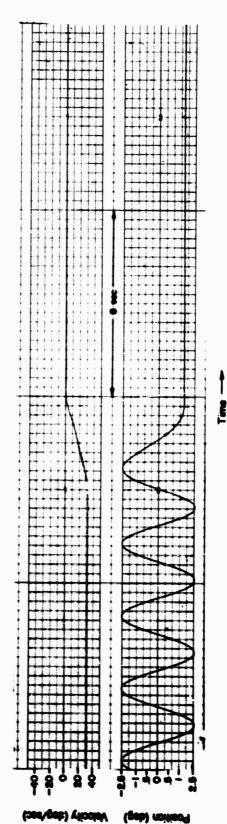
Figure 32. Emergency Limit Stop Test, constant velocity input 15% sec (trained into right emergency limit stop).



Emergency Limit Stop Test, constant velocity input signal of -15%sec (trained into left emergency limit stop). Figure 33



Energency Limit Stop Test, stationary input signals: Fixed point to a point to the right beyond the right emergency limit stop (trained into right emergency limit stop at maximum velocity). Figure 12.



Emergency Limit Stop Test, stationary input signals. Fixed point to a point to the left beyond the left emergency limit stop (trained into left emergency timit stop at maximum velocity). Figure 35

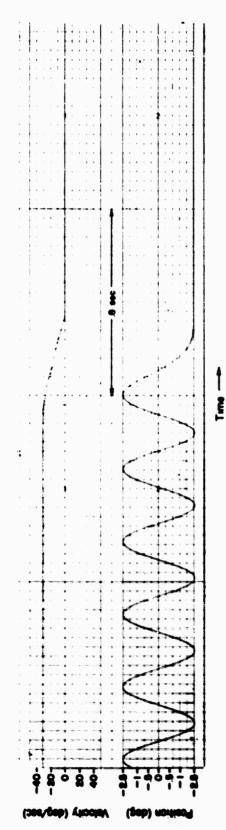


Figure 36. Frequency Response Test, SHM input of 3' amplitude, 0.23 Hz frequency.

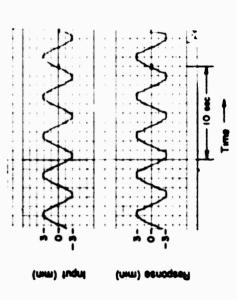


Figure 37. Frequency Response Test, SHM input of 3' amplitude, 0.48Hz frequency.

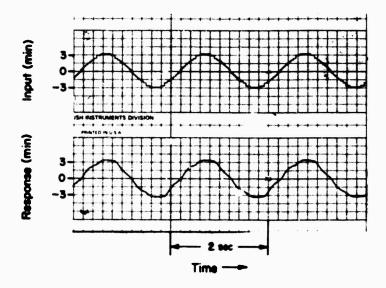


Figure 38. Frequency Response Test, SHM input of 3' amplitude, 0.68 Hz frequency.

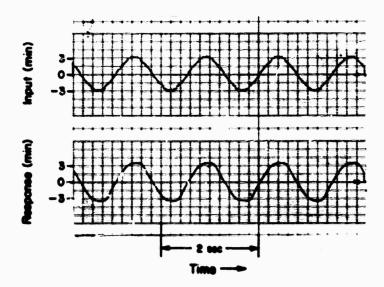


Figure 39. Frequency Response Test, SHM input of 3' amplitude, 0.89Hz frequency.

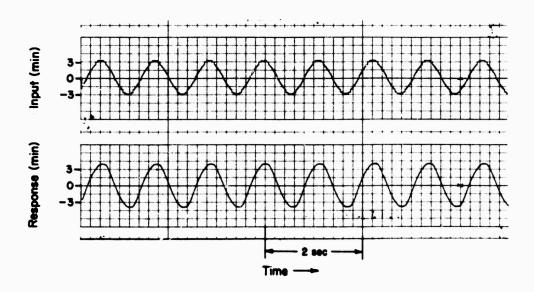


Figure 40. Frequency Response Test, SHM input of 3' amplitude, 1.4 Hz frequency.

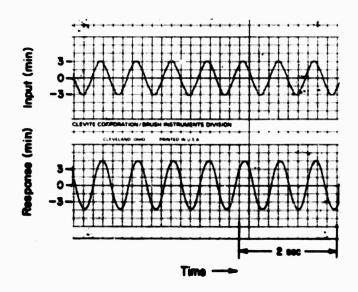


Figure 41. Frequency Response Test, SHM input of 3' amplitude, 2.0Hz frequency.

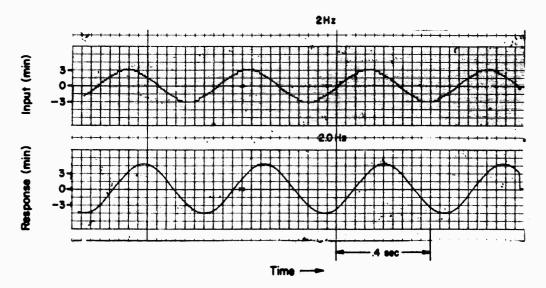


Figure 42. Frequency Response Test, SHM input of 3' amplitude, 2.8 Hz frequency.

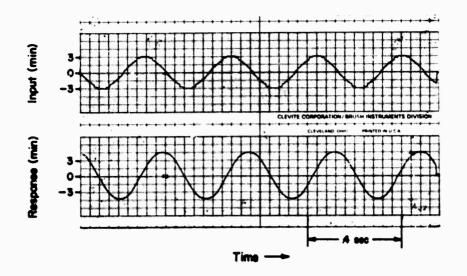


Figure 43. Frequency Response Test, SHM input of 3' amplitude, 3.6Hz frequency.

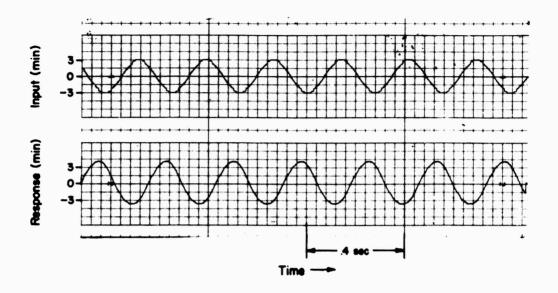


Figure 44. Frequency Response Test, SHM input of 3' amplitude, 3.9Hz frequency.

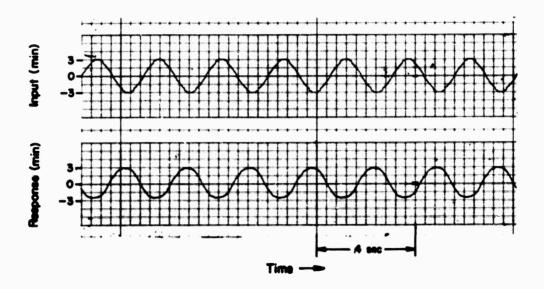


Figure 45. Frequency Response Test, SHM input of 3' amplitude, 5.3Hz frequency.

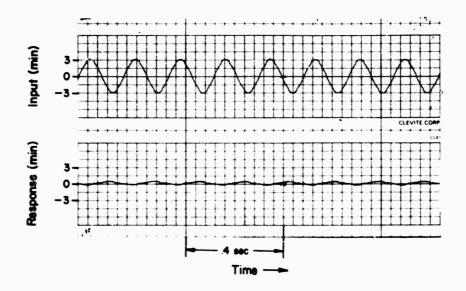
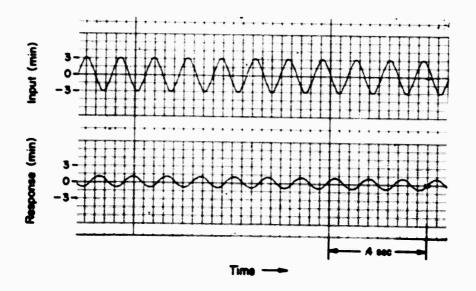
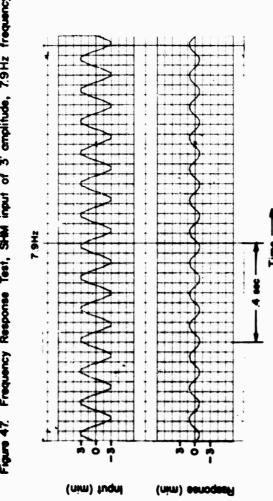


Figure 46. Frequency Response Test, SHM input of 3' amplitude, 7.4 Hz frequency.







APPENDIX E

Elevation Power Drive Test Data Table E-1 Figures 1 Through 50

TABLE E-1

8"/55 MCLGM TECHNICAL EVALUATION
ELEVATION POWER DRIVE TEST DATA

### **ACCURACY TESTS**

Figure Number	Input Signal	Error (min)
	Stationary	
1	2000' (0 deg)	0.1
1	3000' (16.67 deg)	0.5
2	4000' (33.33 deg)	0.3
2	5000' (50.00 deg)	0.1
	Constant Velocity	
3	5°/sec	0.6
3	5°/sec	1.0
	10°/sec	1.0
4 4 5 5	-10°/sec	1.2
5	15°/sec	1.0
5	-15°/sec	2,0
	Simple Harmonic Motion	
	(Amplitude-Period)	
6	5°-4.5 sec	1.75
7	10°-4.5 sec	1.75
8	5°-9 sec	0.75
9	10°-9 sec	1.25
10	15°-9 sec	1.75
11	20°-9 sec	1.50

### MAXIMUM VELOCITY AND ACCELERATION TESTS

	Maximum	Maximum
Figure	Velocity	Acceleration
Number	(deg/sec)	(deg/sec <sup>2</sup> )
12	20.3	-41.4
13	-21.1	41.4

TABLE E-1 (Continued)

### SYNCHRONIZING TIME TESTS

Figure		Input Sign	als	Gun Motion	Synchronizing
Number	Order 1	Order <sup>2</sup>	Initial Error (deg)	Induced	Time (sec)
14	Stationary	Stationary	-5	Elevate	0.78
15	Stationary	Stationary	+5	Depress	0.78
16	Stationary	Stationary	-20	Flevate	1.64
17	Stationary	Stationary	+20	Depress	1.63
18	Stationary	Stationary	-45	Elevate	2,78
19	Stationary	Stationary	+45	Depress	2.76
20	Stationary	C.V. 5°/sec	0	Elevate	0.68
21	Stationary	C.V. 5°/sec	+30	Depress then	
				Elevate	1.88
22	Stationary	C.V. 5°/sec	-60	Elevate	2.45
23	Stationary	C.V. 5°/sec	+60	Depress then	
				Flevate	2.78
24	Stationary	C.V. 10°/sec	+30	Depress then	
				Elevate	1.79
25	Stationary	C.V. $10^{\circ}/\text{sec}$	0	Elevate	0.56
26	Stationary	C.V. 10°/sec	+60	Depress then	
				Flevate	2.72
27	Stationary	C.V. 10°/sec	+90	Depress then	
				Elevate	3.84

### SYNCHRO POWER FAILURE TESTS

Figure Number	Mount Motion	Maximum Deceleration (deg/sec <sup>2</sup> )	Distance Traveled After Failure (From Dials)
28	Elevate	108	6°6′
29	Depress	100	6°18′

### LIMIT STOP TESTS

Figure	Gun Motion	Maximum Deceleration	
Number	Direction	Velocity	$(\text{deg/se}c^2)$
30	Elevate	+10°/sec	25
31	Depress	-10°/sec	25
32	Elevate	Maximum	27.8
33	Depress	Maximum	31.6

TABLE E-1 (Continued)

Figure Number	Mount Order	Gun Motion Out of Stop	Synchronizing Time (sec)
34	Constant Velocity of 10°/sec	Elevate	0.23
35	Constant Velocity of -10°/sec	Depress	0.28

### **EMERGENCY LIMIT STOP TESTS**

Figure		n Into Stop	Maximum Deceleration	Distance Before Setting Gun Stopped
Number	Direction	Velocity	(deg/sec <sup>2</sup> )	(From Dials)
36	Elevate	10°/sec	79	3°26′
37	Depress	-10°/sec	75	3°16′
38	Elevate	Maximum	92	4°0′
39	Depress	Maximum	80	3°59′

Gun stopped position when driven into each actual emergency limit stop at a constant velocity of  $1^{\circ}/\text{sec}$ .

Depression	Elevation
-5° 25'	65° 21′

### FREQUENCY RESPONSE TESTS

Figure Number	SHM Input Signal Frequency (Hz)	Amplitude Rătio Output/Input	Approximate Phase Angle (deg)
40	0.24	1.075	1.76
41	0.49	1.15	1.80
42	0.68	1.25	1.5
43	1.8	1.48	46.5
44	2.0	1.45	53.0
45	2.5	1.30	54.7
46	2.8	1.28	80.5
47	4.0	0.95	145
48	5.6	0.30	124
49	7.8	0.10	137

Figure 1. Accuracy Test, stationary input signal, gun elevated at 2000 min (0°0' elevation from horizontal and 3000 min.

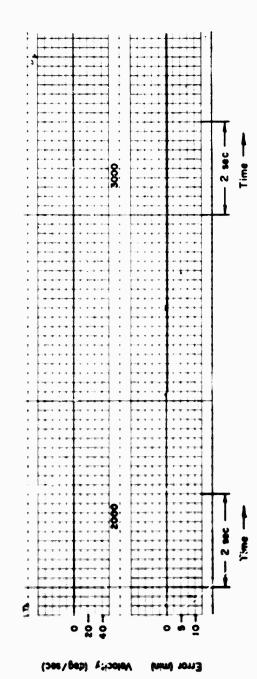
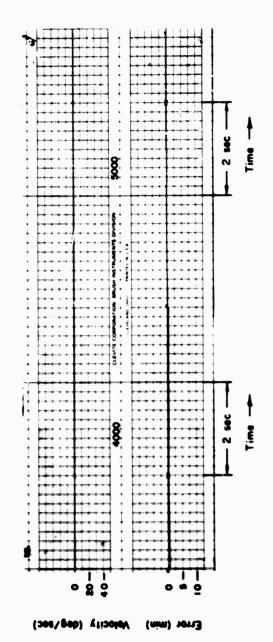


Figure 2. Accuracy Test, stationary input signal, gun elevated at 4000 min and 5000 min



Time T

Figure 3. Accuracy Test, constant velocity input signals of 5°/sec (elevating) and -5°/sec (depressing).

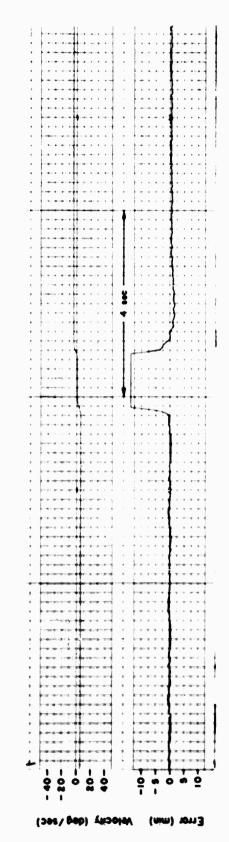


Figure 4. Accuracy Test, constant velocity input signals of 10°/sec and -10°/sec

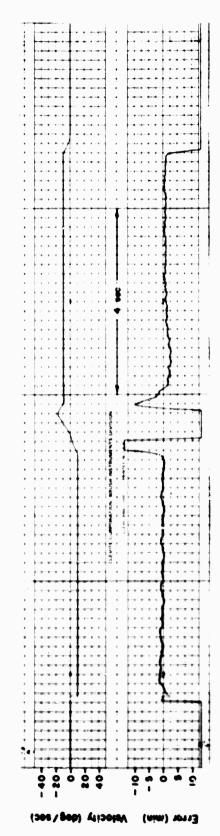


Figure 5. Accuracy Test, constant velocity input signals of 15\*/sec and -15\*/sec.

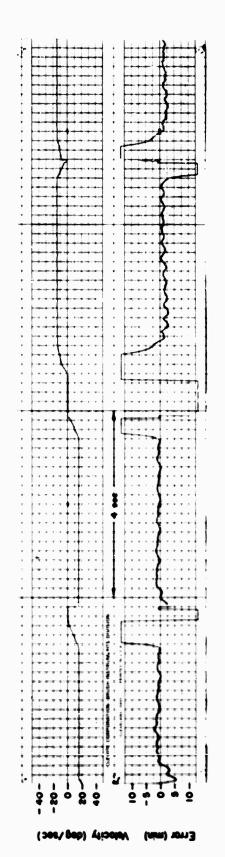
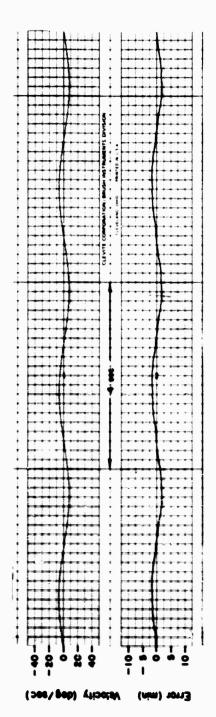
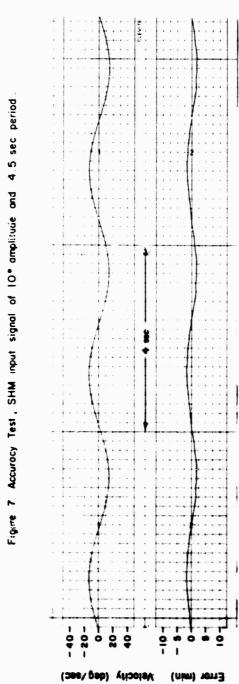


Figure 6. Accuracy Test, SHM input signal of 5° amplitude and 4.5 sec period.



Time





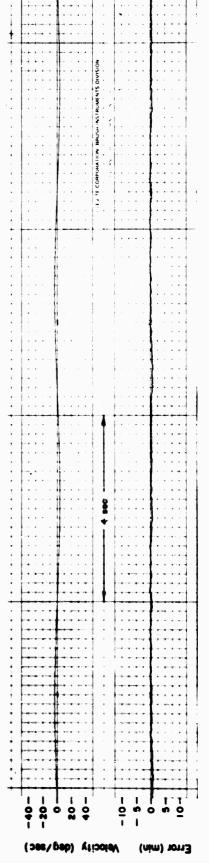
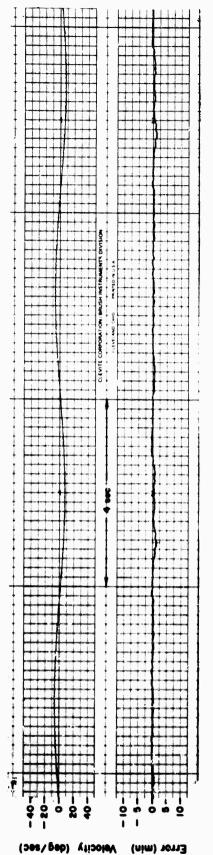


Figure 9. Accuracy Test, SHM Input signal of 10° amplitude and 9 sec period.



Time -

Figure 10. Accuracy Test, SHM Input signal of 15° amplitude and 9 sec periad.

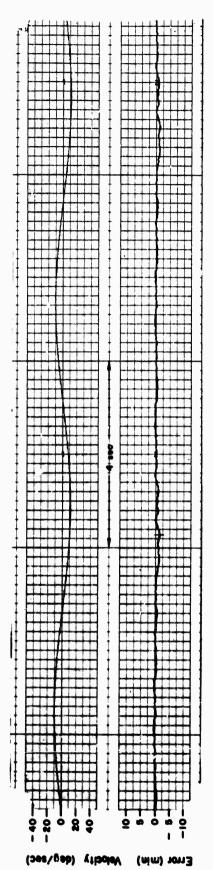
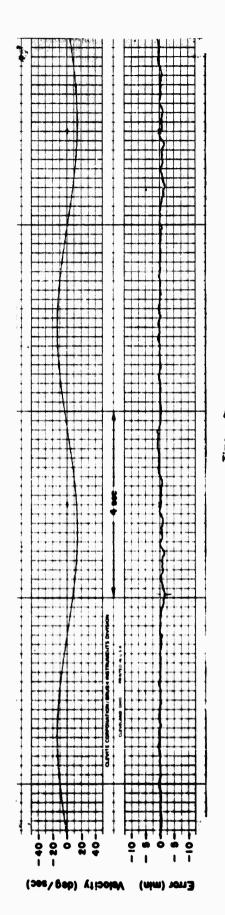
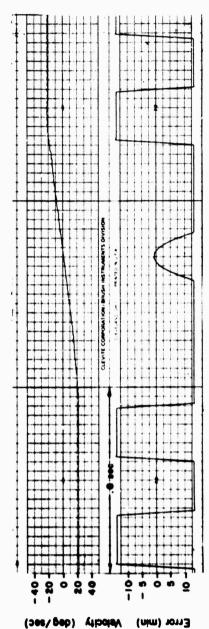


Figure 11. Accuracy Test, SHM input signal of 20° amplitude and 9 sec period.



point Figure 12. Maximum Velocity and Acceleration Test, stationary input signals: Fixed point to a higher fixed and back (after maximum depressing velocity is reached) to the original point.



Time

bock g lower fixed point 9 Maximum Velocity and Acceleration Test, stationary input signals: Fixed point (after maximum depressing velocity is reached) to the original point.

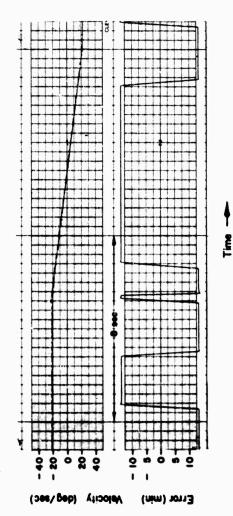
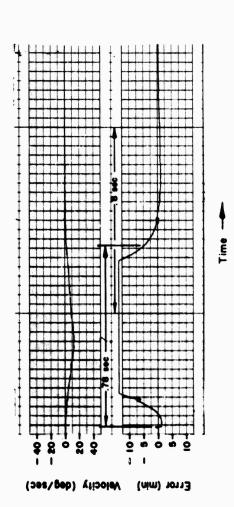


Figure 14. Synchronizing Time Test, stationary input signal 5° high (gun elevates).



Time 1



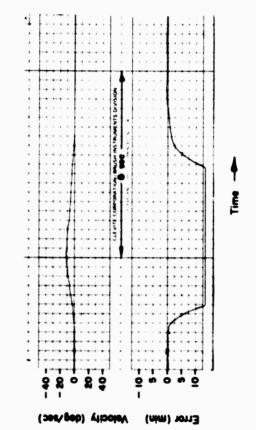


Figure 16. Synchronizing Time Test, stationary input signal 20 high.

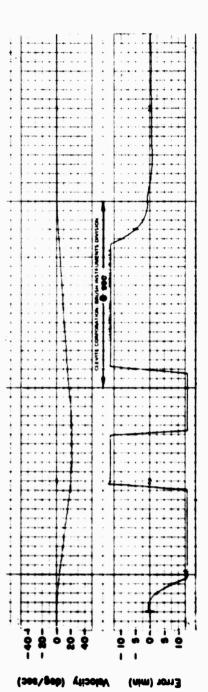


Figure 17. Synchronizing Time Test, stationary input signal 20° low

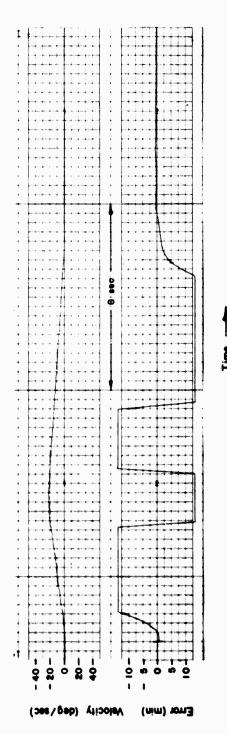


Figure 18. Synchronizing Time Test, stationary input signal 45° high.

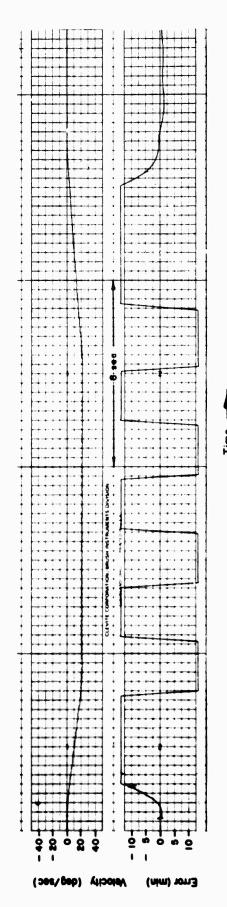
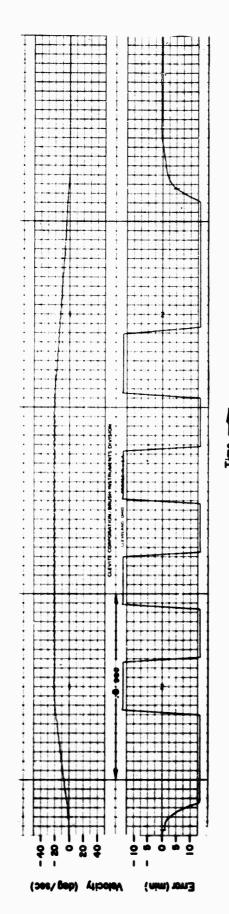
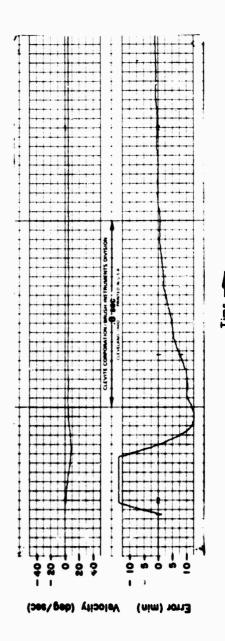


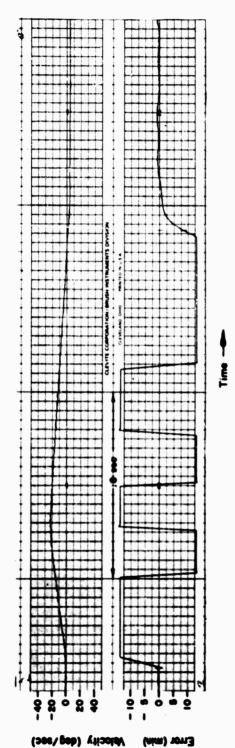
Figura 19. Synchronizing Time Test, stationary input signal 45° low



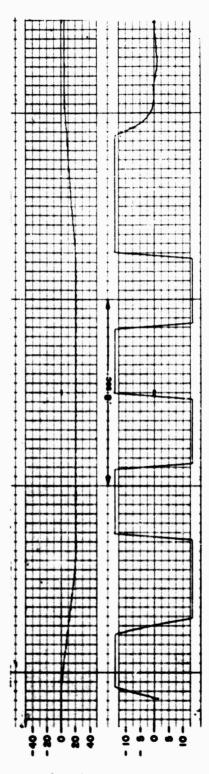
• ð Figure 20. Synchronizing time test, Constant velocity input signal of 5°/sec, initial error



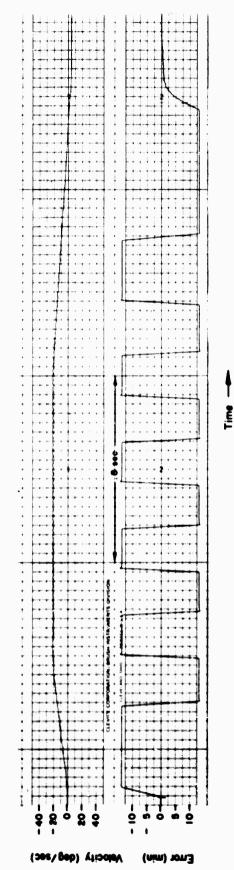
ð error 5./sec, initial Figure 21. Synchronizing Time Test, constant velocity input signal approaching gun position of  $30^{\circ}$  (gun depresses to synchronize, then elevates at  $5^{\circ}$ /sec).



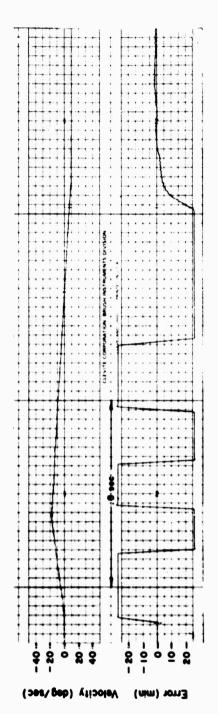
error 5 -/ sec, initial ŏ Figure 22. Synchronizing Time Test, constant velocity input signal departing gun position of -60° (gun elevates to synchronize, then elevates at 5%cc).



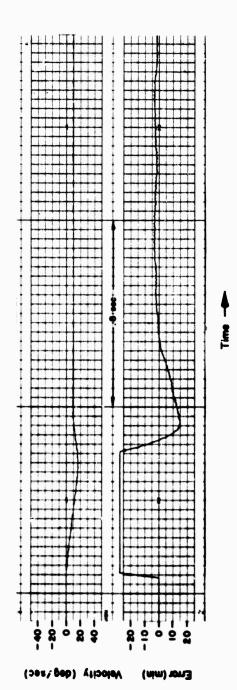
Synchronizing Time Test, Constant velocity input signal approaching gun position of 5"/sec, initial error of 60" (gun depresses to synchronize, then elevates at 5"/sec).



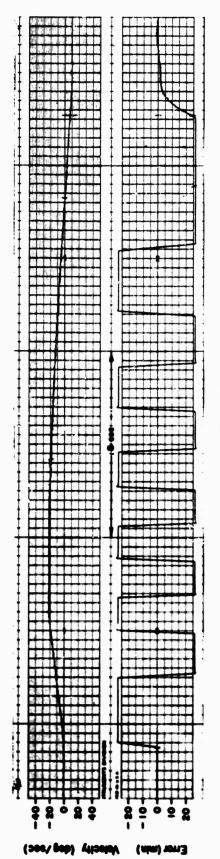
ŏ Figure 24. Synchronizing Time Test, Constant velocity input signal approaching gun position of 10°/sec, initial error 30° (gun depresses to synchronize, then elevates at 10°/sec).



• ð error 10 \*/sec, initial ŏ Signal <u>i</u> velocity Test, Constant Ē Synchronizing 25



8 ð error 10°/sec, initial ŏ position ž approaching nt velocity input signal elevates at 10 °/sec). Time Test, constant synchronize, then et Synchronizing depresses to 2



₽ Figure 27. Synchronizing Time Test, constant velocity input signal approaching gun position of 10"/sec, initial arror of 90", (gun depresses synchronize, then elevates at 10°/sec).

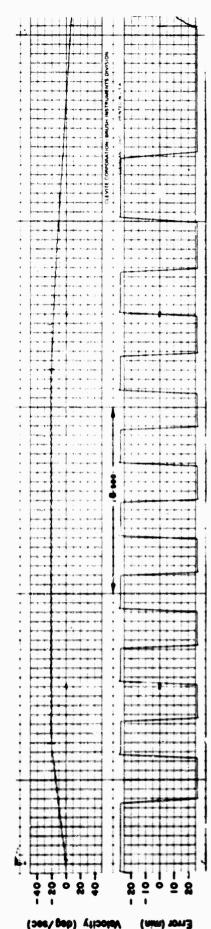
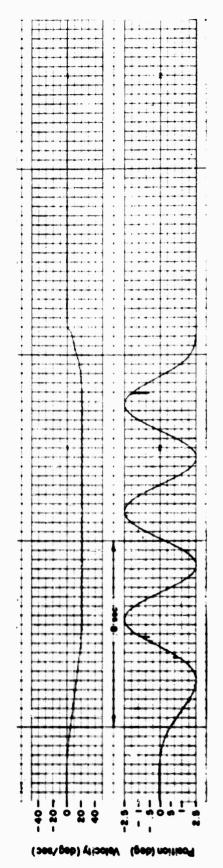
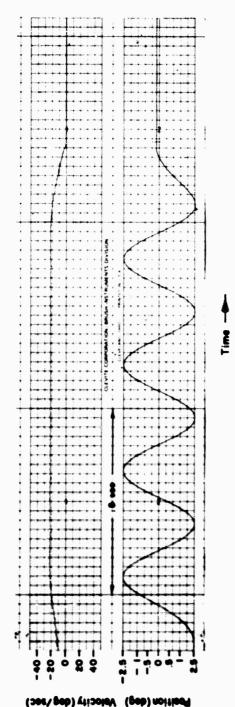


Figure 26. Synchro Power Follure Test, stationary input signals: Fixed point to a higher fixed point. Pip on position trace indicates interruption of synchro power.



ā lower fixed point. 0 2 point Synchro Power Failure Test, stationary input signals: Fixed position trace indicates interruption of synchro power. Figure 29.

5



of 10\*/sec (elevated into upper limit stop). Figure 30. Limit Stop Test, constant velocity input signal

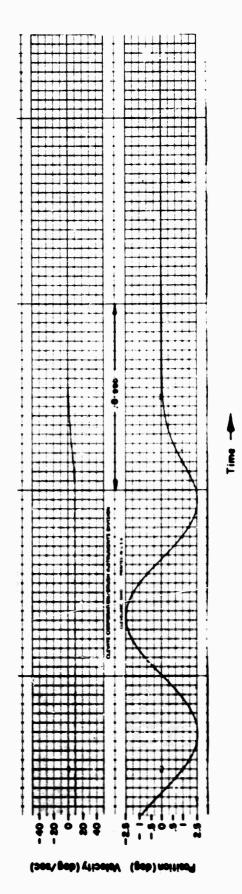
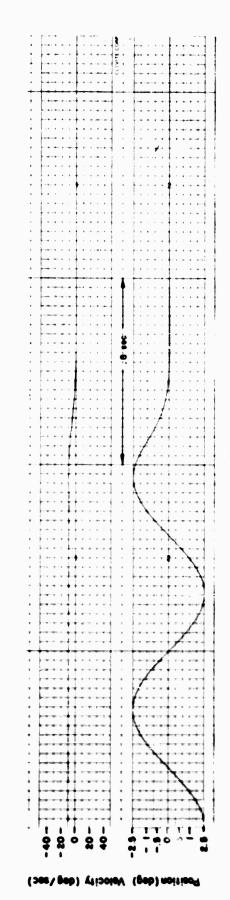
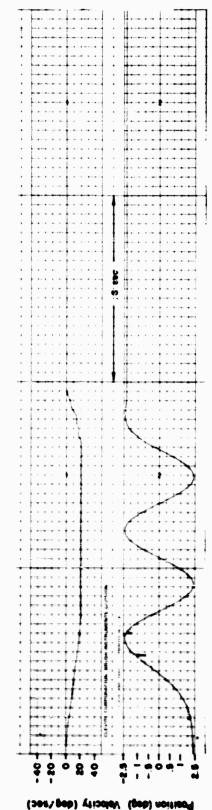


Figure 31. Limit Stop Test, constant velocity input signal of -10°/sec (depressed into lower limit stap).



Limit Stop Test, stationary input signals: fixed point to a higher fixed point beyond the upper limit stop (slevated into upper limit stop at maximum velocity). Figure 32.



33. Limit Stop Test, stationary trout signals: Fixed point to a lower fixed point beyond the lower limit stop (depressed that lower limit stop at maximum velocity).

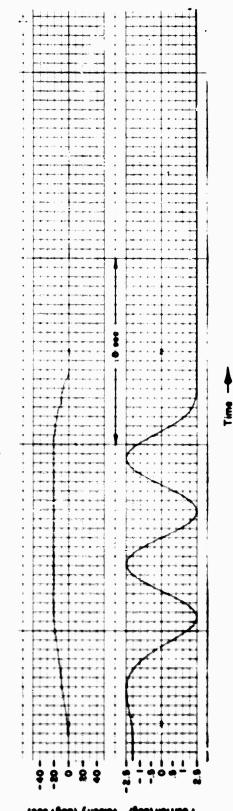
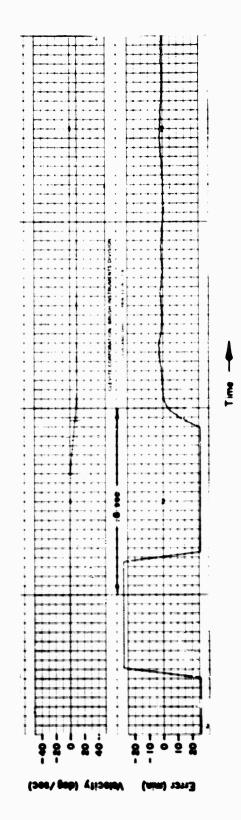
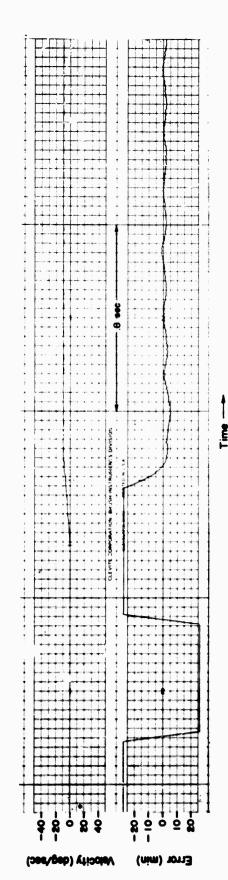


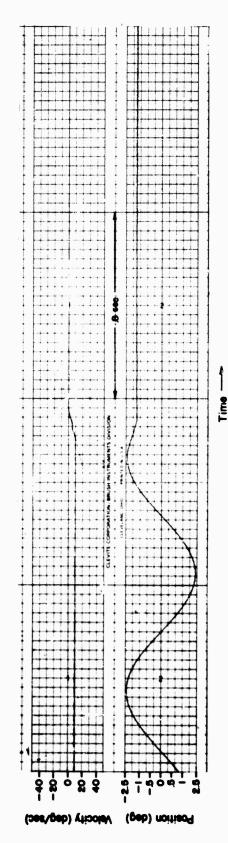
Figure 34. Limit Stop Test, constant velocity input signal of 10°/sec (gun elevated out of lower limit stop).



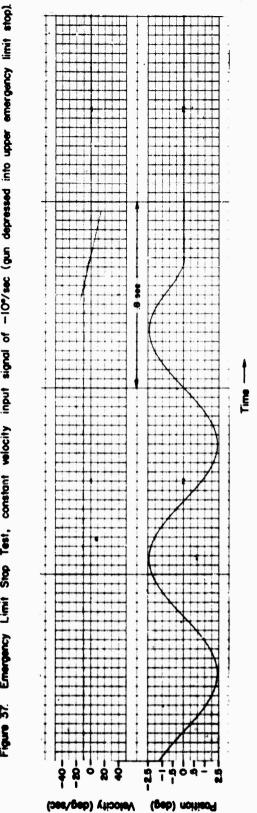
(imit stop) Limit Stop Test, constant velocity input signal of -10% sec (gun depressed out of upper Figure 35.



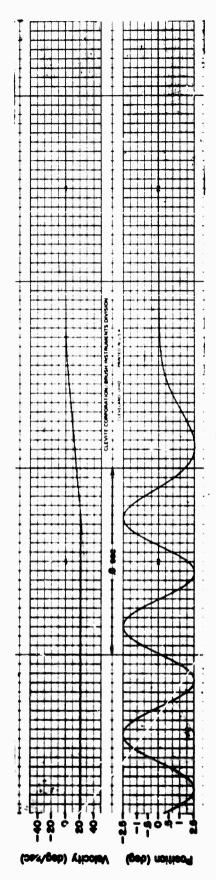
Emergency Limit Stop Test, Constant velocity input signal of 10% sec (gun elevated into upper emergency limit stap). 8 Figure



Ē emergency (gun depressed into -10/sec ð input signal constant velocity Test. Stop Limit Emergency 37



Ē emergency Ę above point rixed point to a higher maximum velocity). ŧ signols: o do emergency limit Por stationary Emergency Limit Stop ' (gun elevated into the Ħ Figure



Emergency Limit Stop Test, stationary input signals: Fixed point to a lower point below the lower emergency limit stop (gun depressed into the lower emergency limit stop at maximum velocity).

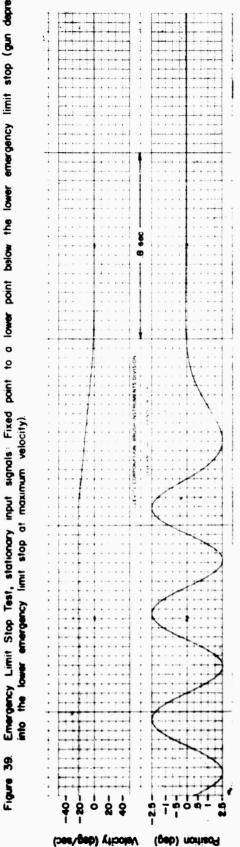


Figure 40. Frequency Response Test, SHM input of 3' amplitude, 0.2Hz frequency.

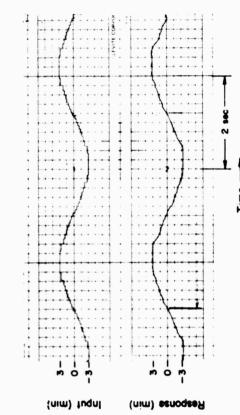
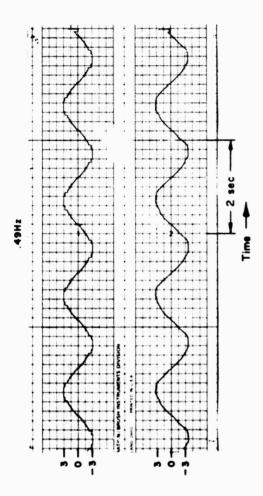


Figure 41. Frequency Response Test, SHM input of 3' amplitude, 0.49Hz frequency



Responsé (min.)

(nim) tuqnl

Figure 42. Frequency Response Test, SHM input of 3'amplitude, 0.68Hz frequency.

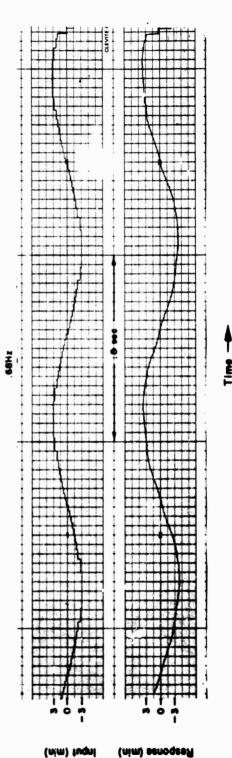


Figure 43. Frequency Response Test, SHM input of  $3^{\circ}$  amplitude, 1.8Hz frequency.

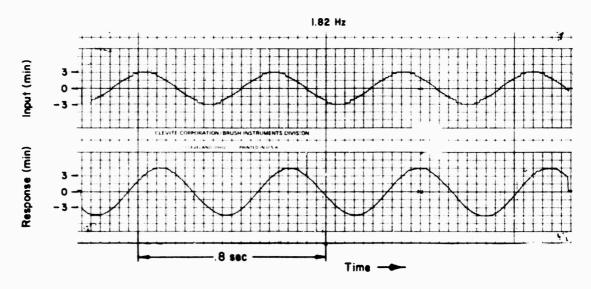


Figure 44. Frequency Response Test, SHM input of 3 amplitude, 2.0Hz frequency.

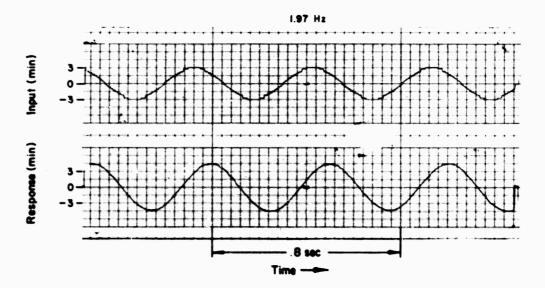


Figure 45. Frequency Response Test, SHM input of 3' amplitude, 2.5Hz frequency.

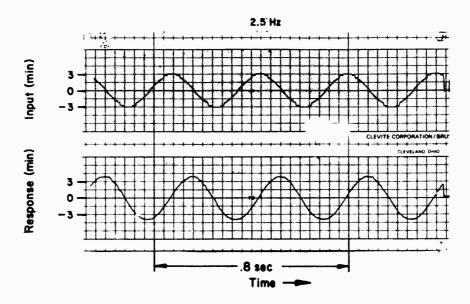


Figure 46. Frequency Response Test, SHM input of 3 amplitude, 2.8Hz frequency.

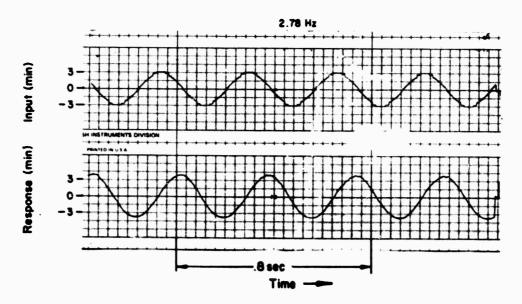


Figure 47. Frequency Response Test , SHM input of 3 amplitude , 4.0Hz frequency .

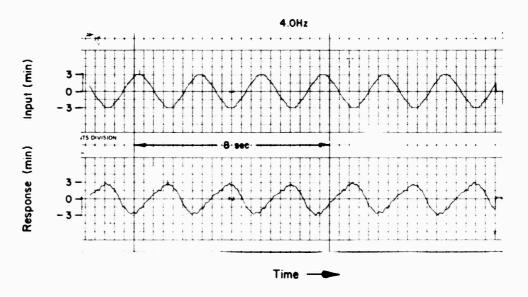


Figure 48. Frequency Response Test, SHM input of 3' amplitude, 5.6Hz frequency.

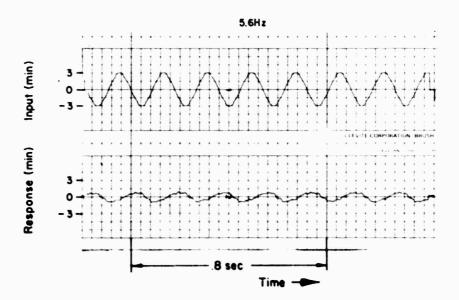
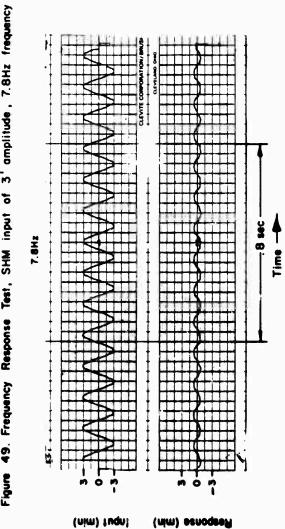


Figure 49. Frequency Response Test, SHM input of 3' amplitude, 7.8Hz frequency.



## APPENDIX F

Loading System Test Data Table F-1 Figures 1 through 24

8"/55 MCLGM TECHNICAL EVALUATION

TABLE F-1

## 8"/55 MCLGM TECHNICAL EVALUATION LOADING SYSTEM TEST DATA

Figure Number	Component Operation	Cycle Time (sec)	Maximum Velocity
	Drum Indexing		•
1	Empty, CW	1.10	32.4°/sec
2 3	Empty, CCW	1.48	34.8°/sec
	Loaded (75 rds) CW	1.12	34.1°/sec
4	Loaded (75 rds) CWW	1.56	34.8°/sec
	Clip Transferring		
5	Loaded (3 rds) clip from transfer		
	station to cell one at hoist	0.96	9.8 ft/sec
6	Loaded (2 rds) clip from cell one at		
	hoist to cell two at hoist	0.81	2.6 ft/sec
7	Loaded (1 rd) clip from cell two at		
	hoist to cell three at hoist	0.84	2.6 ft/sec
8	Empty clip from cell three at hoist		
	to transfer station	1.04	11.6 ft/sec
	Hoist Operation (Short Flight)		
9	Empty Hoist Raised	1.17	14.1 ft/sec
10	Empty Hoist Lowered	1.00	16.9 ft/sec
11	Loaded Hoist Raised	1.24	13.8 ft/sec
12	Loaded Hoist Lowered	1.23	14.4 ft/sec
	Cradle Operation with Gun at 0° Elevation		2
13	Empty Cradle Raised	1.34	120°/sec 210°/sec 115°/sec
14	Empty Cradle Lowered	0.96	210°/sec
15	Loaded Cradle Raised	1.40	115°/sec
16	Loaded Cradle Lowered	0.93	200°/sec
17	Loaded Cradle Raised with Gun Elevating at 20°/sec		
	(motion relative to slide)	1.01	140°/sec
18	Loaded Cradle Raised with Gun		,
	Depressing at 20°/sec	1.32	100°/sec
	Breechblock Operation with Gun at	,	
	Battery		
	Opened	0.19	***
	Closed	0.29	
	Empty Case Fray Operation		
19	Empty Case Tray Lowered Empty	0.56	489°/sec
20	Empty Case Tray Raised Loaded	0.95	240°/sec
	Rammer Operation		
21	Extended Empty at 0° Elevation	0.66	21.1 ft/sec
22	Retracted at 0° Elevation	0.60	23.1 ft/sec
23	Extended Loaded at 0°30' Elevation	0.79	18.5 ft/sec
24	Extended Loaded at 65° Elevation	0.86	15.8 ft/sec

## TABLE F-1 (Continued)

Figure Number	Component Operation	Cycle Time (sec)	Maximum Velocity
	Hoist Pawl Positioner Operation		
	Extended	0.17	
	Retracted	0.17	
	Round Centering Snubber & Latch Operation		
	Extended	0.29	•-
	Retracted	0.29	

Figure I. Empty loader drum indexed one cell clockwise.

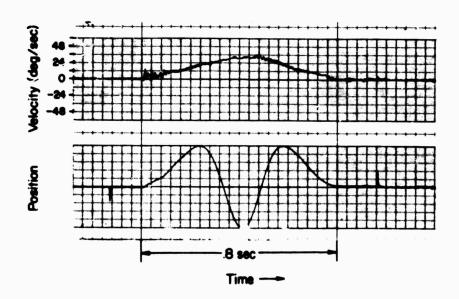


Figure 2. Empty loader drum indexed one cell counterclockwise.

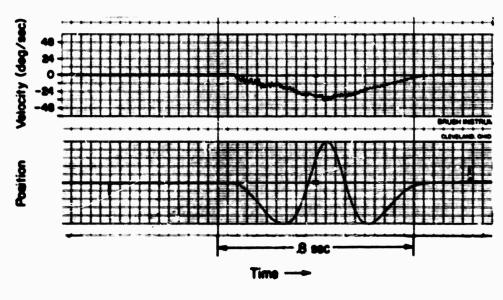


Figure 3. Drum loaded with 75 rounds indexed one cell clockwise.

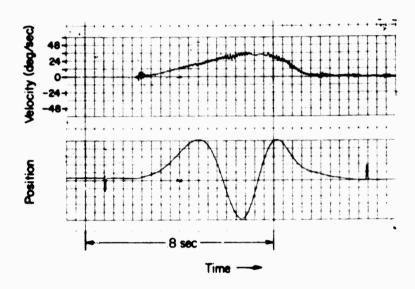


Figure 4. Drum looded with 75 rounds indexed one cell counterclockwise.

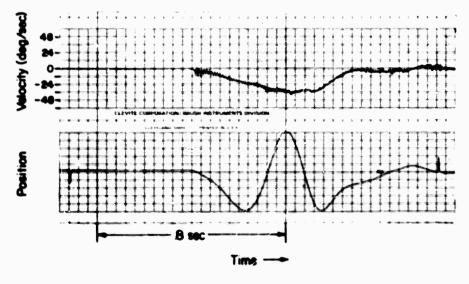


Figure 5. Clip transferred from transfer station to cell one at hoist with three rounds in the clip.

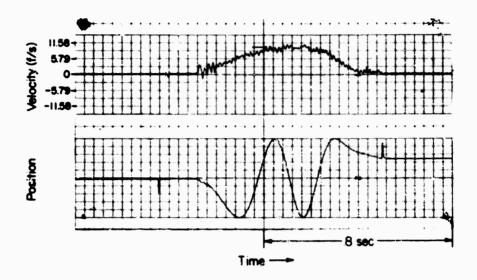


Figure 6. Clip transferred from cell one at hoist to cell two at hoist with two rounds in the clip.

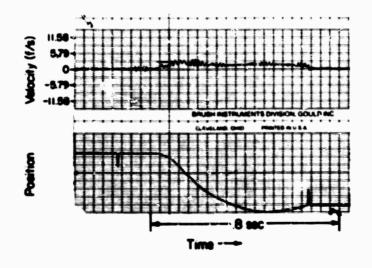


Figure 7. Clip transferred from cell two at hoist to cell three at hoist with one round in the clip.

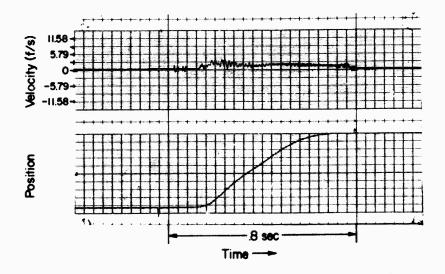


Figure 8. Empty clip transferred from cell three at hoist to clip at transfer station.

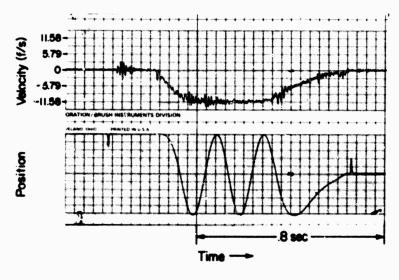


Figure 9. Empty hoist raised (Short Round Flight).

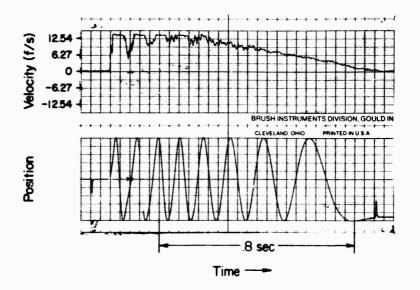


Figure IO. Empty hoist lowered (Short Round Flight).

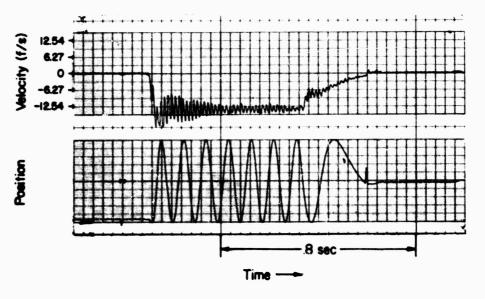


Figure I.J. Hoist raised loaded with a short round.

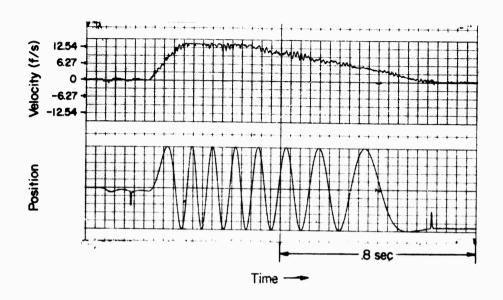


Figure 12. Hoist lowered with a short round.

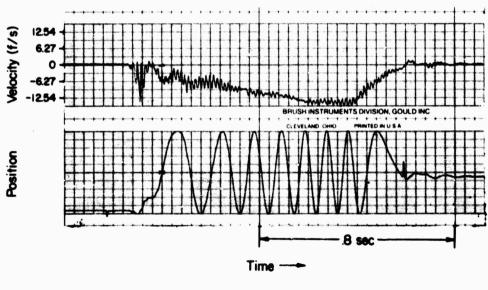


Figure 13. Empty cradle raised with the gun at O° elevation.

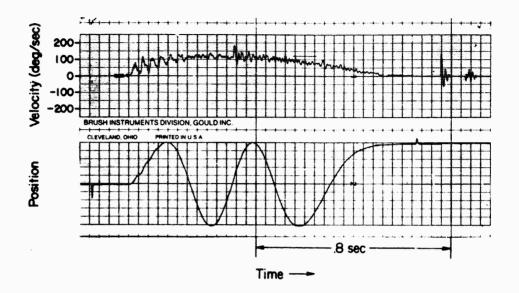


Figure 14. Empty cradle lowered with the gun at 0° elevation.

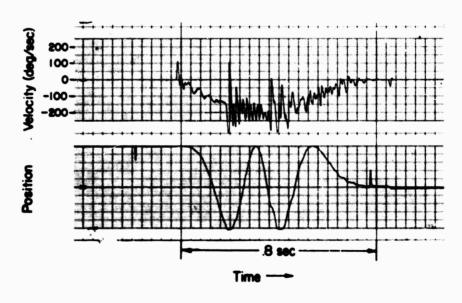


Figure 15. Loaded cradle raised with the gun at 0° elevation.

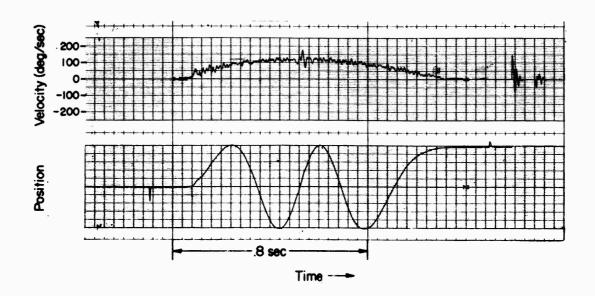
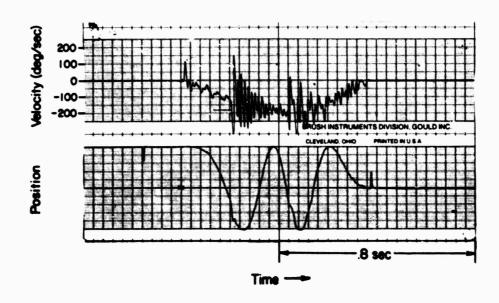
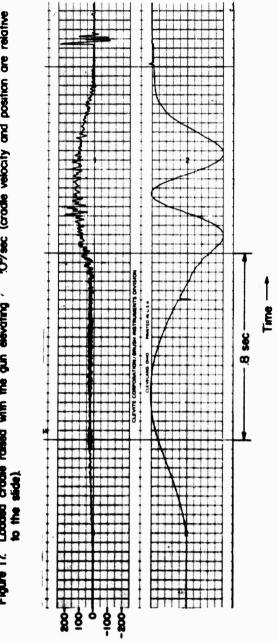


Figure 16. Loaded cradle lowered with the gun at O° elevation.



70% sec (cradle velocity and position are relative Loaded crade raised with the gun elevating ' to the stide).



Velocity

**Position** 

relative with the gun depressing at 20% sec (cradle velocity and position are Design Loaded cradle in the side). Figure 18.

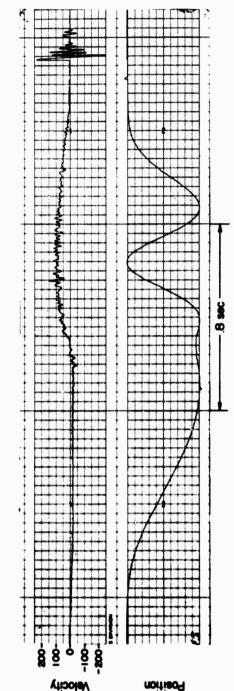


Figure 19. Empty case tray lowered.

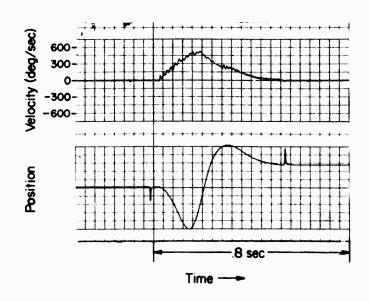
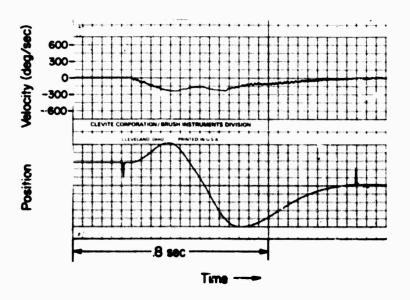


Figure 20. Loaded empty case tray lowered.



F-12

Figure 21. Rammer extended empty with the gun at 0° elevation.

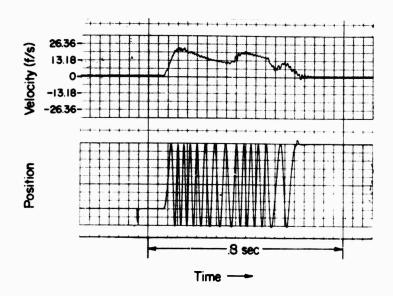


Figure 22. Rammer retracted with the gun at 0° elevation.

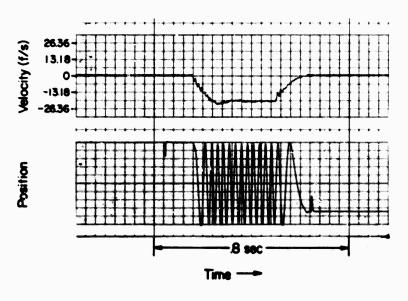


Figure 23. Rammer extended loaded with the gun at 0°30' elevation.

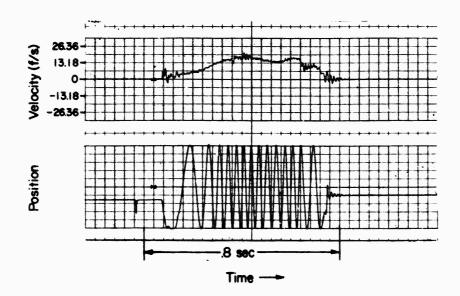
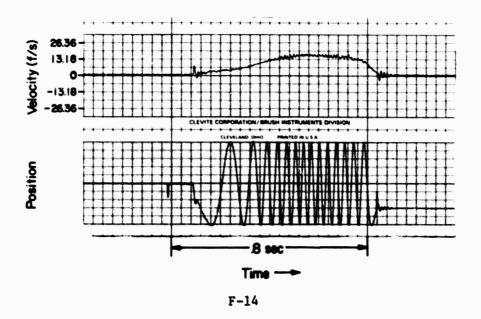


Figure 24. Rammer extended loaded with the gun at 65° elevation.



## APPENDIX G

Tabulated Firing Data
Tables G-1 Through G-3

TABLE G.1 8'755 MAJOR CALIBER LIGHTWEIGHT GUN MOUNT MARK 71 MOD 0

SUMMARY OF TEST DATA

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TABLE G-2 8"/55 MAJOR CALIBER LIGHTWEIGHT GUN MOUNT MARK 71 MOD 0 TABULATION OF BALLISTIC DATA

24.00			1271	27.0		***************************************		_	-	3.	2000	7.2 144	3777		1	10.00
2113	POSSELL	77 PC	-	201.11.11	100-	CAS 534	37747	-	·			Vila	/073	83	10 M C	PALLE
141	Ci		211.0	(		124.1 1Ti	1			7.0.7	5	(1/2)	14.50.1	18.47	(\$CA)	1705)
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		•	1 1 1		1			•		76.44					• •	• •
		•	1:31							45.5		2504			•	•
			15.	1 1						36.35	21.4	27.27			•	
		ŗ.								70.13	21	2:12	. · ·	H. T.	•	•
		•	~ .							623	15.4	7667	». T.	#. T.	•	•
		-	# :	# A D	3		=	H.4.	\$4811-4205	19.4. 1	16.9	2787		4.7.	•	•
		۰,	-							12.53	17.4	27-1		H.T.		
		•								*2.41	4.7	2766	. T.	#. T.	•	•
		• (								A 2 . 5.3	13.5	84.7	#.T.	-	•	
		•								42.21	. u .	2745	4.7.	H.7.	•	•
		•	4 4 1			,	,	,			17.5	2727	z. T.	H. T.	•	•
		<b>,,</b>		26 26	3	.115	<b>:</b>	, F	2266-11973			2240	и.т.		•	
		ا مد	-							67.30	и.т.	2768	#.T.	н. т.	•	•
		•	1							47.68		2273	*.T.	H.T.	•	•
		•	1616		2					67.60	<b>.</b>	2 2 5 2		H.T.	•	•
		•								47.00	H. T.	-522		H.T.	•	•
		•								67.00		27.19	H.T.		•	•
										47.11		2234	7.4	H. T.	•	
		•								67.64		72.1	4.7.	H. T.	•	•
		•								67.:3		5562	7.1.	H. T.	•	•
		2								47.92		75.2	7.1	M.T.	•	•
		=								67.46	#.T.	2 25.7	M.T.	H. T.	•	•
		2								47.88	H. 7.	4222	и.Т.	E. T.	•	•
2	Ţ	-	1118	:	25	E T	. 117	. 1.	Dect-111.3		n. T.		. L . K	#.T.	•	•
		٠,								.7.19	F. T.	.7.		H.T.	•	•
		-								67,46	<u>.</u>	H.T.	<b>*. T.</b>	H.T.	•	•
		•								67.79			#.T.		•	•
		•	,									1.1		,	•	•
	Ž	- (	• • •	:	200	*	=	-	KO4145J3-4		14.4	2 57 2	<u>:</u> :	 -:	•	•
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										19.20		75.9 #		Z.	•	•
		. •									¥.	7645		H. T.	•	•
		, ,								85.43		2546			•	•
		<b>,</b> :								42.63		6 2 2 3	E. 4		•	•
										P 1	» ·	414		# · ·		
		= :								47.5		5.16	٠ د خو	 	•	•
Ş	44	-	101	4	S. 724	. 184	. 3 : 2		SvCf-11373	12.43	. ,	26.38				

TABLE G-2 (Continued)

###### 10 40 10 10 10 10 10 10 10 10 10 10 10 10 10	71			1001	Š		BORE ENLARGET	142641	307E		950440	CH1480	18:1N.	H112215	CORPO	9900	PANGE
Charles   Color   Co	Ç	BARREL	400mg	1141	PLITATION	3C0H	MISTO	#JZZL	CAUSE		175154	(751)	V. L.		RANGE	SANGE	PANGE
111	=	Ex-400	¥0.	61450	(1) [ [ - 4 [ ] ]	FIRTO	( IM. )	(IN.)	(IN.)	JES 16.	14.053	3.5	(1/2)	CH. SQ. 1	(Y)S)	(Y0S)	(705)
### ### ### ### ### ### ### ### ### ##	i		•	•		****							-	7			
1872	7	-12	٨,	1614	: :	34.04	::	. 932		SPCF-11071	1 42.61	N. T.	2760	N.T.		•	•
110		•	-	1153							82.b3	K. T.	2695	7.1.	N. T.	•	•
1840 1840			ı	1827		01484					.2.63	₩.1.			N. T.	•	•
10.00 F. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.			<b>G</b>								12.53	#			z	•	•
1137			•								62.b3	. T.				•	•
197		•	•								#2.6T	H. T.		N. I.	N. T.	•	•
100			•								A 2 . 6 3	K. I.		N. T.		•	•
100 100 100 100 100 100 100 100 100 100			•	1010		20 OF			2	NOSIMAJ346		26.3		N.T.	¥	•	•
20			110	1927								7.16		-		•	•
20-11-11-17			1	35							77.40	7.5.7	2406			•	•
20-01 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			25	1947					9	CPCF-11071	47.03		2729	H.T.	N. T.	•	•
200	13/27	79-1	-	1437		RAPID	210.	200.		SPCF-1187	1 62.53	H. T.		N. II.	K. 1.	•	•
88.68 N.T. 2712 N.T. 2766			~								.5.63	11.T.		h. T.	N.T.		•
200			*								85.63	h. T.		N.T.	N. T.	•	•
## ## ## ## ## ## ## ## ## ## ## ## ##			•								69.20	N.T.		N. T.	N.T.	•	•
### 12			•								85.58	N.T.		M.T.	H.T.	•	•
### ### ### ### ### ### ### ### ### ##			•								15.64	H, T.		N. T.	N.T.	•	•
### ### ### ### ### ### ### ### ### ##											4.2.63	h. T.		N.T.	И.Т.	•	•
# 1			•									Z. 7.		N.T.	M.T.	•	•
### ### ### ### ### ### ### ### ### ##			•									Z.		z	N.T.	•	•
### ### ### ### ### ### ### ### ### ##			-											×. +.	¥	•	
### ### ### ### ### ### ### ### ### ##			r (								30.00		# · T ·			•	•
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20.604 R.T. SPCF-11.074 R.T. R.T. R.T. R.T. R.T. R.T. R.T. R.T											200	1		7		•	
## ## ## ## ## ## ## ## ## ## ## ## ##			2								67.64			N. T.	N. T.	•	•
200 64 March			23								29-29	N.T.	4.1	N.T.	N.T.	•	•
20-64 201- No. 1			22								年 よった 日	N. T.	z. T.		z	•	•
250 10 30 4.7. SPCF-11074 N.T. N.T. N.T. N.T. N.T. N.T. N.T. N.T			<b>.</b>								19.23	z.1.	z.1.		W. T.	•	•
20-01 10 30			2								42.63	r			. T.	•	•
190 00 00 00 00 00 00 00 00 00 00 00 00 0			Ľ									Z.1.	7.7		z.T.	•	•
	•		**	1350	2 2	GIOVA	.914			SPCF-11073			N. T.		N. T.	•	•
			~								35.67		N.T.	M.T.	м.т.		•
			•								97.4	¥.		H. T.		•	•
ZZZZ, ZZZZ, ZZZZZ, ZZZZZ, ZZZZZ,			J								12.61	z	¥			,	•
X.T. X.T. X.T. X.T. X.T. X.T. X.T. X.T.			•								29.28	#. <u>.</u>	Z.T.		N. T.	•	•
Z.1. Z.1. Z.1. Z.1. Z.1. Z.1. Z.1.			4								95.58	.T.	۲.۲			•	•
Z.T. Z.T. Z.T.											87.63					•	•
N.T. N.T. N.T.			•								A2.53	¥. T.		N.T.	N.T.	•	1
			•								42.63	N. 1.		z. 1.	ř.T.	•	•

TABLE G-2 (Continued)

BATE	7566	9	1001	200 200 200 200 200 200 200 200 200 200		#C2E 53		4007 E005004	là es	P 2 4 4 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	CH 24 98	7 2 UM.	FLASH	UNCORR.	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	RANGE
Ē	7. × 3.			00 Th - 97 C ?	-		(14.)	CIN.	JE: 16.	(LAS)	(66)	(6/3)	CM. SQ.1	(Y 0 S)	(YDS)	(YOS)
12/41	1-12	=		10 39	DIAVA	-314	. 903		SPCF-11073	92.63		, i	N. T.	. t		
		- 2 								12.63	, , ,	Z 2.	X X	Z Z	٠.	٠.
		7 3	14.1	20 39	215					12.64	7.	- -	Z .	×	•	•
		<b>:</b>	1536		01470				•	42.63	N. T.	. T.	h. T.	N. 1.	•	•
		<b>:</b> :								15.51			. (			
		-	1313						•	12.63						
		=	1526						•	82.53		-	, -			
		1								\$2.53	7.	7.7	N. T.	N. T	•	•
17.	26-8	* ~	1537	01 00	MG 13		746		F 14 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	42.04	F 2	N.T.	. c	Z 2	• •	
		~	1477							82.43	7	2.30	•		•	•
			1954						•	15.61	H. T.	9116		H. T.	•	•
		• (	1965						•	15.63	N. 7.	2738	•	H. T.	•	•
*****	:	٠.	2 6 6 7		i	1	•	1		62.53	¥:1	2713	•	Z		•
		<b>-</b> (%)			25	-	•		SPCF-11073 6	67.08	L I	2716	o c	 		
		-	1054							2.05		2724	9 0	. T.		•
			1617						•	95.63	R. T.	2731	•	N. T.	•	•
		•	1-19						•	95.63	N.7.	2757	•	H.T.	•	•
		••	271							5.63	# · L	9222	•	# :	•	•
****	-		7 6 6 7		2		•		A SALLANIZON	77.90	63.9	2827	٤,		•	•
	:	• ~	5:91				:			32.63		2719			•	
		~	1534						•	12.54	M.T.	2797	•	N.T.	•	•
			151							82.63	F 1	2737	•	N. T.	•	•
13/00	26-8	۰ -	N. I.	97 74	NO IS	1	A.T.	1,1	CPCF-11A77	47.63		11.7	4			, ,
		~								77.90	24.1		;		•	
21/01	24-1	•	٠		•	.824	100.	.T.					•	•	•	•
123	1-02	<b>96</b> (	1476	1.0	SLOW	200.	.093	#.T.	MOSTM&J3+6	.0.0ª	15.2	2325	H.T.	N. T.	•	•
		N M	5 . 4 . 4 . 4 . 4 . 4 . 4 . 4 . 4 . 4 .						4114	55.03	22.0	2449			• •	• •
		•	5151							75.03	4	2776			•	
		•	1554							14.00	23.2	2493			•	•
171	20-1		15.15	62 81	2107	. 611	-985	#.T.	HOSIMAJ346 6	65.34	15.8	H.T.	N. T.	ñ. T.	•	•
22/15	1-92	<b></b> (	*		St 94	M. T.	Ĭ.	¥.7	SPCF-1107-3 6	67.00		5540	N. 1.	N. T.		
		N, P	5051						•	67.00		6222	, ,	¥:	•	•
			1517									2220				
		*	61.1						• •	67.03	z	2739			•	•
		•	1510							7.30	N. 1	2236	N. T.	. T. X	•	
11/05	24-1	-4 6	1046	63 83	St. 0#	.005	-031	-4.67	SPCF-11073	65.63	H. H.	2721	R. T.	6712	9549	6512
		•	K ?							15.64		2750		6776	2659	6512

TABLE G-2 (Continued)

QAMGE CORP. 19UL (YOS) (			36.23
UNCORR. 24/05/05/05/05/05/05/05/05/05/05/05/05/05/	15687 15687 1687 1687 1687 1687 1687 1688 1688 1		
		22222222222222222222222222222222222222	
0			
			92.53
PONDER ON ORDER OFSIG. SPCF-11873	SPC7-11673	SPCF-11073	
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100 38C w 100 30C w 100 100 100 100 100 100 100 100 100 1	- M - O 	; ;	;
FIGURES AND SERVICE AND SERVIC	\$C04	SLOW AB WILL	
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00000000000000000000000000000000000000	\$ = \$, \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$		12
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11/12 11/12	•	3	

TABLE G-2 (Continued)

RANGE TAPLE RANGE (VCS)	15336	15136	15336	15736	15336	16336	15336	15336	15336	15336	16116	15336	15336	15336	15336	15336	15336	15 336	15336	15335	15336	14336	15336	15.16	14336	15.336	15336	15336	15336	4 2 3 4	15336	15336	15336	15336	15336	15336	15336	15570	15336
CANCE CYDS)	15450	15191	15444	15506	15528	15407	15465	1539	15654	15276	15127	15389	19357	155 Cb	15431	15301	15350	15601	154.30	15276	15321	154 \$	12451	12410	15362	15363	15325	15490	15397	40.4	1531+	15-07	152%	15246	15365	15 16	15.01	12461	15565
⊃ <b>p</b> ~	15175	15067	15213	15244	15138	15999	15161	15110	15350	26648	18117	15051	15062	15125	15068	15046	1524	14945	15423	15484	19195	15455	15 199	12851	15369	15290	15 137	15433	15-27	87731	15493	15542	15 14 3	15202	15153	15 337	15486	19332	15420
MUZZLE FLASH (UN/ CM.SG.)	-	9 7		•	6	263	0	3	1626	1626	14.28	1731	1788	1791	1626	•		•	9	•	9	٥	<b>•</b>	9 6 6	•	٥	•	216	- :		10+3	•	745	7.65	266	1175	415	2043	912
TPJN. VEL: (F/S)	2710	27.36	2715	27.12	2697	5142	5709	2799	27.00	66.42	2710	2703	2718	1692	2100	20.2	90	5772	2715	2732	2717	27.04	2776	17/2	2797	2707	2710	2702	2712		2779	2724	2714	5003	2681	27.05	60	2.7	2692
POESS. [15].	r. T.	, ; ; ;	-	H. T.	K. T.	R.T.			Z :				. T.	N.T.	¥.						#.T.	7. T.						÷:				H. T.	N.T.	2:	*	<b>2</b> :	, i		
392443 MCI'M 19017M	\$ 67.63	42.64	92.53	82.03	42.53	85.63	67.53	85.63	95.0	12.63	A 20.5 A	82.43	42.64	62.63	42.53			12.43	42.64	92.63	19.50	42.63	42.63	42.55	42.63	42.63	95.63	42.63	42.63		92.53	A2.03	42.63	45.63	95.54	82.23	\$2.63	26.08	42.63
POWDE* 149FK 04 26510.	SPCF-11873																Srck-110/3																						
#04E FP05ION 6#95E (IM-)	3															1																							
	200.															į	900																						
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63e1a	44710														i	2008	01472																						
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# 00.45	-	ی و	•	~	•	• ;		=:	2:	2 #	15	2	-	3	<b>.</b>	2	<b>–</b> (	٠.	٠.	•	•	•	• :	•	: =	닭	3	<b>4</b> :	2 :	-	=	=	2	Z:	25	52	23	Ci	:≈
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04TE FINED 1971	5																																						

TABLE G-2 (Continued)

The second second

**		
UNCOPR. RANGE		100
MUZZLE FLASH YUH/ CH-SG+)	10 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
Toun.		10 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
34		
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	E	
8025 6305 611:-1	· I	
2 1		
FIU170	## ## ## ## ## ## ## ## ## ## ## ## ##	
AODE Flord		
1424-9303 1021-8-513 169	2	
13081 1196 1196	•	411
POUND NO.	21222112221233333333333	:35722244222323233257;
1344 1344 1344 1344 1344 1344 1344 1344		
DATE FIRED 1971		

TABLE G-2 (Continued)

KANGE	TAPLE	2000		15336	15336	15336	25500	25500	25500	22530	25509	25590	25500	24500	25500	25500	25500	10557		13600		7	12470	4 3 4 5 5	2042	244	13463	13-80	17400	13403	13400	14490	13400	13400	13600	13500	13400	13400	13400	13400	13450	13400	14400	1.463	00451	13432	•
	200	(Alla)		15674	152.0	15325	26+33	54269	26232	26016	25920	26007	25152	26.134	25993	25934	68652	25493	10000	13332	1 16 96	2000	13261	43244	13242	16.0	13577	13467	13599	13695	13519	13536	13465	13640	13478	1 34 46	20.7	13457	13490	13429	* 166.	13446	1 1915	13366	13374	12821	•
	Urcher.	1007		15212	15167	15191	25661	25+33	26548	26677	25954	61792	55:92	26+35	26153	56159	26 321	25973	13663	13165	72.52	2000	1771	. 316	1376	1771	13211	13113	13145	12955	13121	14100	13267	13106	13194	1314	11956	12999	13020	12691	13115	11046	131.9	12092	15070	15-29	1 516 7
MUZZLE	F. L. S.	C. 50. 3	,	1043	390	7.45	c	0	n	0	•		5.22		766		0	0 (	•	<b>.</b>	•		5 6				261						•	0	0		<b>3</b>	•		012	•		3	c			224
		16/21		2641	2407	2633	25.65	2697	5092	27.25	2677	2711	26.10	2733	70.37	2692	1692	6992	2602	4 /07	2664	2000	25.00	46.00	26.84	96.00	2677	90.4	2404	2056	2673	2599	2698	2061	2576	6 4 4 2	2526	2653	5692	2656	5479	2679	7677	2461	2678	0146	(2)
1 4 1 2				Z. T.	H.7.	.ı. T.	۲. ۲			N. T.	¥.4	N.T.	N. T.	H. T.	K. T.		Z											Malla	- L	7.4	H. 7.	M.T.	H. T.	¥.↑.	¥.		¥ .	F :			×		N. T.	M.T.	¥:		
		1262		_	_	_				47.53	12.63	12.61	R2.63	15.54	42.43	42.63	47.63	42.63		20.78	2 4 6 8	10.00	46.55		4.64	24.6	82.61	27.63	42.03	12.51	42.63	42.53	. 25.20	42.61	15.0	45.63	19.20	10.64	42.53	47.63	42.63	42.03	42.63	19.58	45.6	85.03	12.63
PORTE	1:05						-3-17 SPCF-11073												2011-17-12																												
		14.5		-4.45			-3.17																																								
105477		17.00		.034			. 113											•	•																												
14525763 3108	,			. 411			. 3 . 8												•																												
	7.02	F.L.		A.1010			Cleve																																								
	6.57875.3	10.0-0.01		10 61			27 34											:																													
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TABLE G-2 (Continued)

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SPCF-11075 (175) (			-	1004	8		P34E ENC125HT	T-10.00	Aras FROSTON	POWIER	C44.3	Cubing Darss	1400	FL459	UN 1378.	CORK.	PANUE
1,	CHOCK STATE CHICK	CHOCK STATE CHICK	107 CT L A E 4 C C A	107 C 4 E 4 C 10	-		×10180	272262	470.47	Ĉ	111111	(15:3)	VE	1047	444SE	STACE	35745
4.*. 4.T. SPGF-11077 92-65 N.T. 26-70 0 13-72 13-20 13-72 13	CARLES CARLES CONTROL	MO. PLANT CATALON	CANALON SINAMA	14:4-5 L	1			114.1	-	56.515.	15,131	S	[6/3]	C*. S.*.	CLOSE	(YDS)	(488)
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## 1975   1.1   2717   25576   25976	<b>~</b>	,	,	:	•		:	:					20.98		E	•	•
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-817 -2.78 SPCF-11073 N.T. 2670 0 25713 26127 26	•														2007	62292	01667
### 1975   1975   2544   2644   2614													26.70		25.11	76.42	2000
### 1975   \$25513   \$25013   \$25015   \$	~1	~-									42.53		2645		95157	20217	25530
0.2.67   W.T.   2664   522   26472   26157	-	-									47.64	2. 7.	27:5		25610	263.5	25592
### 1	•	-									62.03	Z. 7.	2689		8/586	26157	25500
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82.65 82	Ξ	_									17.63	1692	1843	67/52	26084	25599
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27 33 80010 W.T. W.T. Spip-11073 77-0 87-6 877- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177- 82-65 3177-	~										42.63	7641	•	25069	56419	25500
Strace St	ĸ.	_									15.63	2651	7.45	59162	26114	25500
27 23 Repig M.T. M.T. Spif-11073 A.T. 82.73 A.T. 82.53 M.T. 82.53	2										85.6	2697	225	0.412	26111	0142
27 33 Repto W.T. W.T. Spif-11073 A7-63 W.T. 62-63 W.T.	2										A2.63	2693	• !	25561	26.0%	2550
27 23 80010 W.T. W.T. W.T. SPOTF-11073 A7.03 ATT. A2.63		_									15.5	1607	7.85	22627	26036	222
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· -	•	_									42.63	27.32	• (	557	26167	2558
	•										42.53	1047	-	25136	6.662	25599
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PANGE				26830			•	25500		25500		23412																																22200
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	PACOND	SAMO	1323	75195	25.3.59	26933	25070	24630	26777	246.5	N. T.	22700	1-126	61622	24374	23119	71012	23338	22766	23947	233+5	24196	22013	21131	23316	61922	71754	27435	21012	2330m	23065	22 465	23175	23372	23756	23076	22.8.	24227	24074	29/22	22540	2272	2527	22332
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	1 0 0 T	E 2 C 2		1 32.61	82.54	12.50	17.63	47.57	14.50	\$2.6.5	46.07	66.97	66.07	,,,,,		. A. 37	44.67	46.07	55.67	66.87	66.07	\$6.07	66.87			76.07	,	6.9	\$6.07	66.37	, t		, v. v.	66.37	66.07	66.37	66.37		-	15.67	15.6	7.63	9.7	82.63
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RANGE	TABLE	PANCE	(405)		26276	12230	00222	27200	22230	27200	\$2225	27433	22200	22200	22200	22260	86224	0 2 2 2 2	25,75	22233	2324	2270	27208	22738	22200	27-110	22200	22200	22203	22200	•	•	•	21017	71917	21917	21937	71917	41616	72415	21015	21917	21937	21937	21937	222	16473
	CORP.	AANGE	(ACS)		55739	22563	224%	22354	55527	22401	27384	22515	22636	75895	22651	227 16	94.422	61977	22693	2265	276.RG	22561	22777	224.80	\$1522	21922	22684	22717	72567	22731	ļ	•	•	1766	22505	22342	42227	22019	22572	22102	19:22	55504	25362	22117	29242	00000	, , , ,
	CHCGGA	PANCE	(SOA)	•	55733	22221	2235u	22 316	22618	22353	22762	97574	17422	20222	25677	23053	57922	17977	22106	22696	22642	27545	22659	12330	16722	22376	14922	\$6722	22.29	22167	21412	21847	25.177	226.0	22122	22327	21349	21145	22217	21439	21737	21932	218-6	15112	21 926	11524	
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SCHOT L																																												¥.			
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TABLE G-2 (Continued)

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		151	-	84.54	42.64	47.6	87.61	47.67	42.53	42.6	12.63	42.63	A 7 . 6 5	15.05	35.56	45.64	2.6			42.63	42.6	47.6	42.63	15.61	42.03	15.63	45.6	. 0.	47.64	14.5	12.6	47.64	45.4	.2.01	15.04	42.03		17.43	19.20	12.63	42.63	A. 6.	5.2		
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	44.5	3	•	-	,	<b>.</b>	**	7	•	;	:	;	•	7	1	3 ;	3 ;	3	: :	: 2	-	~	-	2	ť			-	٠,	۰ 4	-	•	•	<u>:</u>	90 F3	<u>م</u>	<b>:</b>	1	<u>.</u>	••		••	,		
	Stewel.	Cen-17		<b>7:-1</b>																																				76-1					
91.40	6341	1471	•	15/11																						12/16														12/16					

	1001	5		BOLF CHESTON	שטלב כאל ושכעו	100 TON	PONCY R	CHANGE	CHAMARA	A D.J.N.	PUZZLE FLASH	UNCOKA	_	TABLE
į	FIRES	10.3k-9.303	• •	64161w (14.)	Galdin - 122ce	6106F		ME1647	(151)	VÉL. (F/5)	C4. SQ. 1	ZANGE (YOS)	TANGE	RANGE
		0\$c	01.73		;	.1.	5005-11073	1 42.53		7.7		N. T.		
									7.1			7.7	•	•
_								15.63		¥.	H.T.	:	•	•
								A2.64	2	Z.		H. T.	•	•
								12.03			, T	F. 4.	•	•
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_								19.26	. T.	R. T.	M. T.	H. T.	•	•
	3	24 10						C	#. T.	2146	c.	15622	22163	22100
								66.87	<b>.</b>	4173	525	22879	22236	22100
								0.90	. T.	2158	422	72427	22218	22180
								16-37	:	2 1 4 3	***	23365	22339	22100
_			•					15.47		2154	1175	22632	22128	22180
								26.44	H. T.	2147	484	22716	22115	27103
•								46.17	<b>#</b> .T.	8512	91.2	22422	22130	22189
_								46.07	T . T	2164	-	22794	72160	22330
								16.42	1.1	2154	7.85	22763	22967	22108
_								66.87		2171	215	22667	22839	2218
_								64.87	7.	2170	502	22935	22319	27188
								66.39	H. T.	5912	•	04874	22284	22180
_								16.87	#. 4·	2174		22131	69022	22138
_								46.37	#.T.	2165	644	35622	22378	22100
_	1		25.4					<b>*6.0</b> *	#.T.	2610	•	2307-	28222	22100
<b></b> 1	3		44710						H-1	2171	216	22919	22271	2216
										2156	-	25922	52189	22188
								10.00		71cH	•	22610	22110	22138
									:-	7140	745	23847	22235	22108
								64.87		2141	10.3	23026	22256	22149
								66.37	:	2150	P 7 C .	22036	27270	22188
								10.04	:	- 1 L	6.4	55422	22254	2712
								65.37		7171		51467	\$225	22:33
								20.00		9717	224	15.22	22163	
										217		6100	67527	20127
											•	66.02	20122	86177
								40.07		9614	1343	81624	22116	69.7.
										7154	2	96.74	27272	66122
								1000		512	12)	22730	32726	22130
	1									2445	5	22450	24123	22133
			R.J	14.	. 520	-2.67	-2.47 SPUF-1117/3	-		# F 1 F C-	0	5	•	•
	58 31		E 1 2					, ,	-:	3154	225	13659	20224	25432
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						•		af. 37		7145	د	20147	21376	26626
_								6.5.		B * 1.	795	20129	21:14	25.23.2
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という こうしゅう かんしゅう かんしゅう しゅうしゅう

を 1915年 19 5757 一番時間所見 生間は最高過程 1950年 195

						8C2 F	1 34547	18.30	# JC 904		494.47		*U771.e			Range
1			10201	200				NO LYCHA	14061	CHERGE	P. E. C.S.	TRUN.	F. 85-	IMCOVA.	CCOR.	7.40. E
	44. FL	CANON	=======================================	FL TWENTON	#0 X	P12140	377.0	SAUCE	*0	4,7,3	(151)	V.L.	1801	RANGE	3.7MT.	PANGE
141	00# -17	į	おいまいる	1266-4181	07213	1111	1.71	(18.)	S.E.	(trs)	נירו	(5/3)	7.53.5	1172	CLOAY	1SUA?
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		=								65.17		2156	•	2-304	21233	26629
		~								46.27		2114	41.2	94742	21500	20620
		7								66.37	H. F.	2126	•	19116	21112	20620
		-								65.27		2160	9	2011.7	7 21 15	20623
		=								44.27		71:1	-	20142	21777	23020
12/21	1-62	-	411	1.0 0.0	SL 0*	N. 4.	4.1.	W.T.	S-CF-110FF	16.33 F	¥.*	N.T.	H.T.	W. T.	•	•
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12: Institutzanten caline.

13: Desire of least ingly as of lecate attain a quiscusoic forceauce.

13: Desire of Statistic (Leaston.

1ABLE 6-3 B'/56 MAJOR CALIBER LIGHTWEIGHT GUN MOUNT WARK 71 MOD 0 TABULATION OF MOUNT DATA

								ودرون و	CELEMBA COUNTR CYLLYCH	COUNTR C	1021			COUNT?		
2112			35.			25.5		511115		4 1 4 4 4		96.1	1100	110216	100	INTER/BI
531	134296		tat satton	*67	33/943	1121			¥655.	P4£45.	47.55		1134	1.411	11.0	S CHOOL
2	5	9	136 518	(111)	342	12.	18/31	1320	ž	11301	(PSI)		13:51	(Src)	(SEC)	(SEC)
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		•				;	2751		#: <u>-</u> .		# · T ·	H.7.	#· -	1.1.	н. т.	
		•				:	27.5		#.T.	H.1.	.T.		H. T.		H.T.	4.11
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		• (				7	1722						-		#.T.	<b>3.</b> 11
		•				-	× ×			H.T.	#. T.		#. -		H. T.	S. :
		=				-	7577	-	#. T.	H.T.		7.4	H. T.	#.T.	#.T.	4.10
		5					4567	-	-		-	. <del>.</del> .	:	<b>.</b> .	H.T.	29.6
1		-					0.22				# · 4 ·			# · ·		S. 1
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		٠.											H . T		A. T.	
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5	78-1	-	?	\$ C Day			44									
		•				.1.1	8112			-			1	M. T.		•
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į	-2	-	25 62	ž	1		2717								¥.1.	
		№ 1						-					H. 4		X. T.	•
		<b>,</b>					57.46					¥.4	. 7.		#. T	•
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		4 •					260			H. T.					# · 4	K. 12
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						AC27	1. 4.74		4 3 C 7 O C		200		#UZ ZUF			Kayer
DATE			1836 .	100				-	x3CH1	CHARGE	.Soite	TRUM.	FLAS4	UNCORR.	COPR.	TABLE
FIRE	13454	59	Jen 3 2	FLEVATION	M03E	ORIGIN	MUTZLE		0.5	THOTOR	4131)	V56.	(10)	RANGE	BUNGO	PANGE
1471	E H- MO	-01	FTRED	(DEG-4134)	FIRED	(14.1)	(14.) (17.)	(TK.)	9.516.	(LES)	CC.	(£/S)	C+. SQ.)	1201)	(Y07)	(AUA)
												*				
43/22	24-1	•		60 03	SAPID	.14.			SPCF-1107	\$ 65.37	, z	2:54	7	23159	21091	25956
	:									65.37	,	2:43	cə	20:71	2:2:0	20620
		-								56.37	.1.	2156	0	20117	2117	02936
		:								66.07		2136	•	25004	21230	20620
		: 2								66.07	× ×	2116	912	21176	21588	20620
										66.97	7.1	2125	0	19416	21112	20620
		1								65.37	#1. T.	2146	•	20147	731157	25623
		, ,								46.37	7	2134	c)	20182	21777	23028
12/21	28-1	;	9168	1.9.20	St. Ow	M.T.	,	N. T.	SPCF-1187		N. 4.	N.T.		*. ₹	•	
72/23	7-12	٠.	•			.0.	.83	2.75	•		•			¥.	•	•

MOTES (1) VELOCITY FOLLS DAMAGED BY BPEAKUP OF PRIJECTILE MINDSMIELD.

(2) INSTRUMENTATION FAILLNES.

(1) POINT OF THACTY FAILURE.

(3) POINT OF THACTY FAILURE.

(4) OFC. -- OSTILATIVE ELEMATION.

(2) M.T. -- NOT TACEN.

G-17

TABLE G-3 (Continued)

								מיהחוו נ	ACPLIAG	COUNT.	6,4,0,0			4.NOC.		:
11.0			1 5			A						, ,	3 6	31916		4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4
0343	3888	* Dissi	SCIITA STA	305.	CMBPSE	01810		Post 5.5.	. SS 200	PPESS.	PRESS	0157	1104	C.		S0:000
1371	EX-#3	M.C.	1265-4182	C 1613	TYPE	S	15/	(154)	(281)	(PS:1	150	(IN.)	1561	10351	(235)	(SEC)
	2-12	<b>N</b> (	2	SC.38	1		27.7		<b>X</b>	- I		ž ;	· !			
							25.45				2 2	2 2				, ,
				77.10											-	
		f. u					27 78					. 2		7 2	2	
							5112				2	7	2	Z	1	
							2718	7	-					K.T.	7	N. T.
		•		\$10K	P. 00F	7.42	76 62	H. T.		H. T.	N . T .	Z . T	N. T.	N. T.		•
		2				23.4	2485	H. T.	#. T.	H. T.	ж. т.	. T	N.1.	7.7	Z	:
		=				21.7	2486	#.T.		H. T.	N. T.	7. T.	N . T .	. T. W	¥. 4	:
		21			ברו		2728	H		¥. T.	¥.		ž.	2	¥.1	•
727	21-1		7 D	C148	Į		1	1563	8025	2140	35,30	31.2	.155	. 330	0.00	
		<b>~</b> 1					24	\$ 10 mg	7.5	2143	500	11.0	170	. 310	€;	0.58.4
		m.					2112	771	9150	9602	3558	32.6	2 1	. 32.9		9.17
		• (					57.15	1 2 4	9	2000	1560	35.6	-165	. 319		
		<b>.</b>					***			D C		2.5	0	. 350		2000
		• •						1 1 2		2000	934	7			2	26.00
						: 1	****	200		2000	2000					140.07
		• •					27.4	1458	177	2008	386.0	12.6	071		- 1-	18.296
		. 5					:	1.35.0	464	200	35.0	32.4	175	348	N. T.	7.500
		=				H. T.	T.	1350	5110	2191	3550	32.4	.170	.340	N.T.	7.320
		17				H.T.	.1.	1350	\$110	1612	3568	31.8	.178	. 140	.035	
		P.				H. T.		1301	••••	2090	3568	31.8	.170	. 530	.033	4.798
		:				4.1.	H.T.	1300	4364	1672	3569	32.4	.170	.310	. 033	5.100
		51						1300	01.4	2092	3>60	32.4	.170	.305	. 0.37	16.630
		2						1358	7 6 3 1	2390	1560	32.4	.170	. 135	. 0.32	63.010
		2:				÷ ;		1450	201	2002	0000	32.	971	0 2 2 0	0 .	31.460
								200				30.0		9 6 7		7.14.0
		. 2							;	2099	3568	32.6	1 20	330	. 032	7.228
		2				H. T.		1358		2040	3568	45.62	.160	.315	.030	18.620
		٧.				×.T.	#.T.	1458	0069	2994	3550	32.4	.160	.325	.037	24.510
		2				#: <u>.</u>		100	0764	2092	1468	7.25	. 170	. 125	.030	5.080
		2					#.T.	1.00		9602	3569	32.6	. 162	.350	.036	7.400
	1	2						1 16 9	-	2090	0 9 9	4.25	170	. 365	. 036	7.230
ž	7		=	0147	ž	-		1578	205	2178	3.00	31.4	.165	. 330	9	•
		~						141	-	2188	3504	31.4	.170	350	.030	4.670
		<b>80</b>				-		1,1	1215	2100	24.00	11.6	.170	. 5.30	. 030	96.
								1410		2148	3500		.170		250.	7.650
		٠ ١								212	200	35.5		5.15	3 6	
		• •								200		75.5		200		
						-				2012		25.55			200	
		• (					:	191			***	20.7	B (A		100	200
		•				:				••••		36.56				27.0

								_	CY1 THUR	Countra	CAL TAGE			COMME		
						CHANGE						FAX.	PECOIL	RECOLL		THTERVAL
DATE F1850	24646		204	300	Cuberr		13CH	STATIC	PAX.	STATIC	X C	RECOIL	DU24-	DUCA-	f JFCT.	METHERN
1471	E1-100	.0	(36,464)	61960	TABE	5.5	(6/5)	(PSI)	(PST)	(PSI)	(PSI)	12:	(SEC)	(SEC)	(SEC)	(SEC.)
2/1				01414	1		K	1410		2104	3500	32.2	0.1	0.4	. 030	9.250
		::							200	20.0	200	36.2				0.00
			22 00	700					200	2.12	33.00	36.6	.170			0.4
		: :		01474		2		1623		2219	37.20	12.7	170	130	0.0	
		15				¥	H . T	1454	9569	2218	3728	32.2	170	368	.938	4.560
		:				H. T.	#. T	1691	0469	2210	3720	32.2	170	04.	. 829	
		11				;	¥.T.	1651	6950	2213	3720	32.2	.175	. 340	. 032	4.680
		<b>=</b> (				-		15.5	4950	2219	37 40	12.2	.175	.330	.030	
		F :						1410	690	2219	37.28	32.2	. 180	. 335	.00	4.590
	•	N	:			* :	- :	2410	1950	2210	3726	32.2	. 1 8 0	.330	.033	6.940
		- 0	7		יונר	: :	96.7	1500	0 1 4	922	27.70	e e	C.		126.	
		-					2776			2260	9 6 6 6				2 6	• 1
						-	279	1991	5.25	2260	37.50	30.2	163	125	. 031	•
		*					2743	244	8200	2263	1710	30.2	.170	. 320	. 030	
175	20-1	=	22	\$104	44.4ED.	# · ·	2316	1510	1540	2160	7320	2.92	.165	.315	210.	•
		a.			FULL		2751	1649	5170	2160	3648	30.9	.170	.315	.030	
		jr				<u>.</u>	2724	1513	5350	2140	3610	33.9	.170	303	.030	•
		• 1				÷ :	2736	1040	9170	2200	3658	11.0	.170	567.	. 032	•
		<b>,</b>					5642	16.1	\$ 300	2200	3698	31.6	.165	. 329	. 035	•
		•			!		27.26	191	5100	2220	35.98	37.0	.170	. 325	. 029	•
			:		100	23.4	2427	6991	24.30	2249	36.90	31.8	•165	.355	.028	•
		,, f		210#	į		91	154	215	2170	3630	6.0	.160	525	920.	
		<b>.</b> ,				-	27.30	0161	000	2167	35.10	3.5		200		•
		٠.				•			2070	2017	26.50		2 1			•
						;			200	0017						•
10/01	24-8	۰.		40.0	FIRE		1	725	8260	2000	96.55		27.		028	• (
		•	}		PROOF	26.1	7.	1628	5340	2090	3570	31.0	170	.355	. 028	•
7	1	•	•	•	•		٠,	¥.1	¥. 4	M.T.	#.#	W.T.		+	4.4	ż
1613	1-12	-1	10 01	Stow	PAF.SER.	14.2	2375	1593	15 30	2160	3340	73.6	.170	. 360	.030	•
		•				15.9	6**2	1548	3410	2250	3560	25.5	.175	. 350	.041	•
		м,					25.52	1569	2 1	2240	1617	25.6	-170		.030	•
		, ,				***	27.36			D W 2	37.50	9.0	0/1-		520	
4.74		۰ م	:	30.0	-		2 4 1 2	1,70		877	2	,	. 103	• 1		
		٠.	: :											972	22.0	• •
		• ~	:				2229		25	33		(1)		(1)		
		. ~				1	2298	1679	11.10	2243	3660	25.	0.150	330	92.0	
						,	2278	1579	31.36	2243	3450	25.7	1.0	440	0.0	•
		•				-	2230	151	31.90	2260	3460	75.7	.175	. 100	.033	•
		•				. T.	2236	1578	31 90	2200	3460	74.7	.170	.300	.031	•
11/15	7-12	<b></b> -	-	2105	בונ	-	27.21	1541	-11	1622	3710	20.6	.178	.330	820°	•
							2750	1649	21.00	2240	3750	28.4	.165	.330	. 031	

								RECOTL	CYL TNUR	COUNTR	CYLINA			COUNTR		٠
						CALLO			******	•		MAX.	PECOIL	AECOIL		INTERVAL
1			20.5	1	,	0912Pe	- TOP	STATIC	"AK"	STATTC	TAK.	RECOIL	OURA-	DU&A-	EJECT.	PETMEEN
1 144			KOLLAN TO	100	24.60	121	VEt.	PACSS.	04E SS.	Parss.	PDESS	U152L.	1 I I I	TION	7 I ME	ROUNDS
1261	E H- MOD	2	(A( 3( )	03214	110	5	(5,5)	11541	13541	(PSI)	(PST)	CIN.	(SEC)	(SEC)	(SEC)	(SEC)
					•											
29/11	1-62	m	-	30.75	FULL		2729	16.8	0.64	2290	4740	24.4	.165	. 100	. 028	•
		•		2010		;	27.16	1563	4900	2220	3646	28.4	.170	.305	.030	•
		<b>.</b>					2704	1474	4410	2400	3540	28.4	.170	.305	.026	4.769
		٠					27.17	147	111	2199	3640	28.4	.175	. 310	.028	5.110
	٠	•				H.T.	2588	1478	4670	2163	3640	28.7	.170	. 715	. 02A	5.140
		•				H.T.	2703	14.78	4760	2150	3648	24.7	.170	. 324	.076	5.100
		•				H. T.	1632	3497	4730	2160	36 40	7.9.7	.175	.315	. 028	4.995
		:				H. T.	27.82	1470	67.40	2163	3640	24.1	.165	. 345	.030	065.4
		=					25.45	1440	47.30	2159	36.0	29.1	.170	. 355	.025	4.925
		2				H. T.	2710	1440	4760	2160	3640	23.1	.175	. 350	. 025	4.970
		~					27.15	111	4780	2163	3640	29.1	.170	.370	920.	4.985
		=				.1.	5092	1441	6768	2164	2992	29.1	.175	.370	. 025	4.945
:	;	15				H.T.	26.84	1.40	4768	2160	3640	29.1	.170	. 385	3	N. T.
11/1:	1-02	-	<u></u>	\$1.04	450.03	N.T.	9002	1544	00 97	2168	3030	22.1	.175	. 335	. 239	
		a.				4.T.	2172	16+5	2970	2173	3090	23.6	. 165	.350	.020	
		m					2140	1649	3120	2173	41.0	24.5	.130	. 130	• 025	
		•				-	2253	1645	3320	2170	3170	25.6	.173	.335	. 025	•
		ю.					55.28	1645	36.60	2700	3230	56.5	.170	.340	. 025	•
		•					2334	7 64 6	36.10	2203	3230	56.5	.170	.330	. 022	•
			20 02	GIATA	FULL	#.±	2717	1578	5100	2130	3640	31.8	.185	340	. 930	•
		•				;	2442	1-28	166*	21.0	1680	32.1	.180	.380	.030	4.830
		•				,	2715	1421	6993	2133	3640	32.4	.175	.380	.030	5 . BoF
		*					2714	1420	1990	2130	3643	15.7	.170	. 185	.031	4.975
		= 1					2712	1+20	1667	2130	3590	32.7	.143	.340	. 078	5.020
		21					57.39	1623	2664	2133	3643	12.7	. 165	. 383	.031	\$66.4
		2 ;					27.05	1420	4930	2130	3643	32.7	.140	. 396	.078	4.959
		2					2798	1420	4930	2133	3640	32.7	.185	. 390	.031	5, 002
		2				7	3100	1676	4938	2133	3680	32.7	. 180	007.	.030	5.000
		::					2714	1429	000	2136	3549	32.7	.167	004.	.031	5.073
				100			40.7	1460	200	L2 17	35.43	32.4	67.	.360	620.	. :
11786	24-1		9.1 2.9	0.00	First 1		7	1473	7 4		200	**>	1		0 70 0	
		٠,				3			2 4	7.5	4 4				3	4. 46.7
		, 17					2	014	0561	2110	14.10	41.5	5 2 5 7	4	3	5.217
								14.0	6000	2119	161.9	4 4 5	170	44.	3	F. 125
		ď				,			650	21.10	46.8			0 2	::	5.11.3
		£				,	7	1613	4950	2133	1610	4.5	177		: 5	5.167
		٠.						1411	6950	2139	36 50		1175	365	3	5,115
		4				7	1	1410	506	2133	3610	32.1	17.5	150	: 3	5.130
		F						1+10	4950	71.50	3614	32.1	.175	.355	9	5, 155
		-1				N. T.	4.7.	1410	4950	2130	3610	32.1	.175	.365	13	5.109
		11				A.T.	7.	1413	4950	2130	3610	32.1	.175	.370	3	5.153
		12				. T. X	÷:	1380	5060	2130	3610	32.4	.165	.390	3	5.060
11/13	24-1		:	01 dr	FULL	.T.	1692	1457	5110	2233	3640	30.3	.160	.390	620.	•
		•				;	27.16	1620	0115	2200	3670	31.5	.190	.390	.030	4.300

								Prente	FCOIL CYLINDR COUNTA	COUNTR	CYLIND			COUNTR		
1						CHENIC						· A N ·	RECOIL	DECOIL		INTERVAL
1			250			PEFSS.	T &UN.	STATIC	464.	STATEC	HAX.	RE CO IL	DJRA-	JURA-	EJFCT.	BETHERN
1	13446	Carried A	FLEVATION.	3COK	CMBAGE	(121)	VEL.	Parss.	P 36.55.	FRESS.	Parss.	O1SPL.	TION	NCIL	TIME	RUUNIS
111	000-13	Š	(3FG-AT4)	FIRED	TVPE	(5.3)	(F/S)	(PSI)	(150)	(PS:1	1051)	(IM.)	(SEC)	(\$60)	(SEC)	(SEC)
	-				•	****										
1	20-1	ør,		CIATA	FULL	M.T.	2719	1621	5148	2200	3670	31.5	.180	.370	.031	5.105
		J				4.1	27.16	1520	5140	2200	3078	31.5	.170	.370	020	5.180
		*				¥.7.	2643	1020	\$020	2200	3670	31.5	.175	.380	.030	4.980
		•				. T.	2715	1623	5040	2177	3670	31.5	.180	.370	620.	4.975
		٨.					2112	1537	£020	2160	3673	31.5	.164	184	.031	5.06t
		•				A.1.	1691	1563	4960	2160	3550	31.2	.182	.401	.031	4.986
		r					50.92	1534	4410	2160	3558	11.2	.166	.437	.032	5.092
		=					5942	1563	9667	2163	3650	31.5	.188	.405	.031	5.030
		=				F. E	2709	1430	4960	2160	3650	21.5	.172	. 413	0.0.	4.975
		~					27 09	1530	5780	2100	1679	31.5	.179	.411	. 030	5.069
		7				-	516	1513	5020	2160	3670	31.5	.176	.425	.031	5.156
		2				#. T.	1692	1478	1967	2163	3670	21.8	.151	26	. 034	5.062
		51				#.T.	2718	15+0	1967	2170	3678	31.8	.177	. 423	.034	5.043
		2					2783	1477	0667	2160	3678	31.5	.161	.431	.033	5.158
						H.T.	2740	1500		7168	3470	31.0	.157	674.	. 035	5.076
		2				x.1.	2640	1530	7,7,	2168	3640	31.5	.161	. 661	.028	5.072
		-					2700	1530	5050	2160	36.98	41.6	.153	. 447	. 932	5.159
		2		*10*			2702	1738	5010	2240	3730	30.9	.180	.345	. 027	•
2171	1-27	-	=======================================	44F	fur		2640	1441	5000	2118	3600	30.4	.165	.395	.032	•
		~				# ·	2775	1-50	6478	2110	3500	31.4	.165	. 340	.034	4.855
		~				*	27.07	1488	6.6.5	2118	1526	31.4	.170	.380	.033	5.005
		•				#.T.	5142	1630	*367	2890	3549	31.7	.163	.390	.034	5.156
		•				# · ·	2732	1650	5000	2048	3608	31.7	.160	.372	.032	5.157
		•				# · T	2717	1410	603	2888	1550	32.0	.180	. 369	.032	5.078
		-				.T.	2706	1691	4969	2140	3550	37.0	.172	. 348	.031	5.045
		•					2786	2642	957	2048	3400	31.7	.169	. 395	.032	4.934
		•				# · T	2771	1489	4690	P+02	3 = 3 @	32.3	.170	. 400	. 030	4.066
							2710	14.00	4490	2940	3<50	32.3	.171	.491	.037	5.150
		= :					2042	11.0	£ 4 6 8	2390	3578	32.0	.172	904.	.035	5.039
		71				-	24.2	2 6 6 5	6920	2040	3540	32.3	.173	.415	. 030	2.047
		2 ;					2710	1650	200	1 0 N	3590	32.0	174	. 617	200	5.138
		2 1				-	20.2	1		0002	3590	31.7	178	1690	620.	5.037
		£ :					2112			6662	357	32.3	. 169	. 4 5		5.065
		2 :					2662		200	6502	32.55	32.3	104	654.	.031	5.124
		<b>.</b> .					2711	1608		7602	46.0	32.0	176	624	. 033	196.9
						-	272		265	***	25.5	32.0	177	. 432	***	5.052
						7.	2724	14.30	1 to	2040	157	32.0	.178	• + 35	SE	5.046
							51.12	0.7	4.50	2043	34 30	31.7	.177	.431	. 036	5.120
							5646	0111	6926	2141	8.57	31.4	.177	924.	. 033	5.041
		22					1862			2111	2500	31.1	172	.438		4.977
		2					57.05			068 2	25.00	31.7	.175	MM4.	.032	5.144
		2				×.	6944	1510		2110	75.0	31.4	.17.	924.	- 034	5.035
		C:					2783	1450	6970	2119	3600	31.1	.173	. 430	. 033	5.046
		2					2719	1510	\$ 0.40 \$	2110	1620	31.4	.173	. 620	. 032	5.125
	•	À					2692	154	9264	2150	3663	31.1	.170	.335	3	

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CONTROL OF THE PROPERTY OF THE

	INTERVAL	BETHEEN BOILEDE	(SFC)		4.7.4	5.073	5.06€	5.148	5.030	5.042	5.128	5.035	5.037	9.124	5.038	4. 826	5.151	5.030	5.030	5.115	5.036	5.030	5.09B	4.922		5.125	5.077	5.036	5.065	5.827	5.015	2000	629.6	5.087	4.953	4.930	2.060	5.023	276.0	2.0.6	3.010	9.010		7.00	5.175	5.017	566.4	
	1	EJECT.	(384)		3	N. T.	F . E	ж. т.	. T	¥.7	×.	H.T.	H. T.	H.T.	H. 7.	H. T.	#. T.	H. T.	. T.	#.T.	H. T.	H. T.	z.	H.T.	H. T.	H. T.		H.T.	Z .	Z .	÷ ;				N. T.	H. T.	H.T.	2	E :	E :				. 2		N. T.		
COUNTR	R.COIL	- 20	(SEC)		. 345	. 345	. 353	. 352	. 370	. 300	. 364	. 373	. 369	.339	177.	.376	. 372	. 369	.375	. 362	. 371	. 367	. 367	H.T.	. 316	. 325	.323	. 325	. 327	. 331	. 330	965.	922	151	.360	. 530	. 342	5000		P 16 17 1		F 4			125	455	. 363	
	Fr.OIL	- 4200	(SEC)		.175	.170	.170	.177	.163	.170	.177	.107	.170	.172	.173	.174	-172	.175	.171	.167	.172	.176	.173	H. T.	.172	.170	.178	.160	.169	. 168	-172	-1/1		174	.172	-178	.168	-167	291.	171	101		601			174	172	
	H CX.	1100.0			31.1	31.1	31.6	31.7	31.4	31.7	31.7	31.6	31.7	31.7	31.2	31.7	31.4	32.1	37.3	31.7	31.7	31.4	31.4	31.4	31.4	32.3	12.3	32.7	31.0	33.0	43.0	N	• · ·		33.6	33.6	33.4	P (0 )		23.0	33.	<b>9</b> 6	77	1.75	12.1	31.7	11.7	
CAL INDA		MAX.	(PST)		1640	3650	1650	3640	3633	3639	4634	36 30	36.50	3638	3658	3648	3620	36.50	36.30	25.00	36.30	3600	3548	1570	3640	3690	76.0	36.0	3644	3600	1650	996	26.4	3858	3668	3651	366	4668	0 t	1695	200	900	997		26.0	4669	3660	
414000		578710	(6.2)		2173	2150	2148	2128	2120	2118	2110	2110	2110	1112	2111	2110	2119	2118	2118	2110	2114	2110	2110	2110	2220	2170	21.0	2120	2120	2120	212	0112	212	2113	2110	2129	1124	2128	B 21 2	2118	212	1212	212		212	7214	212	
WAL INDA		7 TO TO	(82)		0264	1960	7.96	124		£ 8 3		4750	1420	1420	• u • •	1267	<b>,,</b> ,	4490	1420	:	4428	1920	<b>.</b> 7	;	1720		**	•	£7 BB	199	194			;	14.	*7	*728	67.20	22.5	# ( ) ( ) ( ) ( ) ( ) ( ) ( ) ( ) ( ) (	4758					8464	•	
110314		STATIC	(651)		1450	7695	140	1448	1434	0741	1450	1.04	24.63	141	1400	::	1414	1478	146	13.0	1340	1340	1 76	1348	154	1450	14.30		1450	1400	1698	14.50		14.0	14.00	140	1 71	744			D / 7 P	r				-	1317	
					2707	764.	2111	5092	56.44	24.98	25.49	2047	2721	2704	2640	2714	25.77	2691	2714	2643	2711	2012	2672	2647	2673	2716	2447	2672	2762	1692	2707	68/2	****	2676	7714	2781	278.	2		5 P P P	2543	3			711	26.91	27.83	
	CHEMIN		190		1.1.	7		H. T.	4.4	H.T.	H.T.	.T.		H.T.		H. T.	H.T.			K . T.			H.T.			#: ±:	Z .	#.T.							4.1.		#· T.	2 1	-		- ·			;			-	
		27.00.00	1786	-	1756																																											
		*	51450	•	5																																											
	į	F. C. S. S. S.	1)[6-4]41	****	:																																											
		-	2		Z	£	P	ī	2	<b>5</b>	1	ž	ż	2	Į	<u> </u>	;	7	}	7	;	*	;	;	3	5	5	5	2	5	:	•	r C			;	3	3	;;	:	:	:	:	3	<u> </u>	. 5	2	
		BA 2 271	CE-400		1-02				•																																							
			1471		21/15																																											

								٠.	TLINDA	_	SYLIND			COUNTR		
2110			•									× * *	RECOIL	AECO IL		INTERVAL
			5			- 15 3 A	5	STATE		STATE	MAK.	RECOIL	00₹A-	-V-06	_	RETHEEN
	1000		ELFOAT 100	300	3	1151	VEL.	PRESS.	PR. 55.	PAESS.	PATSS.	OTSPL.	110M	TION		ROUNDS
		i	4.0 C-414	43614	144	100	(5/3)	11547	35	(154)	(154)	CIN.	(SFC)	(Sec)		(SEC)
										217		2 1 0 8	. 100	9 (		2.0.0
		: 2					16.57			6662	37.00		,,,			
41/11	1-14	•		PASTO			7776			2000			. 176	200		
		Ą	•		1		25.52			2 2 2 2	16.0	4	991			4. 720
		. ~				-	7.7	141		2002		32.5		36.2		3 1 1
		•				H. T.	27.25		0209	2404	464	200	141	180	031	5.068
		۴				M. T.	2677	1498	4770	1478	3510	3.4.6	191	157		4.779
		•					2711	1490	;	1471	3510	33.4	. 170	.333		4.709
						.T.	2636	149	474	1474	3510	31.1	.160	. 354		5.890
		•					2783	1631	1203	1179	3540	33.1	241.	.354		4.776
		• ;					2002	14.0		14.	3510	33.1	. 185	. 370		4.676
		=					2642	144	479	1970	100	33.1	.172	. 176		5.960
		=					1992		477	1470	35.0	13.1	.172	. 365		4.764
		~		1			7669	1-1	177	1970	3418	33.1	.175	. 167		4.691
22/17	7	<b>-</b>	= E	270	į	#: <u>.</u>	2692	1520	•••	2224	1679	33.4	.164	. 337		•
		~				.T.	2674	:	;	2229	3679	31.0	-182	.334		6.743
		m				7.	5992	1430	;	2204	35.50	31.3	177	. 337		5.134
		•				# · #	7992	146	4631	2168	3698	41.7	.171	. 320		5.128
		<b>.</b>					1647	24.4		2209	3710	31.1	170	970.		4.476
		•					2647	166	7	2100	3710	44.4	150	. 447		5.091
							5676	1.20	177	2140	-1-1	31.	.173	4800		9.110
		< (					2646	1.4		2190	3710	31.0		.362		5.093
		•				2	2609	14.3		2100	1,10	31.3	.163	. 167		5.079
		2 ;					2677		477	2160	3643	M . M	111	. 263		5.116
		= :					2679	3		2100	37.1.0	71.5	.176	.365		990 .
		<b>V</b> 1						4		8822	27.18	31.5				7
		2 ;								22	21.5					2.00
		::					26.00				11.6	? .				
		: ::					5632			;		2	-	N		Z .
		11					7661	-	H . T .	H. T.	N. T.	H. T.	H.T.	K.T.		N. T.
		:				7.1.	2476	H. T.	7.	. T.	H. T.	H. T.	н.т.	ж.т.		¥.1.
		2				4.T.	2638	#.T.	M.T.	. T.	. T.	. T.	¥. T.	W. T.		, z
		2					2656	#.T.	H. T.	.T.	H. T.		-	¥.		×.
		<u>ت</u>				# ·	2643		¥. T.	*	¥.	H . 4		-		
		22				H.T.	5992		#: <u>T</u>		¥.	H.T.		H.T.		¥.
		2					2566						, T	H.T.		Z:
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		L.				- K	6752		H. T.	7.X	. I	H. H.		# · L		2
		Z					2677			H.T.		H. T.		N. T.		
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		Z					11/2					2				
		2				#.±	3	. T.	. i	# 1	¥. T	. T	-	-		-

								שנכסנר כ	TL 1 MOR	FCOIL CYLINOR COUNTY CYLINDS	TYL THOR			COUNTR		
			1			CHANGE						MAN.	RECOIL	FC011		INTERVAL
110			3			22.2	5	STATIC		STATEC	HAY.	RECOIL	DU24-	-40P		BETWEY'N
	ייייייייייייייייייייייייייייייייייייייי	2	ELF + ATT 04	#03£	CHERCE	125	٨,	Perss.	Perss.	PAESS.	PKF SS.	CISPL.	MOIL	TION		ROUNDS
5	-10	į	10f6-HIN)	61419	301	CCC	(6/2)	:5:	Ē	(124)	(PSI)	(14.)	(SEC)	(SEC)		(SEC)
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77/55		::	-	0111	1		2002						× :			
		7 .					26.42									
		;					207				Z .	-		- L		
	1-82	<b>pa</b> (		2747	1		20.2	121		2101	3710	15.0	.164	. 295		•
	*	<b>N</b> (				-	5698	14.0	5120	2100	37.48	32.3	.174	. 130		4.737
		n,					2717		210	2150	37.50	32.9	.177	F 95 .		4.90
		2 (				H. H.	2701	2440	199	2151	37.39	32.9	.173	.376		5.047
		•					760	7	:	2153	37 18	D	.173	. 3A &		F . B.
		٠				#.T.	2640	1643	1111	2130	1710	31.7	.177	.381		4.826
						4.1	7646	1110	<b>6.</b> 7.8	2130	3710	31.6	.174	. 346		5.031
		•.				4.7.	2713	1363	:	2130	37.18	33.6	.17	. 393		4.620
		•				;	Z	::	•	2130	37.10	33.6	.175	•		120.4
		=				#. T.	2729	***	5020	21.13	3710	33.6	.177	:		5.00
		=				.T.	2478	***	6110	2150	3713	33.6	176	.410		4.839
		2							6930	2130	3711	37.6	.174	.415		4.023
		2;					27.09	1510	***	2150	1710	33.6	.111	£14.		5.036
		2					5692	1:1	5	2170	2 1 T	33.6	.17			4.051
			1				2072	111	•	2170	3710	31.6	.162	.420		1.11
		2	<b>3</b> 6			-			5	2140	2674		.160	.329		•
		2:						200		2100	3670	51.7	173	91		9.618
		::					•	7		1017	-	11.		121		910.
		2 :								2109	367	41.4	-	111		261.42
										210	200	31.	2/10	. 336		6.51
	ī	-	=======================================	01-01	3	- 1	E (1)					# ·				•
		N 1					27.23			•				- 1		19.4
		н.							•		H					9-120
		• •										•				2010
		A 4														
		•														
		•														. 1 .
		•				;	2707	F. T.			. T.	7	1			5.11
		:				H.T.	2683	H.T.		#. T.	H.T.	#. H	H.T.	H. T.		6.11
		11					2714	K. T.		#. T.	H.T.	H.T.	. T.	H.T.		9.11
		~					252	#.T.	. T	H. T.	H.T.					5.100
		*					2712							H.		6.130
		2				-	***	-	-					-		5.100
		*					2616			H. T.		#. ¥	H. T.	H.T.		5. 69
		2	, ,				H.T.	. ·					H.	-		•
		2							H. T.	#.T.			H. T.	H.T.		4.75
		=							#.T.		H. T.	#: <u>-</u>	#.T.			5.030
										-		H.H	H. T.			34.410
		2					N. T.	.T.	H.T.	H.T.	H. T.	H.T.	H.T.	T.T.		5.110
		2				#.T.	.1.		H.T.		H. T.	H.T.	H.T.	#.T.		5.120
		2				.1.	H. T.		H.T.	#.T.	B. T.	H.T.	H.T.	1.1.	E.T.	31.690

								AFLOR C	VL. 1 MDR	POTTINDE COMMIN CYLLLING	4C1.774			COUNTR		
						CHARAR						HAY.	PECOIL	RECOIL		INTERVAL
22			5,			AC 56.	7.5%.	STAFTC		STATEC	HAY.	REFORE	0U4A-	DU 4A-		BETWEEK
1140	BLACK	8050	ELTATION.	#03£	CHARCE	17211	vel.	PAESS.		PRESS.	PRESS.	.7.510	110M	110#		ROUNDS
1471	CE-110	\$	10f6-4fer	£1413	110	100	(5/2)	(154)		(151)	(PSI)	(1×.)	(256)	(215)	(250)	(SEC)
		2 :	: :	2111	į		:									1 1 1 1 1
		<b>4</b> 1							•							24. 426
		Ĉ i														6.878
		22					•			-						20.00
		Z.					•									20.10
		۲.														20.6
1	l	=			į							2	-			2.0
123	1-02	_	*	2	מבר	-	# # # # # # # # # # # # # # # # # # #									
		~				;	2446			M. T.			-		2	
		-					7611			H.T.	×.	. ·		.1.	H. T	4.650
		<b>.</b>				# · T	2545	*	H. T.	¥.T.	Z.				H. T.	5.140
		•				H.T.	2641	#.T.	:	. ·	. T.	H.T.	ж. 	#. <del>.</del>		
		•				4.1.	2765	#: <u>T</u>		H. T.		#. T.	#.T.	N . 4	H .	
							***	#.T.	#. T.	#.T.	N. T.	#.T.	H. T.	-	x. T.	5.136
		-				H. 4	26.93	H.T.	7.4	#:T:	H.T.	H.T.	#. T.		H.T.	· · · · · ·
		•				H.T.	2746	H. T.		H.T.	H. T.	#.T.	H. T.	#.T.	#. T.	4.050
						H . T	1692	H.T.	H. T.	#.T.	. T.	H.T.	H. T.	H. T.	H.T.	5.125
		4					2642	.1.1	W.T.	H.T.	M.T.	H.T.	H.T.	N. T.	#.T.	4.84
		~				H.T.	2643	H.T.	. T.	H.T.		H.T.	H. T.		H.T.	** **
		-				#.T.	2477	#.T.	. 4		H. T.	. T.	H. T.	#.T.		8.628
		1				#:T:	2692	#.T.		H. T.	#.T.	H.T.	#.T.	m.T.	#. T.	::
		2				<b>.</b> 1.	247			<b>.</b>		#.T.	H.T.	. T.	H. T.	6. <b>8</b> 5
		=		202		H.T.	2720	H. T.	H. 4.	H. 1.	H.T.	#. T.	#. ·	H.T.	H.T.	•
		-				#.T.	14.52	H. T.		#.T.	#.T.		H. T.	H.T.	H.T.	•
		=		2		#.E.	27.22	H. T.	.T.	H. T.	R. T.	H.T.	H.T.	H.T.	H.7.	•
		=				#.T.	1692	#.T.		. T.					H.T.	4.720
		Z					26%	H.T.	#.T.	#: <u>T</u> .	#.	M.T.	#. T	- Z	Z.	9.0
		<b>Z</b>				3	2643		÷:			- I				2.100
		22					202									
		2														
		٤;				:	1007									3 8
		C X					7647									4.020
						,	20.53					*	1.1	H.T.	N.T.	5.130
		2				-	1652			1.1	H. F.		#. T.	N. T.	M.1.	4.870
		2		51.0m	*£ 3.	4.7.	M. T.	H.T.		H.7.	н.Т.	.T.	m.T.	¥. T.	H.1.	
12/03	1-02	••	27 01	01414	FULL	×.1.	1952		f. T.	Z.	#.T.	H.T.	H.T.	m.T.	H.T.	•
		~				W. F.	(2)	#. T.	M. T.	#. T.	H.7.	#: <u>.</u>	#. T.	 	H.T.	19.229
		*				::	9492	H.T.	 	.T.	H. T.	H.1.	H. T.	m.T.	H.T.	14.200
		•				. T.	- 25	M.T.	. · ·	M. T.		. T.	H. T.		-	15.780
							27.72		#. T.	.T.	Z . Z	- Z	¥.7	#.T.	#: #	16.470
		•					764.	×. 4.	;		*			H.1	# · #	14.130
	•	٠					\$2.52	M.T.	H. 7.		T.:		H.T.	F. T.	# ·	15.020
		•				÷.	\$448	¥ . ¥ .	H.T.	ř. T.	N. 7.		#. T.	H.7.	H.T.	19.120

								PERMIT	FFEL FYLIND® COUNT® CYLTHUS	C)UN1*	CYLTYCE		į	COUNTR		
****						-				•		MAK.	RECOL	Fr CO IL		INTERVAL
5.4.2	1,48.75	_	101.44.5.7		97867	1655	K .	54 B1 3C	446.	21.1.10	# C	110000	503	-4-00	£ JEC7.	BLTMFEN
17	C4-430	63	1350-550				,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		2000	•			200	P (		4000
		٠											1361			10000
11/11	1-92		11 .7	Clara	FULL	#. T.	2692		, L		,	7.7	N . Y	N. 1.	7	16.75
		0				4.1.	9 H & C	-			7	N . Y			7	14.640
		=				#.T.	2692	F. T.	. T	H.T.	H. T.			× . 7	N. T.	20.120
		2				4.1	2666	N. T.	H. T.	#.T.		*.T.	#. T.	R.4	W. T.	13.320
		-				4.7.	2502	H.T.	, ,	٠,٠	7.1	#.T.	H. I.	N. T.	H.1.	14.650
		2				F	2016		. T.		N. T.	¥.7.	. T		, z	21.750
		2					26 76		F.	N . Y .	. T.	Z	7.7		N. T.	12.790
		• :	•	51.0e	ė.			2	2	7			-	7.7	Z. T.	•
			;				P 1 2		7			. K			2	• :
							1512	2	2		Z					7.800
		2					212								2 :	7.790
		::												- 1		0.76.9
		: 3				;	1017								- h	4.780
							21.6		, 1							0.7.
							1									
		£					111			,						3 5
		£					2151				1					4.00
		1					1165	H. T.	H. T.						1	4.770
		ž					\$414	# · T ·		7.2		N.T.	M. T.	1.1	M. T.	4.770
		Z					2155	.T.	H.T.	¥. T.	#.T.	N.1.	H. T.	#.T.	H. T.	4.920
		2					2137	7.2	# -		#.T.		¥. 4	# 1.	H. T.	4.780
							2170			#.	7.7	K. T		¥. 7	H.1.	4.770
		~					2144	1.1.	. T.	. T.	¥.T.	H.T.	H . H	M.T.	H . T	•
						-	41.4	-		H. T.	¥.T.	-			H. T.	4.810
							2131		2		. H.	<b>.</b> .	2		F. 1	4.720
							112				2 :		Z			7.850
		:													· .	018.7
		E					25									2
							2175								T. T.	6.810
		;					5512	H.T.	-	H.T.	R. T.	N. T.	# · 1	H.1.	N.T.	7.800
		<b>;</b>					512	H.T.	¥.		H.T.	H.T.	#.T.	N. 7.	H.T.	4.830
		;				;	7.				-	Z			H. T.	4.790
		5					2712	-				- ·			¥.	018.7
		<b>;</b> ;				- i	7178	2	in in		N . Y	N. K		H.1.	F . F	4.820
		;					2167		H.						*	4.810
	;	3	:	į		; F						÷.*	-	H	¥.	7.820
Ş	1-12	- 1	2	2075	ž	**	25.5	¥.			2	h.1.			; ;	•
				9			20.42									•
		٠,								- ·	2				-	
		• •					\$				2		- (			5.070
		۸,					7847		· ·			. 1			E	010
		• •				: .										016.9
								:			ř					

						2000		Afron.	Frott Cylinde Counte Cylinge	COJNTR	PC#1740	1	1000	PENCOS		
### Company of the co			3			. 22.	T& JH.	514716	. 174			Pr CO Te	0034	Die 3-	F. 15 C.Y.	BE THE PARTY
### ### ### ### ### ### ### ### ### ##	-	\$ 00kg	FLESSTEEN	4034	S sect	(151)	W.t.	PRESS.	PPE CS.		Per ,S.	74526	TICM	1174	111	ROUNDS
		3	(A) 3 aL)	.146	Tree	503	(6/3)	11541	(154)		(PS1)	114.1	יצינ	(560)	(35)	(235)
						•					:	-		•		1 1 0 0 0
		<b>5</b> (	27 81	01	1		2.1				× ×	¥.			N. T.	4.933
		P :				•	1492					z		-	7.1	0.850
							76.7				2			*	N. T.	5.160
		= :					2676			7		7			H. T.	4.920
		= :				- 1	21.26						7			
		2 :					24.			2			2	2	2	5.350
		2					1644			-			H.		H. T.	026.7
		y					11.2		<b>.</b>		¥. f.	¥.1	¥.	. T.	H.T.	4.930
		•					27.84				N . T	N. T.	¥. T.	N. T.	M.T.	5.130
		<b>.</b>					1994		-	#.T.	Z	F . E	ж. Т.	M.T.		064.4
		:					7647		1.1	H.T.	F. T.	₽.4	H. 7.		N.T.	4.900
		-				Z . 4	7729	#.T.		H. T.	7.7		н.Т.	¥.1.	N. 1.	5.130
		23					27.88	N. T.	H.T.	H.1.	ř. ř.	7.4	N	M.T.	H.7.	906**
		Z				#.7.	2637	H. T.		¥.4	N. 7	N.T.	#.T.	H. 7	N. T.	4.910
		22				#.T.	76.7	#. <del>.</del>	#.T.	#.T.	7.4	7	H. T.	H. H	N. T.	5.160
		2					44.94	#.T.		F.1	#.T.		F. 7	W. T.	¥.7.	4.410
		Z				#. T.	1192		#.T.	H.T.	7.4	M.T.	N.T.	H. T.	H.T.	4.920
		X:					2192	H.7	#·1	#.T.			H.T.	X.1.	#.T.	5.126
		Z:				:	10.2		÷	7.		. T.	#.T.		H. T.	4.918
		2 8					26.75				* (		F :	H.1	¥.	900
		Z.					-							H.1		5.160
		ζ:					400				¥.	- I	# : d	E	H. H.	91
		2:				<u>.</u>	2007			E	2	- (				9.9
		= 1					20.00								1.	021.6
		7:					200		•	- 1					- 1	826-4
		: :						•			K			- 1		0.10
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		;					2444									21.100
		ï					2041					1	H. T.		T. T.	16.620
		;				#. T.	707	F. 7		H.7.		1.1	7.1	H.T.	H.T.	17.250
		£ 3				<u>.</u>	26.95			1.1	M. T.	2	H.1	1.1	H. T.	20.650
74.00 74		,					1094	#.4	7.1	M.T.	7.4	N.7.	M.T.	W.T.	M.T.	16.330
2474		ş					2052		H. T.	H.1.	W.T.	H. T.	H. T.	H.T.	H. T.	1600
2667		;					7474	4.7	7.7.	4.7.	#.T.	h. T.	H.T.	7.4	M. T.	24.480
2445		7				<u>:</u>	1052	m.T.		X	H. T.	M.T.	H . 4	H.T.	H.T.	16.360
2446 M.T. M.T. M.T. M.T. M.T. M.T. M.T. M.T		5				÷.7.	1992	#.T.	H.T.	#.T.	H.1	N. 4	#. T.	H.7.		17.910
2670 E.T. Z.T. Z.T. Z.T. Z.T. Z.T. Z.T. Z.T.		5				:	3342	H.T.	<b>.</b>		H.T.	¥.1.	H.T.	<b>H.</b> T.	H. T.	17.720
265 m.1. m.1. m.1. m.1. m.1. 267 m.1. m.1.		=					2676	7.				H.4		. T	H.T.	14.120
.I.M .I.M .T.V W.T. M.T. M.T.		<b>;</b>					300				. H . H		Z	Z		14.680
		<b>:</b>					2442			:	-				¥. ĭ.	16. 76

TABLE G-3 (Continued)

						C+8+3		Pr.CIL.		רס שות סי	9(14)A	,	110036	CHIMTA		THTERUAL
	110	900	PC2117277	100	Cabron	.1521	7.0M.	57 47 10		51111	PAC 25.	11	30.44-	1104	EJECT.	POINCE
	430	è	184 - 5 TG1	01-11	7	5	15/31	11501		1154)	(bell)	14.3	(Src)	(257)	(SEC)	(SEC)
		:							•							
	•			01414	1		200							· • ·		
		<u>,</u>					26,47	7								17.75.0
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		ě.				7.1	7694	H.T.	#. T.			H.T.	N. T.	N.T.	W. T.	20.750
		5				¥. T.	1644	n. 1	H.T.		7.2			7.2	7.1	16.410
		ij				×. 1.	2152		. T.	. T.	n.1	¥.T.	-	N. T.		17.130
		-					3.94	R. T.		F. T.	¥.7.	₹.1		÷.	7.7	200 707
		3							*		.1.4		· ·		7.7	15.250
						7.	26.54		7.1	;	N. T.				, . , .	14.050
		:				,	467		<u>.</u>					. I . z	. <del>.</del> .	20-1/0
		Ş					2645	#. <del>.</del>	4.1.	z	H. T.			H. T.	#.T.	15.1.0
		*				7.7.	2592	H.T.	H. 4	H. T.	#.T.	H.T.	¥.↑.	n. T.	¥. T	15.470
		2					E	-	. T.	7.1	Z. T.	n.7.	7.7	Z . T	N. T.	19.468
							1652		#. T.	H. 7.		7.1	ĭ.1	N. T.	H.T.	14.360
		3				۲.۲.	2478	#.T.	H.T.	H.7.			N. T.	N. T.	¥.7.	16.270
		<u>,</u>					ε	ř.	H. T.			м.Т.	4.7.	.T.W	H.T.	19.838
		-					3	H. T.	H.T.	.1.2	H . T.		<u>.</u>	N.T.	H. T.	16. 36 2
		4 C					ť			H. T.	#.T.	H.T.	¥. T.			15.040
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		2 :		207			Ê		7				Z			•
	٠	ζ,		1			1092	-		-			-			• :
	-	-		2	3	7.	-	#, #	-		# · ·	. ·	#. T.	H.T.	#. T.	H.T.
		۰.									#. ·		. T.	H.T.	H.T.	H.T.
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		e 1						-			¥.		H.T.	K.7	÷	. T.
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10   10   10   10   10   10   10   10		<u>.</u>				::			H. T.	.1.	n. T.	2.4		#.T.	. T.	#. T.
105C+ 46010 FOAL		<u>*</u>					-:-	.1.		. T.	¥.	z	κ. T.	H.T.	H.T.	H. T.
# 1	ī	-	<b>36.</b>	0794	35	::		#: <u>T</u>		 			H.T.	#·	H.T.	•
### ### #### #### #### ##### #########		•				7.4	#.T.		H. 7.	H.).			H.T.	H.T.	H. T.	4.760
# 1		-				H. T.	<u>:</u>		<u>.</u>	#: <u>-</u> :	H.7.	¥.1		H.T.	H.T.	5.130
		•				;					.1.	7.7	H. T.	H.T.	#.T.	33.860
#.		ŧ				.T.	4.T.	-	#.T.	H.7.	R, T.		H. T.	H.T.	Z . T	4.438
The series and the series and the series of		•				4.7.	· ·	-		×.	#.T.		N. T.	#.T.	H.T.	4.950
		•					T-1		-				H. T.	H. T.		47.130

								SFC014 (	T. 1 1004	FCOLL CYLINDA FOUNTR CY. INJA	CA. ENJA			CO:14TR		
1						-			•			MAK.	F COLL	RE COLL	I	INTERVA
			5			. 22.		STATE	. 17		. A TH	120.2		- 4	t JECT.	FE THEFIN
			C.C. 187 184	100	3	141	AEL.	.223.	out es.	F175.	PRESS.	01301.	4 KON	1104	1 1 ME	BOUNDS
		2	1066-4640		*	3	5/2	1150	Ē		11541		(25,0)	STC	(SEC)	(SEC)
		• •			1		•	•								200
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## APPENDIX H

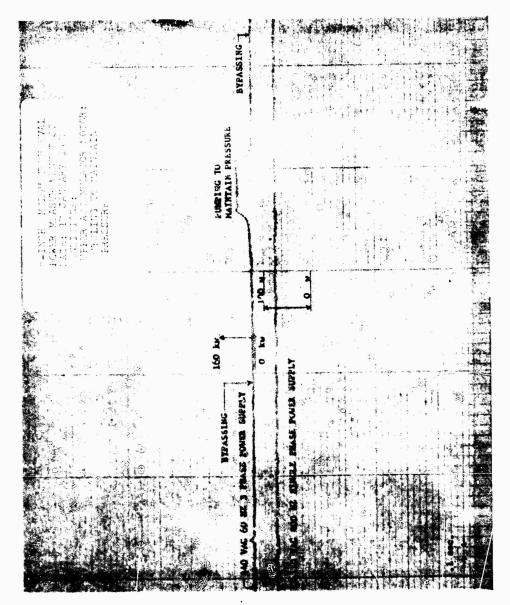
Power Measurement Data Table H-1 Figures 1 Through 12

TABLE H-1

8"/55 MCLGM TECHNICAL EVALUATION
TABULATION OF POWER MEASUREMENTS
440 VAC 60 Hz 3-PHASE POWER SUPPLY

MAJOR COMPONENTS	Figure Number	Measurement (KW)
Upper Accumulator		
Start	1	166
Maximum Pumping	1, 2	83
By-Passing	1, 2	10
Lower Accumulator		•
Start	3	127
Maximum Pumping	3, 4	64
By-Passing	3 3, 4 3, 4	3
Train Power Drive		
Start	5	287
Standby	5, 6	22
Synchronizing	6	367
Synchronized to SHM, 30°		
ampl 9 sec period	6	6 to 92
Elevation Fower Drive		
Start	7	220
Standby	7, 8	12
Synchronizing	8	89
Synchronized to SHM, 30°		
ampl 9 sec period	8	0 to 57
Blower on	Q	19
Anti-Icing System On	10	38
And long System On	10	
MOUNT OPERATIONS		
All Motors Star: (Maximum)	11	414
Loading System Operating in	11	7.7
Simulate, laying system		
synchronizing at maximum		
velocity, blower on, anti-icing		
system off (peak power		
condition except the anti-icing		
system is off)	12	599
110 VAC 400 Hz SINGL	E PHASE POWER S	UPPLY
Peak Reading	12	.75

FIGURE 1



11.4

FIGURE 3

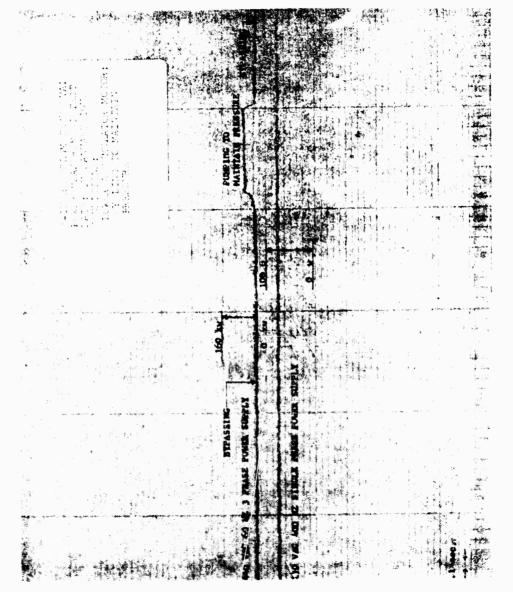
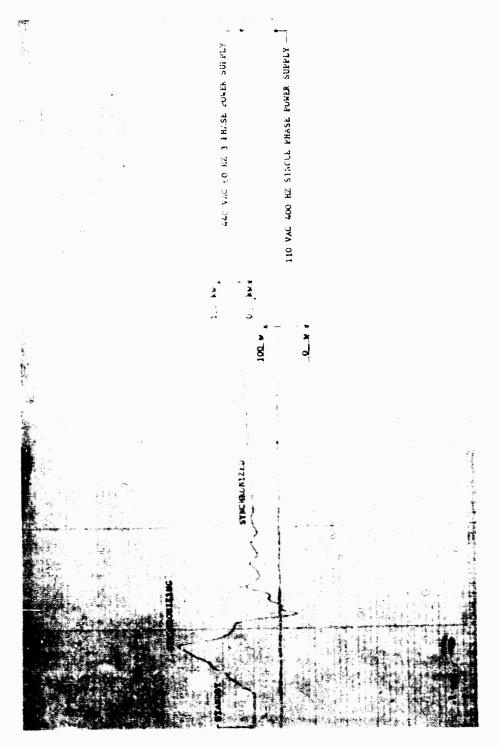


FIGURE 5



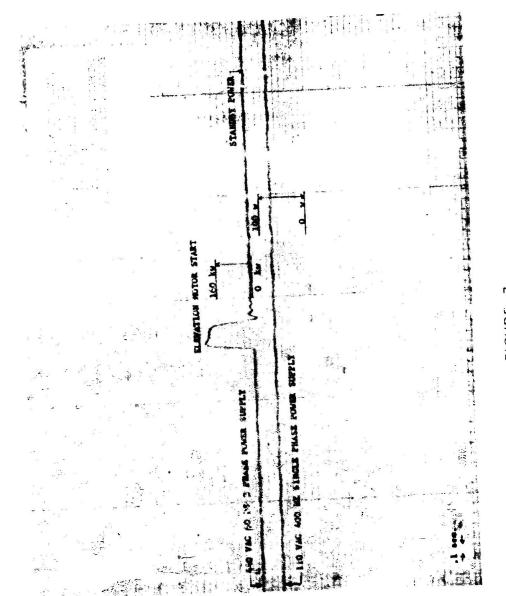
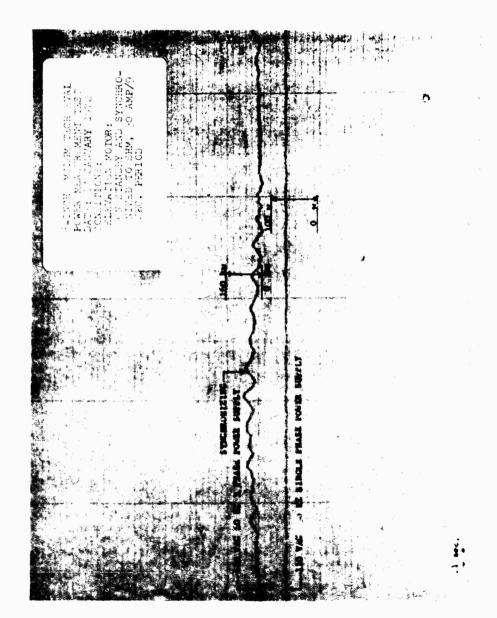


FIGURE 7



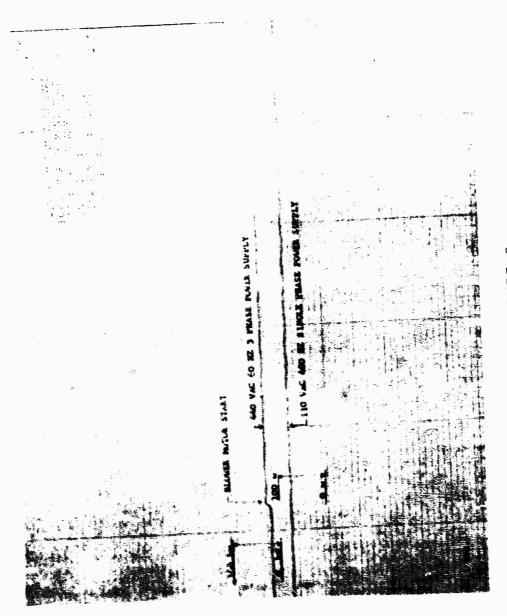


FIGURE 9

IGURE 10

FIGURE 11

FIGURE 12

APPENDIX I

Gun Jump Data Figure 1

a series

.

# 8"/55 MCLGM TECHNICAL EVALUATION DISCUSSION OF ATTEMPTS TO MEASURE GUN JUMP

#### General

It was decided not to include a gun jump test as a discrete Tech Eval test, but to attempt some experimentation prior to the planned rate of fire test of 2 November which might give an approximation of gun jump for one angle of elevation and give some basis for refinement of experimental techniques. Three different methods of determining gun jump were investigated.

#### Method I

Method I involved placing a camera at point A (Refer to Figure 1) at a distance R behind the gun and a distance H above the gun. The camera was aimed at an angle near  $\alpha$  such that point D will be within the field of view. Point D is defined as the point along the trajectory where angle  $\alpha$  equals angle  $\theta$ . Then

$$\tan \alpha = \tan \theta \tag{1}$$

$$\tan \theta = \frac{\dot{z}}{\dot{x}} \tag{2}$$

$$\tan \alpha = \frac{z - H}{R + x} \tag{3}$$

$$\frac{\dot{z}}{\dot{x}} - \frac{z - H}{R + x} = 0 \tag{4}$$

Equation (4) can be satisfied for a time, t, by using particle trajectory data and interpolating. Using this value of t, z and x can be found from Equation (4) and  $\alpha$  can then be obtained from Equation (3). The gun jump value can then be determined by measuring the displacement of the projectile above or below the center of the film (on the frame where the projectile has reached a maximum height).

A - Location of cameras 1 % 2 B - Location of gun trunnion C - Location of center of grid D - Point where line A-D is tangent to trajectory E - Location of camera 3 - Trajectory

Geometrical Arrangement for Gun Jump Tests

FIGURE 1

#### Method II

A scaled rectangular grid was placed at point C and centered by boresighting. The projectile was photographed passing through the grid from points A and E. The displacement of the projectile from the center of the grid after subtracting trajectory drop distance, determined from particle trajectory data, gives the resulting angular displacement.

#### Method III

This method involves comparing observed ranges with theoretical ranges and computing gun jump based on the difference between the two.

#### Results

On 2 November, three rounds were fired at three degrees elevation with equipment set for applying the three methods. The camera set to observe point D (Method I) was not successful, however, because the angle  $\alpha$  could not be estimated closely enough prior to the test. Method II yielded gun jump values of 4.14 minutes for round one and -1.07 minutes for round two. The camera at point A failed for round three. Method III gave a difference in departure angle between rounds one and two of 2.4 minutes. These results do not demonstrate a consistency needed for a good experiment and additional effort is needed before conclusions can be drawn.

# APPENDIX J

8".0 MCLGM TECHEVAL Blast Test Data Figure 1 Table J-1

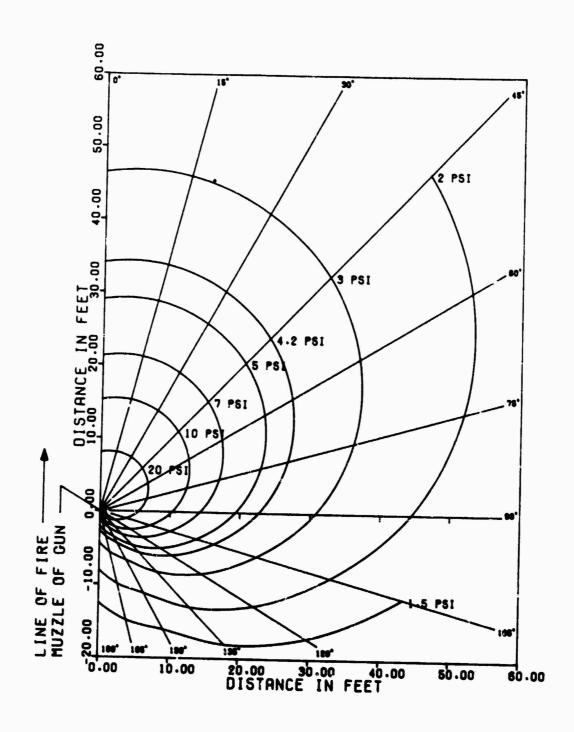


FIGURE 1
Free-Air Peak Overpressures for 8-Inch/56 MCLG Mount

TABLE J-1 EX 71 MOD 0 8"/55 MCLGM: TABULATION OF BLAST DATA

From Muzzle No. of Overpressure Inne (Feet.)  43.0  5 3.7  26.6  3.7  26.6  3.6  27.5  3.6  27.5  3.7  26.5  3.7  26.5  3.7  26.5  37.7  26.5  37.7  26.5  37.7  26.5  37.7  27.5  27.5  36.7  27.5  36.7  27.5  36.7  27.5  36.7  27.5  37.7  27.5  36.7  27.5  37.7  2
From Muzzle (Feet.) 43.0 54.0 21.5

TABLE J-1 (Continued)

TABLE J-1 (Continued)

Angle from Trajectory (Degrees)	Radial Distance From Muzzle (Feet)	No. of Rounds Fired	Free-air Peak Overpressure (PSI)	Arrival Time (MS)	Positive Pulse Duration (MS)	Positive Impulse (PSI-MS)
45	20.02	v	4.9	12.0	9.6	
			6.7	12.1	3.7	
			5.9	11.9	3.6	
			<del>7</del> .9	1.9	3.3	
			6.2	2. E	2.8	
		Averages:	: 6.3	11.9	3.3	7.8
45	41.0	•	5,	28.0	4.1	
			3.7	28.1	. ec	
			9.6	28.0	e c	
			6.6	27.9	1.7	
			3.6	27.9	· •	
		Averages:	3.7	28.0	6.4	6.7
45	61.0	•	2.4	42.1	0.4	
			2.3	6.14	4.7	
			2.3	8.	. S	
			2.2	41.9	£.3	
		Averages:		41.9	7.7	3.7
45	71.0	-	1.6	52.7	5.7	3.4
3	17.0	\$	9.6	10.9	2.6	
			8.6	10.5	3.5	
			8.6	10.5	2.6	
			6.6	10.1	3.0	
			10.1	10.4	1.9	
		Averages:		10.5	2.7	6.6

TABLE J-1 (Continued)

Positive Impulse (PSI-MS)	6.3	æ. <b>₹</b>	7.8	5.1
Positive Pulse Duration (MS)	9.6.4.4.4.6.5.6.5.6.6.6.6.6.6.6.6.6.6.6.6	4-4000 4-4000	2.2 2.3 2.3 2.2 function on 1 Rd. 2.2	8 6 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8
Shock Arrival Time (MS)	23.4 22.8 22.9 24.2 25.7 23.8	8.85.7 7.08.92.92 8.83.3 8.83.3	8.9 7.4 8.8 10.3 stion Halfund	20.9 20.9 20.8 20.8 30.8
Free-air Peak Overpressure (PSI)			11.4 8.8 8.8 8.7 Instrument: 9.4	
No. of Rounds Fired	5 Averages:	5 Averages:	5 Averages :	S Averages:
Radial Distance From Muzzle (Feet)	35.5	<b>5.</b> .5	3.6	<b>9</b> .0
Angle from Trajectory (Degrees)	8	<b>9</b>	25	22

TABLE J-1 (Continued)

Angle from Trajectory	Radial Distance From Muzzle	No. of	Free-air Peak Overpressure	Shock Arrival Time	Positive Pulse Duration	Positive Impulse
Carried States	73237	Konnas Fired	7167		(CH)	(PSI-R5)
75	44.5	s	2.2	31.7	6.4	
			2.5	31.1	5.5	
			2.4	31.2	9.	
			2.3	32.6	-	
			2.3	35.7	3.8	
		Averages:	.: 2.3	32.4	4.5	3.9
8	9	•		•		
2		n	15.0	7.	0.	
			7.	7.3	2.0	
			10.5	7.1	6.	
			9.1.	7.7	1.7	
			9.00	7.1	1.8	
		Averages:	: 11.3	7.2	8.1	7.5
8	* *	¥	-	200	•	
<b>X</b> .	•	*	,	7.6	, 6	
			,,		7.0	
				0.6		
			3.3	0.6	3.9	
			3.5	19.0	4.2	
		Averages:	m.e ::	19.1	3.8	4.7
\$	42.0	•	• -	32.0	1 1	
2		•	N (	20.00		
			<b>10</b>	31.9	7.7	
			2.0	31.7	3.5	
			8.	31.7	3.2	
			2.0	31.8	3.8	
		Averages:		31.8	3.7	7.8

TABLE J.1 (Continued)

ise Positive Impulse (PSI-MS)		14.3		on a	o. 90.
Positive Pulse Duration (MS)	 		000000 00044	ง ผลเลยเ ง ก่อก-	000-000
Shock Arrival Time (MS)	244.2 7.2.2.7	5.4	22.22.25.6 6.6.6 8.22.8	2 2225 2.7.7.4.0.7 2.0.7.4.0.7	4 11 14 14 14 14 14 14 14 14 14 14 14 14
Free-air Peak Overpressure (PSI)	9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9				
No. of Rounds Fired	w	Averages:	w		S Averages:
Radial Distance From Muzzle (Feet)	7.0		17.0	27.0	vo vi
Angle from Trajectory (Degrees)	ž		25	ğ	<b>021</b>

TABLE J-1 (Continued)

ngle from rajectory Degrees)	Radial Distance From Muzzle (Feet)	No. of Rounds Fired	Free-air Peak Overpressure (PSI)	Shock Arrival Time (MS)	Positive Pulse Duration (MS)	Positive Impulse (PSI-MS)
<b>8</b> 20	14.0	un:	6 4. 6	10.6	2.2	
		Average	ກໍພຸພຸພຸ	12.9		ŗ
120	26.0	· .		20.9	2.7	-
			2.5 0.6 0.0	8.8.8 8.8.8	3.1 3.0	
		Averages:		8.8 8.8	3.1	2.1
135	4.2	s	8.5.8	4.4	1.0	
		Averages:	8. 4. L. L.	प्रस् प्रमुख	0.0.6.0	2.2
135	11.0	w	e, e	8.6	2.3	
				່ວຸວຸດ		
		Averages:	, e.	9.6 9.6	2.3	2.7

TABLE J-1 (Continued)

Positive Imp lse (PSMS)		1.6		1.6		1.7		1.6
Positive Pulse Duration (MS)	4.8.6.1.4.	2.5	-::-::	1.2	9985	67.	v e s s s s	2.3
Shock Arrival Tine (MS)	18.8 18.8 18.6	18.6 18.7	7.6.4.7. 4.0.0.6.	6.7 5.1	0.00.00 4.4.1.	8 8 8 8 8 8	11.2 10.8 11.0	11.8
Free-air Peak Overpressure (PSI)	1.8	1.7 1.7 1.7	ଷ ଦ ଧ ନ ଶ କ ଶ ଶ	3.9 3.8	2.5 2.8 2.8		2.9	
No. of Rounds Fired		Averages:		Averages:		Averages:		Averages:
Roun.	\$	₹	<b>v</b>	~	ហ	¥	vo	
Radial Distance From Muzzle (Feet)	5 22.0	ě.	5.0	4	9.0	×	3.5	

TABLE J-1 (Continued)

Angle from Trajectory (Degrees)	Radial Distance From Muzzle (Feet)	No. of Rounds Fired	free-air Peak Overpressure (PSI)	Shock Arrival Time (MS)	Positive Pulse Duration (MS)	Positive Impulse (PSI-MS)
165	8.6	v	2.9 2.9 2.9	6.3 6.3 6.2 6.1	22.26	
		Averages:	2.8	6.1	æ. æ.	1.9
165	9.0	vs	0.22.5	7.5	e.e.e.e.e.e.e.e.e.e.e.e.e.e.e.e.e.e.e	
		Averages:	. 5.3.2 2.3.3.2	0.9 0.3 7.8	1.7	1.6
165	13.0	s	7.1	12.2	2.5. 2.4.	
			1.7 Instrumenta 1.7	12.0 12.0 ntation Malfur 12.2	2.4 nction on 1 Rd. 2.4	1.5

APPENDIX K

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III ABSTRACT				

The prototype 8-Inch Major Caliber Lightweight Gun Mount, MARK 71 MOD 0 was subjected to a technical evaluation from 24 September through 12 November 1972. A description of the evaluation and test results are given, with tabulations of ballistic and mount performance parameters. Results and conclusions are discussed.

(PAGE 1)

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